



Topology Optimization - 1 - OPTISHAPE

1. Topology Optimization
2. Homogenization Design Method
3. Background of the New Approach
4. Mathematical Formulation
5. Optimality Criteria Method



What is Topology Design ?

- Shape design keeps the initial topology, while the shape of exterior/interior domain is designed.
- If an extra hole is generated, or if two holes are merged to a single one, we say topology has changed.
- Finding the number, location, and shape of the holes is a typical topology problem.



OPTISHAPE

The key idea is to transfer
shape/topology design

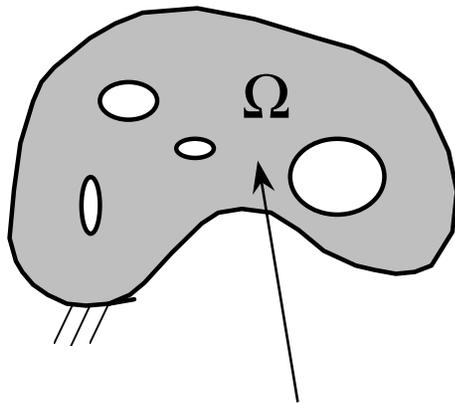
to

Optimum material distribution
with on/off switch condition



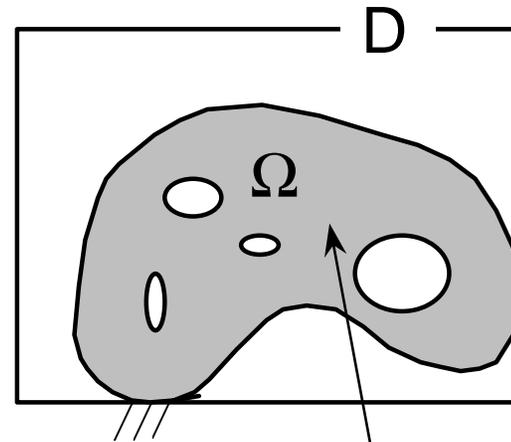
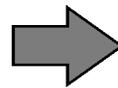
Idea in OPTISHAPE

Extension Ω to the fixed domain D



The elasticity tensor E

$$c_{\Omega}(x) = \begin{cases} 1 & \text{if } x \in \Omega \\ 0 & \text{if } x \notin \Omega \end{cases}$$



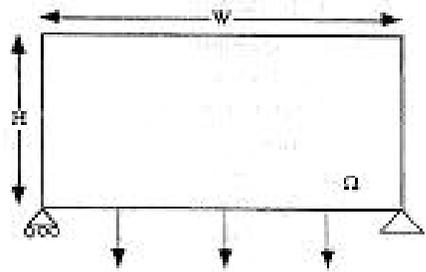
$c_{\Omega}E$



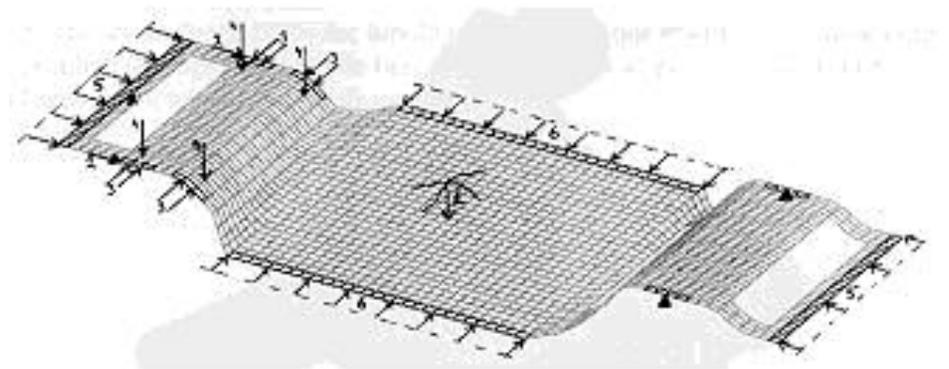
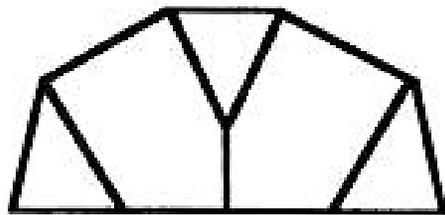
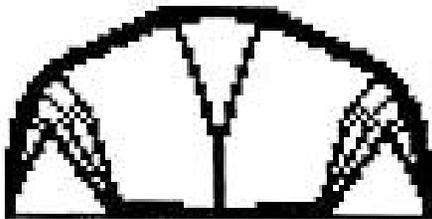
$$c_{\Omega}E \in L^{\infty}(\Omega)$$



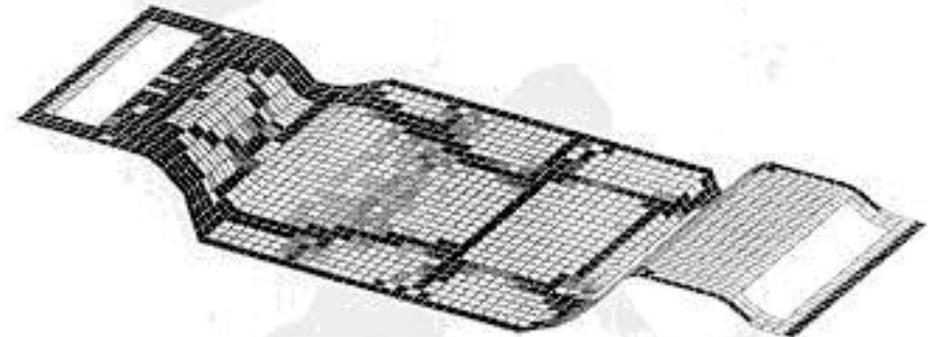
Examples by OPTISHAPE



bridge design



1/2 oor panel r einfor cement





Topology Optimization

- has become an important and well recognized sub-area of structural optimization
 - Design Sensitivity Analysis (1960s & 70s)
 - linear and nonlinear problems
 - Sizing Optimization (1960s)
 - Shape Optimization (1970s & 80s)
 - Topology (Layout) Optimization (90s)
 - Discrete and Continuum Topology Optimization
 - Material Based Optimization
 - Extension to MEMS area



Evidence

- Last Two : 1st and 2nd World Congress on Structural and Multi-disciplinary Optimization (Germany95 & Poland97)
- There are numerous sessions on topology optimization related
- Commercial Codes
 - OPTISHAPE(Japan), OPTISTRUC(TUS)
 - MSC-NASTRAN, ANSYS ---- Fall 97



OPTISHAPE : Present

- Maximization of the global stiffness of an elastic structure
- Maximization of the mean eigenvalue problems for free vibration
- Combination of the above two
- Maximization of the dynamic stiffness for frequency response problems
- Heat Conduction/Thermal Loading



OPTISHAPE : Near Future

- include SHAPE OPTIMIZATION capability based on Azekami and Shimoda's Method (at Mitsubishi Motor) for detailed shape design after the standard topology optimization
- include sensitivity analysis for sizing

- **TOPOLOGY + SHAPE + SIZING**



OPTISHAPE : Future

- Compliant Mechanism, Mechanism, and Flexible Body Design
 - to control deformation and motion of structures, flexible multi-bodies, compliant mechanisms, and even mechanisms to have integrated synthesis study of mechanical systems
 - toward smart structure design with control
- Material Design
 - Young's and Shear moduli and Poisson's ratios
 - Piezo-electric material design for MEMS



<http://www02.so-net.or.jp/~quint>

for
more information

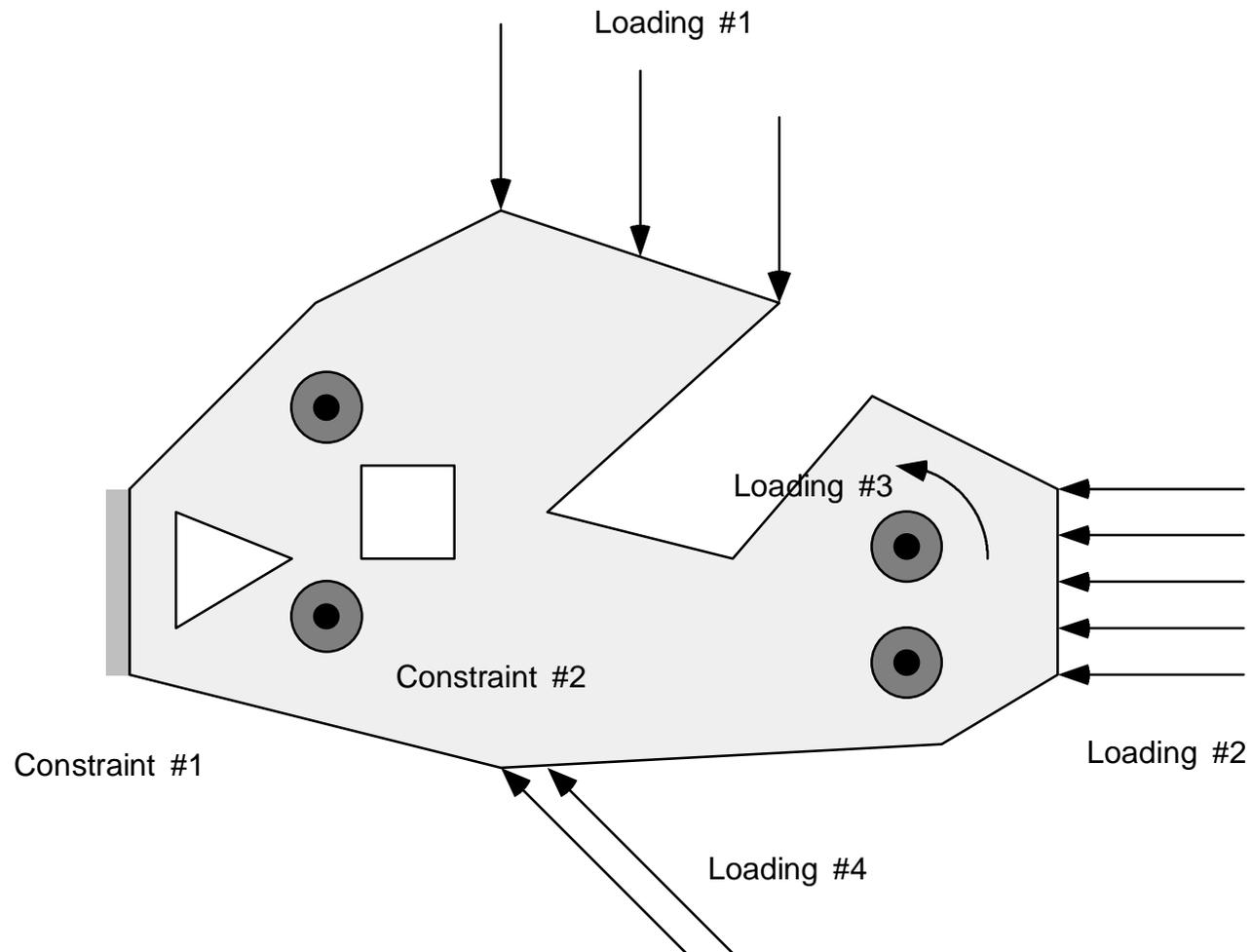


Typical Procedure 1

- (1) Define a design domain which contains the final optimum structure
 - geometric restriction for the on/off condition
 - on-flag : solid structure always exists
 - off-flag : void (hole) must be assigned
- (2) Define the loading and displacement constraint
 - multiple loadings and multiple constraints are possible in OPTISHAPE



Multiple Loadings & Multiple Constraints





Typical Procedure 2

- (3) Define the volume (or weight) constraint

$$\int_{\Omega} \rho d\Omega \leq W_1$$

- (4) Applying OPTISHAPE, and obtain
 - the Optimum Layout (Topology & Shape)
 - the Maximum Mises Equivalent Stress
 - the Mean Compliance and Strain Energy Density

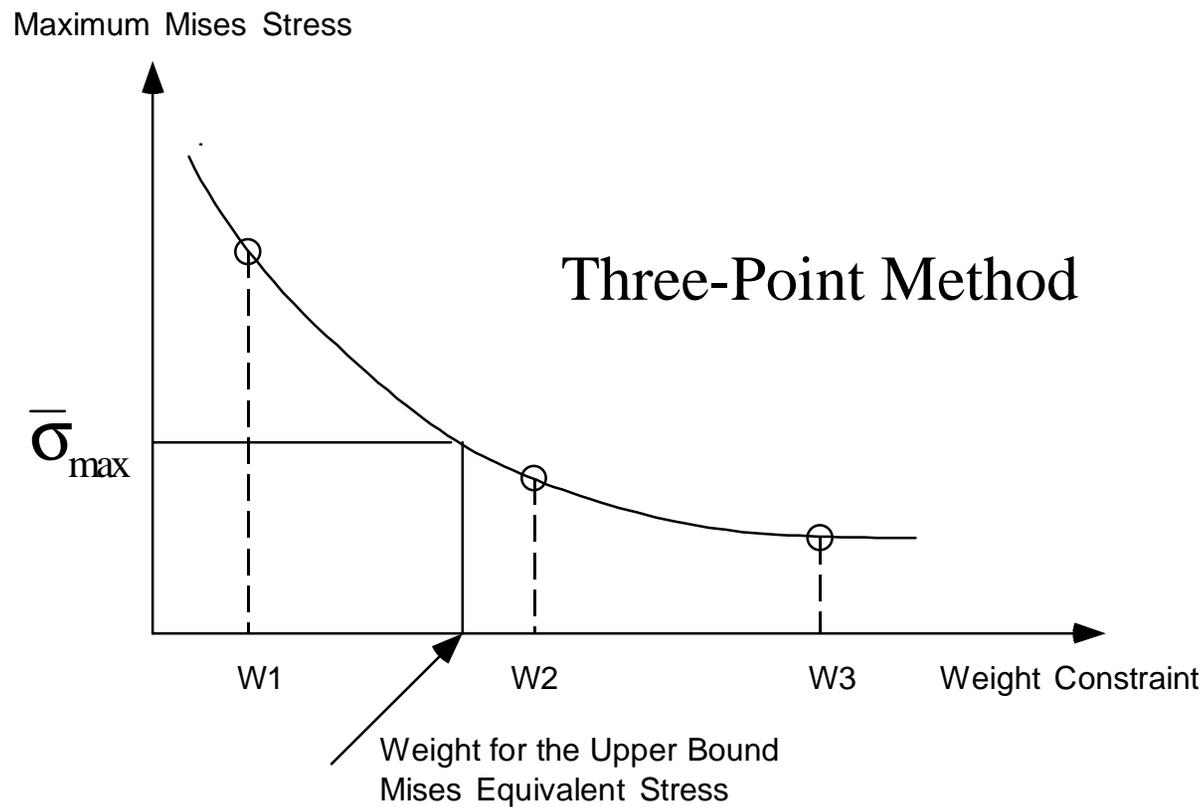


Typical Procedure #3

- (5) Repeat the above steps for two more different weight constraints W_2 & W_3
- (6) Obtain the maximum Mises stress and the mean compliance
- (7) Using the quadratic interpolation of
 - Maximum Mises Stresses & Weights
 - compute the weight for the allowable stress constraints



Determination of the Weight Constraint to enforce the stress constraint if exist





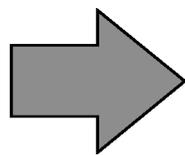
Extended Domain by χ_Ω

Internal Virtual Work

$$\int_{\Omega} \mathbf{e} \cdot \mathbf{b} \mathbf{v} \cdot \mathbf{g}^T \mathbf{E} \mathbf{e} \cdot \mathbf{b} \mathbf{u} \cdot \mathbf{g} d\Omega = \int_D \mathbf{e} \cdot \mathbf{b} \mathbf{v} \cdot \mathbf{g}^T \chi_\Omega \mathbf{E} \mathbf{e} \cdot \mathbf{b} \mathbf{u} \cdot \mathbf{g} dD$$

New Material Constants (Extended Elasticity Matrix)

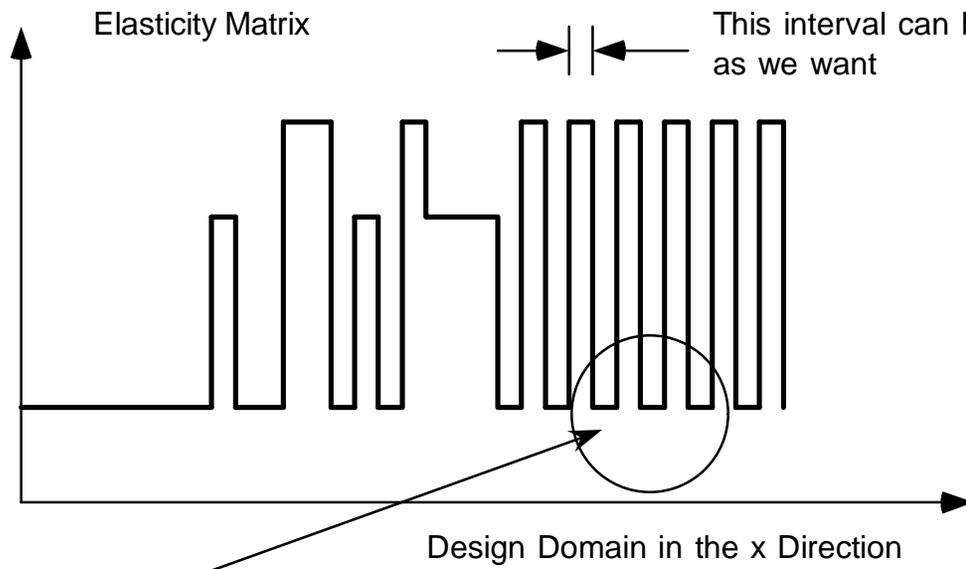
$\chi_\Omega \mathbf{E} \in L^\infty(D)$ is very discontinuous



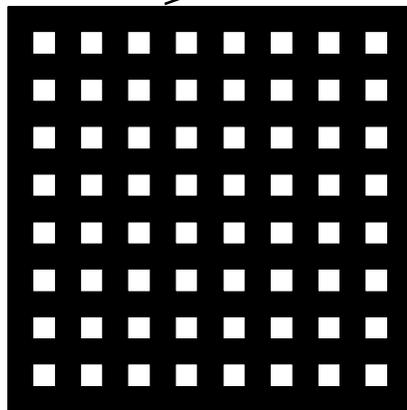
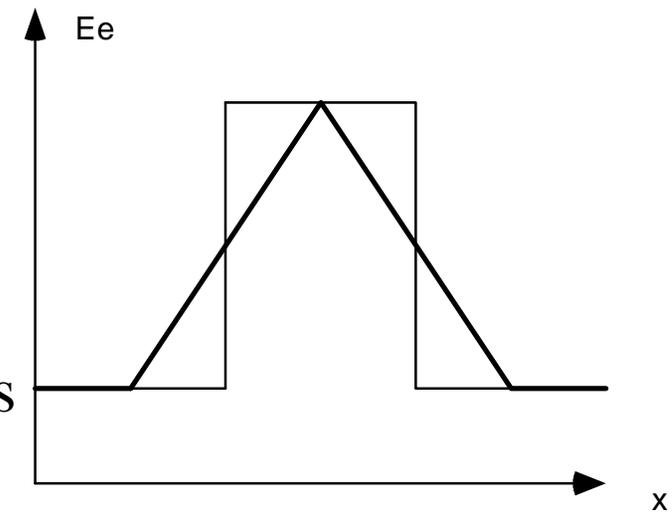
Impossible to take its derivative
that is, no design sensitivity analysis



Possible Approximation



Possibly Rapidly Varying



Introduce Two Scales at an arbitrary point

No Way to Approximate by a Standard Way with a distribution function



Relaxation

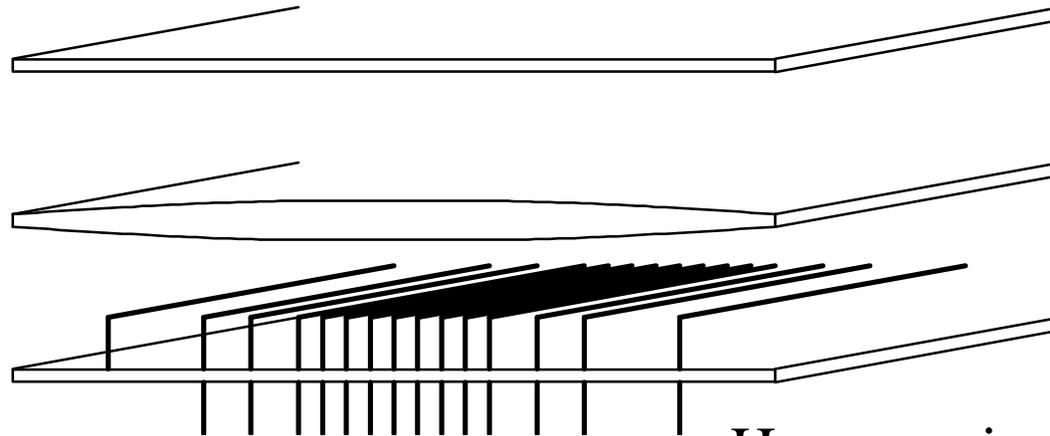
- Very Rapidly Varying Function $\chi_\Omega E$ cannot be approximated by a differentiable function of position x in the standard way

- Introduce the two scales $\left\| \begin{array}{l} x \\ y = \frac{x}{\varepsilon} \end{array} \right\|$ and the micro-scale perforation, and then $\chi_\Omega E$ is approximated by the homogenized average elasticity matrix E^H



Origin of the Idea

- G. Cheng and N. Olhoff
 - in plate thickness optimization
 - smoothly varying thickness is not optimum
 - optimum involves rapidly changing ribs



Homogenization is required



Mathematicians

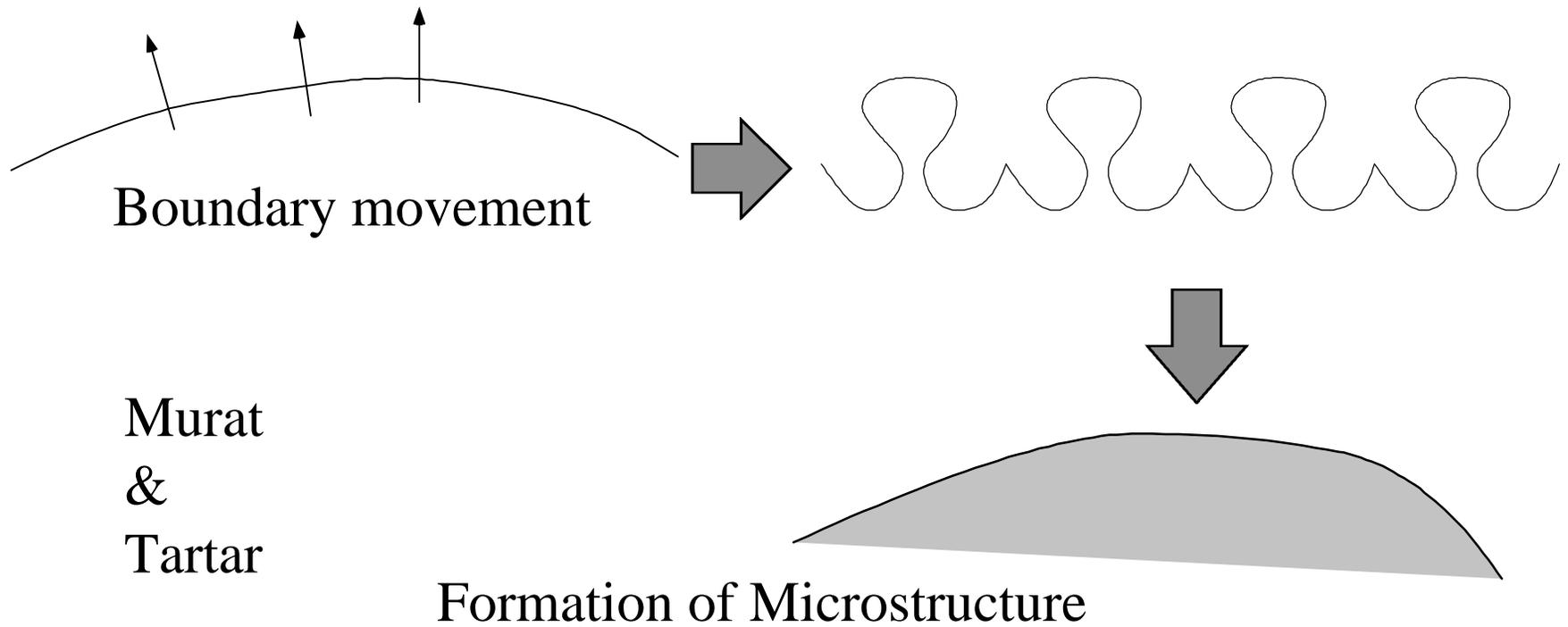
- Lurier, Cherkaev, and Fedrov (1981)
 - Notion of G-convergence that is in the specially designed average sense convergence
- Kohn and Strang (1984)
 - Microscale performance and specialized variational principles
- Murat and Tartar (1983)
 - Homogenization Theory from Hadamard Shape Design Problem



Hadamard v.s. Homogenization

Hadamard

Rapidly Varying



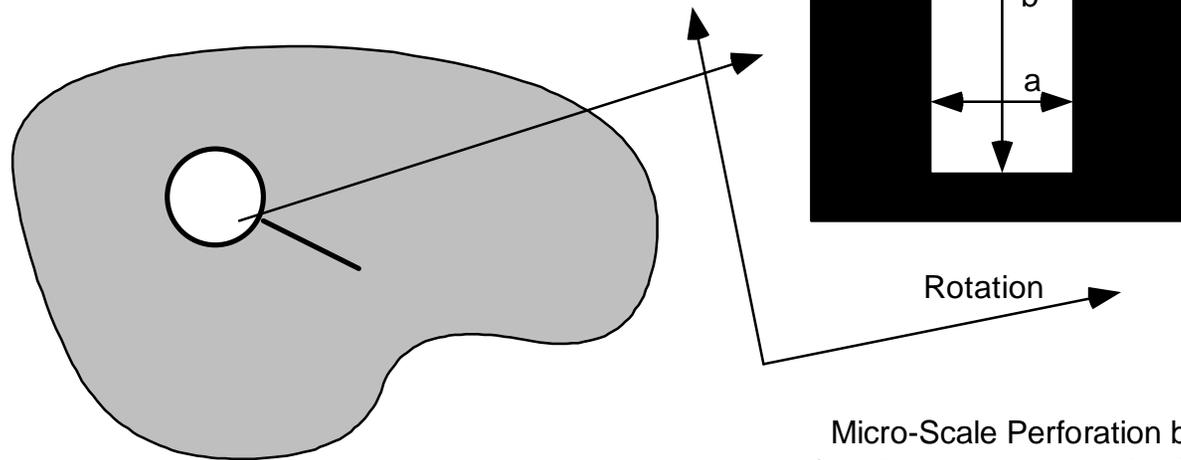
Murat
&
Tartar



Bendsoe & Kikuchi 1986

$a = b = 1 \Leftrightarrow$ void / hole

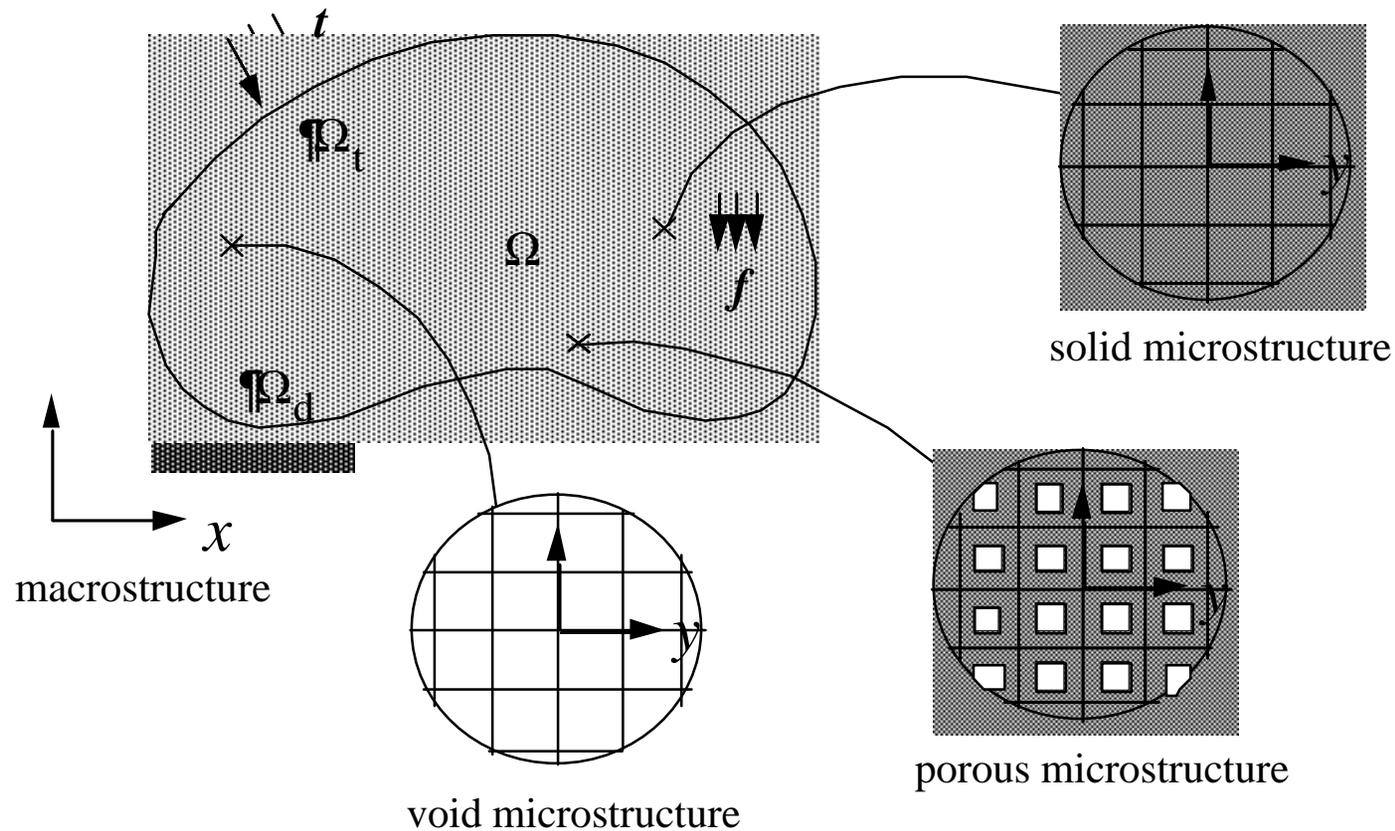
$a = b = 0 \Leftrightarrow$ solid structure



Design Variables are $\{a, b, \theta\}$ at every where



The Homogenization Design

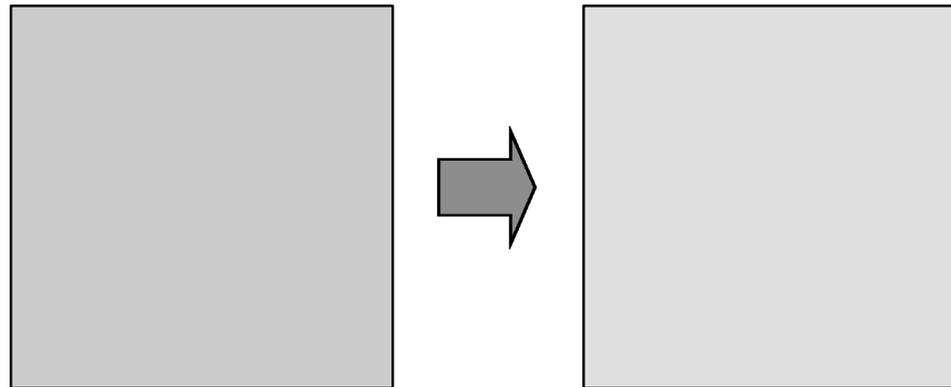




Why Homogenization ?

Small Scale Rapidly
Varying Heterogeneity

This idealization is regarded
as the homogenization in
theoretical mechanics



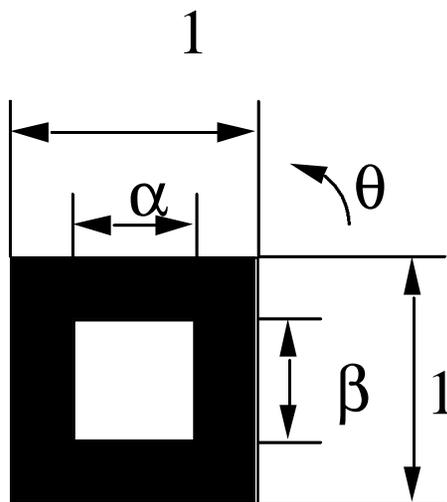
If the exact heterogeneity is used
in mechanics, we must introduce
so fine finite elements to represent
all the detail. This is a difficult task.

Equivalent Homogeneous
Material

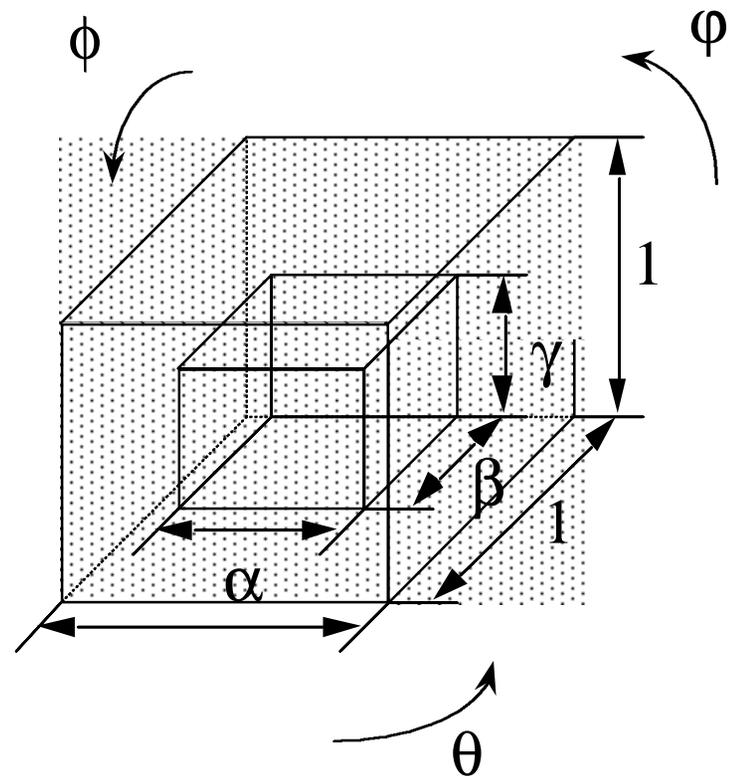


Microstructures

2D



3D





Many Choices

- The key feature is an approximation of the extended elasticity matrix $\chi_{\Omega} \mathbf{E} \in L^{\infty}(\mathcal{D})$
- There are infinitely many ways to approximate this by using
 - Generalized Porous Media Constitutive Equations (bio-mechanics, Geo-mechanics)
 - Power Law of Density/Elasticity Constants
 - Rank 1 & Rank 2 Orthotropic Materials
 - Others



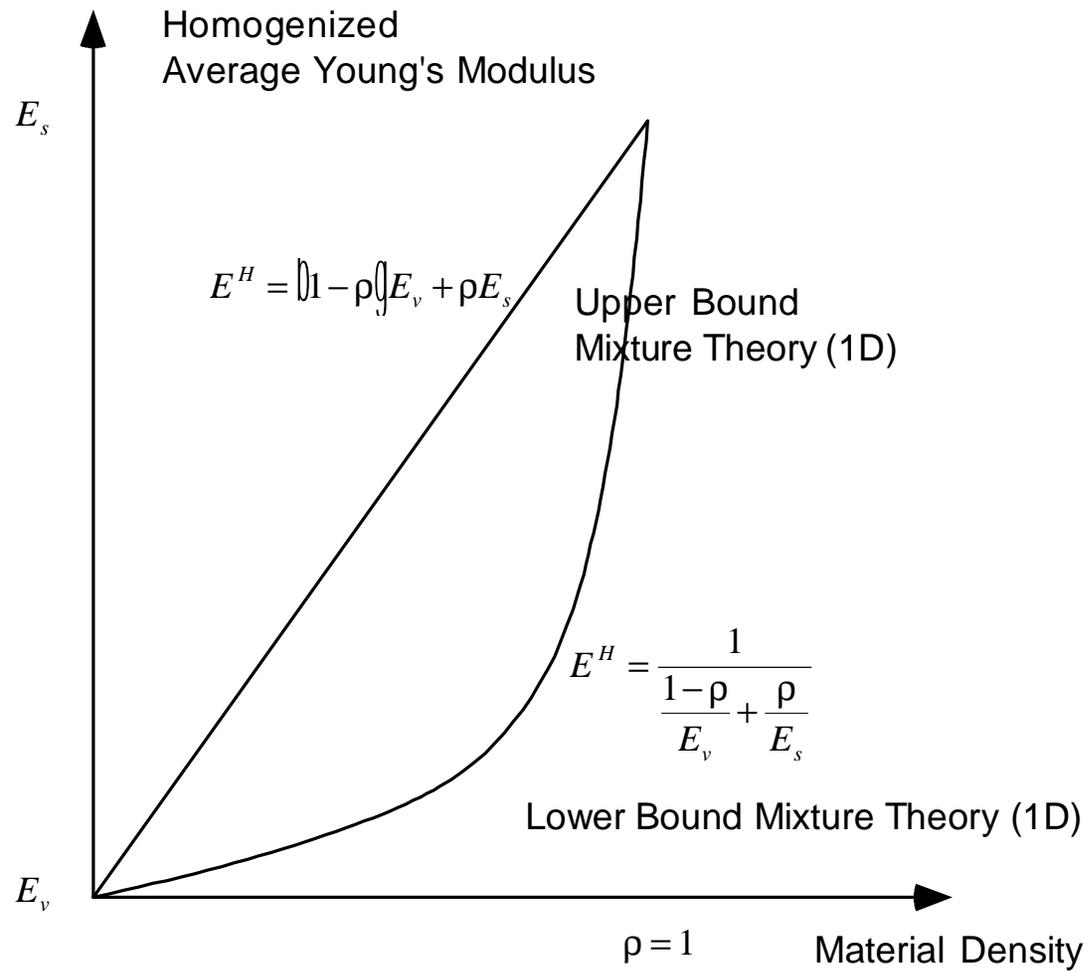
Power Law

Most popular approach at present (Meljck, Yang, ...)
Altair/OPTISTRUCT is now assuming this approach

$$\chi_{\Omega} \mathbf{E} \approx \rho^p \mathbf{E}$$

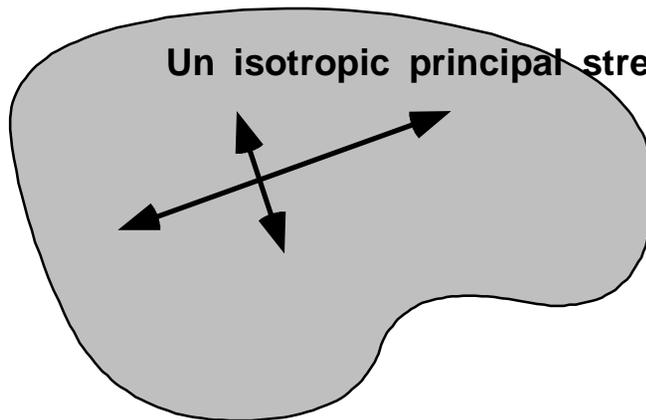
For $p=2$ or 4 , the design variable becomes the density ρ such that

$$\rho = \begin{cases} 1 & \text{if solid structure} \\ 0 & \text{if void / hole} \end{cases}$$



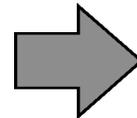


Short Coming of Power Law



Un isotropic principal stress distribution

Density approach can make only isotropic performance



This requires a lot of meshing to have reasonable layout

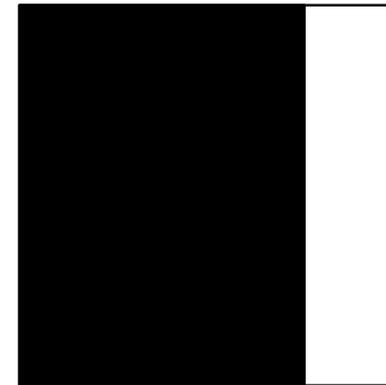
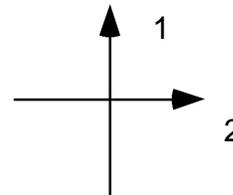
However, easy programing and handy design variable



Rank 1 & Rank 2 Materials

Rank 1 Lamination

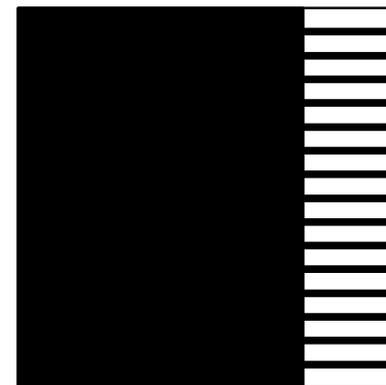
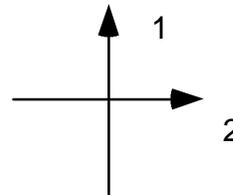
Zero E in 2
Zero Shear



Rank 2 Lamination

Zero Shear

Rectangular hole is
very close to Rank 2
Material





Elasticity Matrix

Rank 1 Material

$$E^H = \begin{pmatrix} (1-a)E_s & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{pmatrix}$$

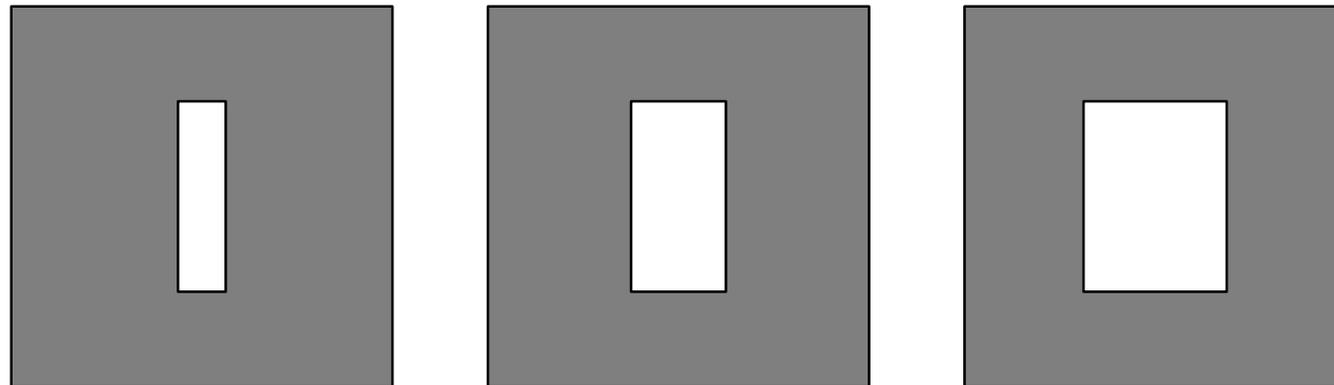
Rank 2 Material

$$E^H = \frac{E_s}{(1-\nu^2)(1-ab)} \begin{pmatrix} (1-\nu^2)(1-ab) & \nu(1-b) & 0 \\ \nu(1-b) & (1-a) + \nu^2(1-b) & 0 \\ 0 & 0 & 0 \end{pmatrix}$$



Advantage of Rank 2

Rank 2 elasticity matrix can be computed in the closed form while rectangular hole requires FE calculation





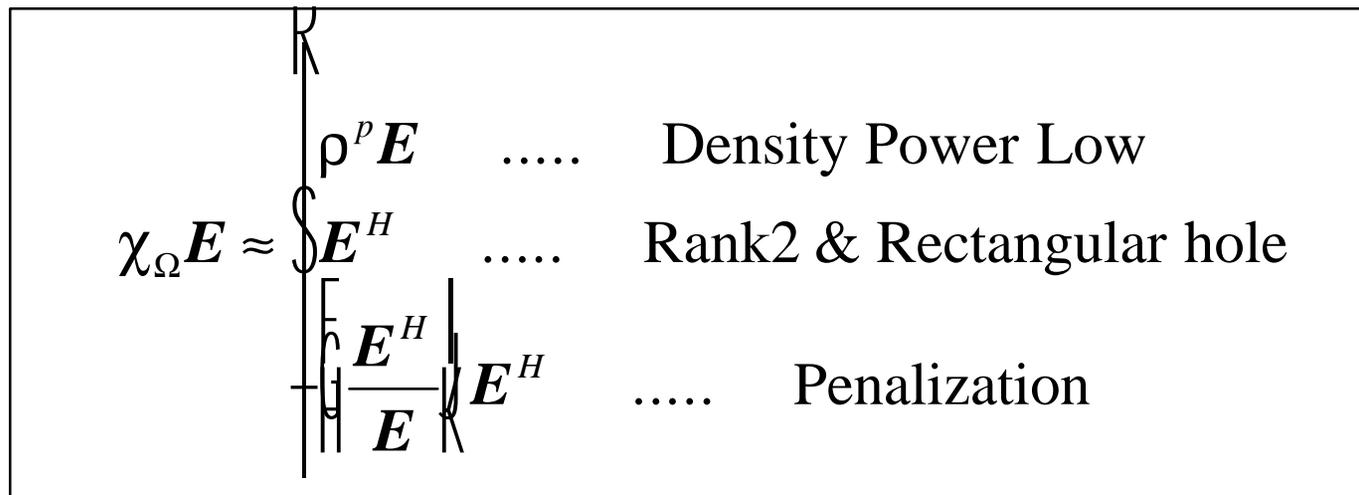
Rank 2 and Rectangular Hole

- Rank 2 material is regarded as the optimum orthotropic material (since its shear modulus becomes zero)
- Then this yield the stable optimum solution mathematically verified in the sense that FE mesh dependency cannot be observed
- However, this results considerably lots of gray perforated medium in the optimum



Penalization

- We would like to have clear segregation of solid and void portions to define precise shape and topology, that is
- No gray scale solution is desirable





Multi-Loading 1

Inequality Relation

$$\begin{aligned} \max_{\{a,b,\theta\}_{i=1,2,\dots,i_{\max}}} \min_{i=1,2,\dots,i_{\max}} \frac{1}{2} a_i \langle \mathbf{b} \mathbf{u}_i, \mathbf{u}_i \rangle - f_i \langle \mathbf{b} \mathbf{u}_i \rangle &= \max_{\{a,b,\theta\}_{i=1,2,\dots,i_{\max}}} \min_{i=1,2,\dots,i_{\max}} -\frac{1}{2} a_i \langle \mathbf{b} \mathbf{u}_i, \mathbf{u}_i \rangle \\ &\leq \max_{\{a,b,\theta\}} -\frac{1}{2} a_m \langle \mathbf{b} \mathbf{u}_m, \mathbf{u}_m \rangle = \max_{\{a,b,\theta\}} \frac{1}{2} a_m \langle \mathbf{b} \mathbf{u}_m, \mathbf{u}_m \rangle - f_m \langle \mathbf{b} \mathbf{u}_m \rangle \end{aligned}$$

Minimum Principle to the I-th Load (Equilibrium)

$$\frac{1}{2} a_i \langle \mathbf{b} \mathbf{u}_i, \mathbf{u}_i \rangle - f_i \langle \mathbf{b} \mathbf{u} \rangle = \min_{\mathbf{v}_i} \frac{1}{2} a_i \langle \mathbf{b} \mathbf{v}_i, \mathbf{v}_i \rangle - f_i \langle \mathbf{b} \mathbf{v} \rangle$$



Multiple Loading 2

Formation of a Single Objective Function

$$a_m(u_m, u_m) = \int_D \max_{i=1, \dots, i_{\max}} (u_i)^T \chi_{\Omega} E \max_{i=1, \dots, i_{\max}} (u_i) dD$$

$$f_m(u_m) = \int_D \max_{i=1, \dots, i_{\max}} (u_i)^T \chi_{\Omega} E s_0 dD + \int_D \max_{i=1, \dots, i_{\max}} u_i^T \chi_{\Omega} \rho b dD + \dots$$

Approximated Design Problem

$$\begin{aligned} & \max_{(a, b, \theta)} \min_v \frac{1}{2} a_m(v, v) - f_m(v) \\ & \text{subject to} \\ & \int_D \rho dD \leq W_0 \end{aligned}$$



Alternate

Weighted Sum Approach

$$\begin{array}{l} \max_{\{a, b, \theta\}} \\ \text{subject to} \\ \int_D \rho dD \leq W_0 \end{array} \sum_{i=1}^{i_{\max}} w_i \min_v \frac{1}{2} a_i \|v, v\| - f_i \|v\|$$

Diaz and Bendsoe

OPTISHAPE can do both ways by user's choice



Lagrangian

Optimization Problem

$$\max_{\{a, b, \theta\}} \min_{\nu} \frac{1}{2} a \langle \nu, \nu \rangle - f \langle \nu, \nu \rangle$$

subject to

$$\int_D \rho dD \leq W_0$$

Taylor & Prager in 1967

Lagrangian

$$L = \frac{1}{2} a \langle \nu, \nu \rangle - f \langle \nu, \nu \rangle - \lambda \left(\int_D \rho dD - W_0 \right)$$

First Variation

$$\begin{aligned} \delta L = & a \langle \nu, \delta \nu \rangle - f \langle \delta \nu, \nu \rangle - \delta \lambda \left(\int_D \rho dD - W_0 \right) \\ & + \int_D \left[\langle \nu, \nu \rangle^T \frac{\partial \chi_{\Omega} \mathbf{E}}{\partial a} \langle \nu, \nu \rangle - \lambda \frac{\partial \rho}{\partial a} \right] \delta a dD \\ & + \int_D \left[\langle \nu, \nu \rangle^T \frac{\partial \chi_{\Omega} \mathbf{E}}{\partial b} \langle \nu, \nu \rangle - \lambda \frac{\partial \rho}{\partial b} \right] \delta b dD \\ & + \int_D \left[\langle \nu, \nu \rangle^T \frac{\partial \chi_{\Omega} \mathbf{E}}{\partial \theta} \langle \nu, \nu \rangle \right] \delta \theta dD \end{aligned}$$



Optimality Condition

Equilibrium

$$a \delta v, \delta v \int = f \delta v \int \quad \forall \delta v$$

Weight Constraint : Kuhn-Tucker Condition

$$\lambda \int_D \rho dD - W_0 = 0 \quad , \quad \lambda \leq 0 \quad , \quad \int_D \rho dD - W_0 \leq 0$$

Optimality Condition

$$\int_D \bar{a} - a \int e b v g^T \frac{\partial \chi_\Omega \mathbf{E}}{\partial a} e b v g - \lambda \frac{\partial \rho}{\partial a} dD \geq 0 \quad , \quad 0 \leq \forall \bar{a} \leq 1$$

$$\int_D \bar{b} - b \int e b v g^T \frac{\partial \chi_\Omega \mathbf{E}}{\partial b} e b v g - \lambda \frac{\partial \rho}{\partial b} dD \geq 0 \quad , \quad 0 \leq \forall \bar{b} \leq 1$$

$$e b v g^T \frac{\partial \chi_\Omega \mathbf{E}}{\partial \theta} e b v g = 0$$



Optimality Criteria Method 1

$$\int_D \left(\bar{a} - a \right) \left(\mathbf{e} \cdot \mathbf{v} \right)^T \frac{\partial \chi_\Omega \mathbf{E}}{\partial a} \mathbf{e} \cdot \mathbf{v} - \lambda \frac{\partial \rho}{\partial a} dD \geq 0, \quad 0 \leq \forall \bar{a} \leq 1$$

$$\Leftrightarrow \mathbf{e} \cdot \mathbf{v} \left(\mathbf{v} \cdot \mathbf{e} \right)^T \frac{\partial \chi_\Omega \mathbf{E}}{\partial a} \mathbf{e} \cdot \mathbf{v} - \lambda \frac{\partial \rho}{\partial a} = 0 \quad \text{if } a \neq 0 \text{ and } a \neq 1$$

$$\Leftrightarrow \frac{\mathbf{e} \cdot \mathbf{v} \left(\mathbf{v} \cdot \mathbf{e} \right)^T \frac{\partial \chi_\Omega \mathbf{E}}{\partial a} \mathbf{e} \cdot \mathbf{v}}{\lambda \frac{\partial \rho}{\partial a}} = 1 \quad \text{if } a \neq 0 \text{ and } a \neq 1$$

$$\Rightarrow a^{b_{k+1}} = a^{b_k} \frac{\mathbf{e} \cdot \mathbf{v} \left(\mathbf{v} \cdot \mathbf{e} \right)^T \frac{\partial \chi_\Omega \mathbf{E}}{\partial a} \mathbf{e} \cdot \mathbf{v} \left(a^{b_k}, b^{b_k}, \theta^{b_k} \right)}{\lambda^{b_k} \frac{\partial \rho}{\partial a} \left(a^{b_k}, b^{b_k}, \theta^{b_k} \right)}, \quad 0 < \alpha \leq 1$$



Optimality Criteria Method 2

With design constraint $0 \leq a \leq 1$

$$a^{b_{k+1}g} = \max \left\{ 0, \min \left\{ 1, a^{b_k g} \frac{e^{e_{\nu^{b_k g} j^T} \frac{\partial \chi_{\Omega} E}{\partial a} e^{a^{b_k g}, b^{b_k g}, \theta^{b_k g} j} e^{e_{\nu^{b_k g} j}} \right\}^{\alpha} \right\} \right\} \frac{\lambda^{b_k g} \frac{\partial \rho}{\partial a} e^{a^{b_k g}, b^{b_k g}, \theta^{b_k g} j}}{\lambda^{b_k g} \frac{\partial \rho}{\partial a} e^{a^{b_k g}, b^{b_k g}, \theta^{b_k g} j}}$$

Algorithm of the optimality criteria method is very similar with the fully stressed design

Choice of Parameter

$$\alpha \approx 0.75$$



Lagrange Multiplier

$\lambda^{b_k g}$ is computed by the bisection method

$$\int_D \rho \theta \left[a^{b_k g}, b^{b_k g} \right] dD = W_0$$

based the implicit function theorem

Volume constraint is always saturated.
This is correct, but for eigenvalue related problems. this is not true.



Optimum Angle : Pedersen

Noting that

$$\begin{aligned}
 e \mathbf{b}_v \mathbf{g}^T \mathbf{b}_{\chi_\Omega} \mathbf{E} \mathbf{g} e \mathbf{b}_v \mathbf{g} &= s \mathbf{b}_v \mathbf{g}^T \mathbf{C}_{\chi_\Omega} \mathbf{E}^{-1} \mathbf{h} s \mathbf{b}_v \mathbf{g} \\
 &= \begin{vmatrix} \sigma_1 & \sigma_2 \\ \mathbf{0} & \mathbf{0} \end{vmatrix} \mathbf{T} \mathbf{b}_\phi \mathbf{g}^T \mathbf{C}_{\chi_\Omega} \mathbf{E}^{-1} \mathbf{h} \mathbf{T} \mathbf{b}_\phi \mathbf{g} \begin{matrix} R \sigma_1 \\ \psi \\ \sigma_2 \\ \psi \\ 0 \end{matrix}
 \end{aligned}$$

we have

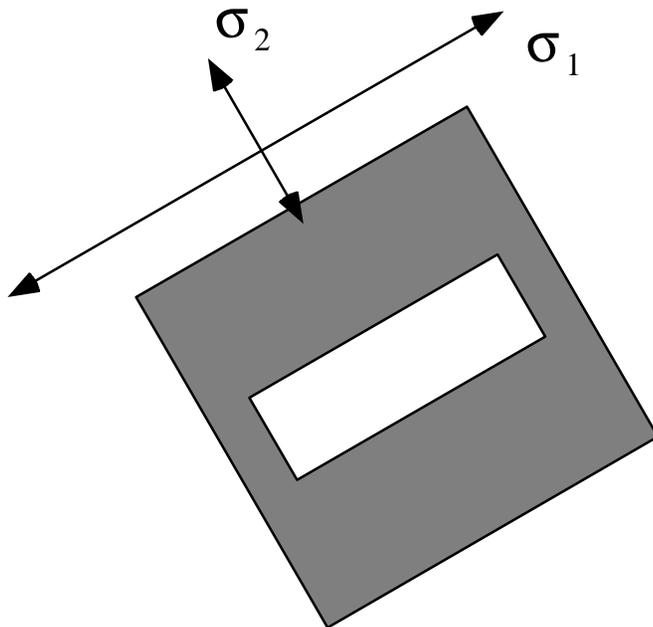
$$e \mathbf{b}_v \mathbf{g}^T \frac{\partial \chi_\Omega \mathbf{E}}{\partial \theta} e \mathbf{b}_v \mathbf{g} = 0 \quad \Rightarrow \quad \frac{\partial}{\partial \theta} \mathbf{T} \mathbf{b}_\phi \mathbf{g}^T \mathbf{C}_{\chi_\Omega} \mathbf{E}^{-1} \mathbf{h} \mathbf{T} \mathbf{b}_\phi \mathbf{g} = 0 \quad \Leftrightarrow \quad \phi = 0$$

Optimum angle is the one for the principal stress !



Engineering Idea

Optimality Condition can be explained as



Rectangular hole should be aligned in the principal stress direction

Large hole can be assumed in the small principal stress direction

Small hole must be placed in the large principal stress direction



Elaboration

Using the relation

$$\rho = 1 - ab \Rightarrow a = \frac{1 - \rho}{b}$$

Habor
Jog
Bendsoe

we can change the design variable

$$\{a, b, \theta\} \Rightarrow \{\rho, b, \theta\}$$

and then we can show

$$\begin{aligned} & \max_{\{\rho, b, \theta\}} \min_{\nu} \frac{1}{2} a \{b, \nu\} - f \{b, \nu\} \\ \Leftrightarrow & \max_{\rho} \min_{\nu} \max_{\{b, \theta\}} \frac{1}{2} a \{b, \nu\} - f \{b, \nu\} \end{aligned}$$

Solve b and angle analytically, then apply optimality criteria method only to the density



Monotonic Convergence

- Optimality criteria method is monotonically converging to a local optima
- The local optima obtained may be strongly dependent of the initial condition
 - Uniformly Biased Initial Condition in OPTISHAPE
- The local optima may depend on the FE mesh



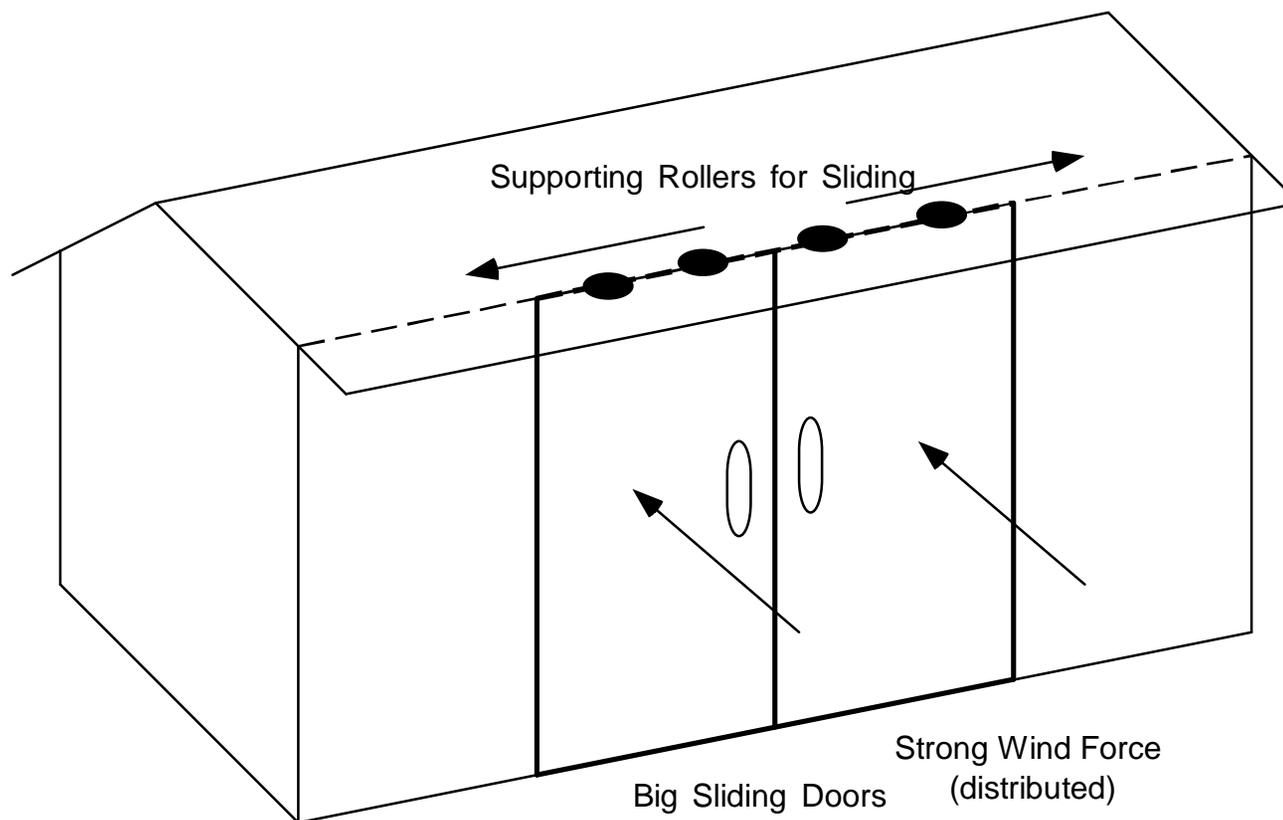
Many Problems

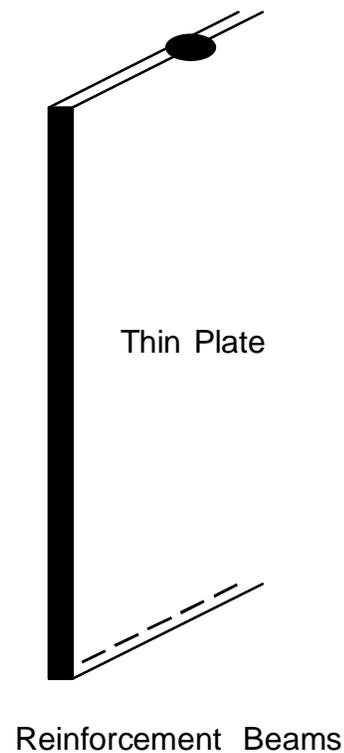
but

It is so powerful !



Excercise #11





Find a pattern of reinforcement
of a thin plate subjected to a
strong wind force by using
OPTISHAPE



Topology Optimization -2- OPTISHAPE

1. Extension of HMD
2. Free Vibration Problem
3. Frequency Response Problem
4. Buckling Problem
5. Flexible Bodies



Extension of OPTISHAPE

- Free Vibration Problem
 - Maximization of lowest frequency
 - Maximization of the distance of two frequencies
 - Inverse frequency identification problem
- Frequency Response Problem
- Buckling Problem for Stability
- Flexible Multi-Body Design
- Material Microstructure Design



Eigenvalue Problem

- Maximizing the lowest eigenvalue

$$\max_{\{a,b,\theta\}} \lambda_{\min} \quad \mathbf{Ku} = \lambda \mathbf{Mu}$$

- Several eigenvalues are crashing while optimization is performed

One eigenvalue with n number of eigenvectors
----- Ultimate Optima



Free Vibration

Discrete Free Vibration Problem

$$\mathbf{M} \frac{d^2 \mathbf{x}}{dt^2} + \mathbf{K} \mathbf{x} = \mathbf{0}$$

Separation of variable

$$\mathbf{x}(t) = \exp(i\omega t) \mathbf{u}$$

$$-\omega^2 \mathbf{M} \exp(i\omega t) \mathbf{u} + \mathbf{K} \exp(i\omega t) \mathbf{u} = \mathbf{0}$$

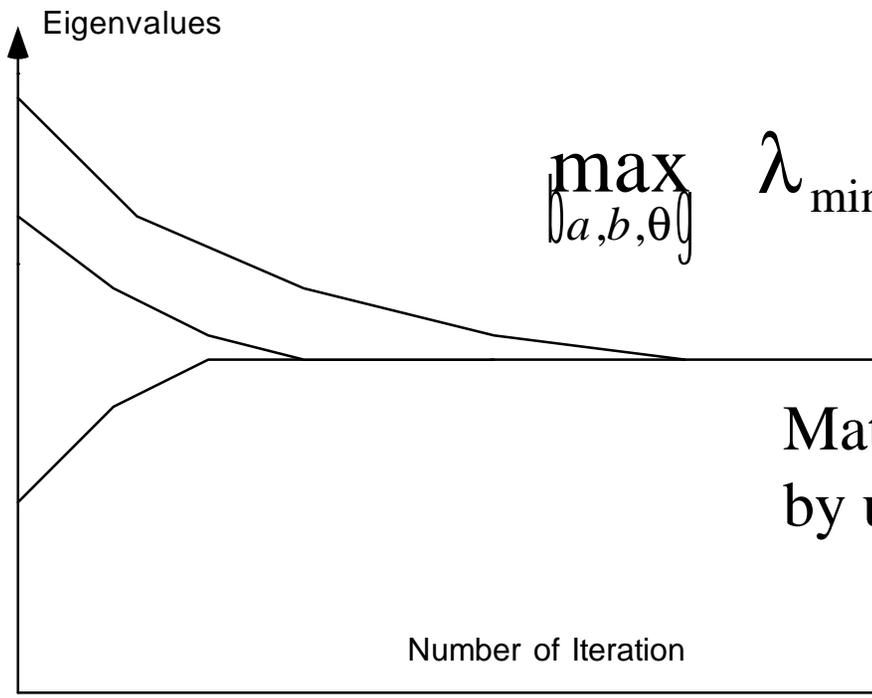
$$(-\omega^2 \mathbf{M} \mathbf{u} + \mathbf{K} \mathbf{u}) \exp(i\omega t) = \mathbf{0}$$

$$-\omega^2 \mathbf{M} \mathbf{u} + \mathbf{K} \mathbf{u} = \mathbf{0}$$

$$\mathbf{K} \mathbf{u} = \omega^2 \mathbf{M} \mathbf{u} = \lambda \mathbf{M} \mathbf{u}$$



Modification by Ma, Cheng, Kikuchi



$$\max_{\{a,b,\theta\}} \lambda_{\min} \Rightarrow \max_{\{a,b,\theta\}} \sum_{i=1}^{i_{\max}} w_i \lambda_{\min i}$$

Mathematics can be constructed by using sub-differentiability

----- Bendsoe and Rodrigues

i_{\max} = number of crashing eigenvalues

w_i = weights , $i = 1, \dots, i_{\max}$



Further Modification

$$\max_{\{a,b,\theta\}} \lambda_{\min} \Rightarrow \max_{\{a,b,\theta\}} \sum_{i=1}^{i_{\max}} w_i \lambda_{\min i} + \sum_{j=1}^{j_{\max}} z_j \lambda_l$$

$\lambda_{\min i}$ = minimum crashing eigenvalue

λ_l = 2nd, 3rd....., higher eigenvalues

We shall maximize not only the lowest frequency
but also several eigenvalues at once

.... yields sub-optima, but make sense in engineering



Maximization of Distance

$$\max_{\{a,b,\theta\}} \sum_{j=1}^{j_{\max}} w_j \lambda_j - \sum_{i=1}^{i_{\max}} w_i \lambda_i$$

$$\sum_{j=1}^{j_{\max}} w_j \lambda_j = \text{higher eigenvalue}$$

$$\sum_{i=1}^{i_{\max}} w_i \lambda_i = \text{lower eigenvalue}$$



Inverse Frequency Problem

$$\max_{\{a,b,\theta\}} \sum_{i=1}^{i_{\max}} w_i \left\| \frac{1}{2} (\lambda_i - \lambda_{i}^{\text{target}}) \right\|^p$$

Identify the structural configuration so that it has specified eigenvalues for the first small set of eigenvalues



Frequency Response Problem

$$\max_{\{a,b,\theta\}} \min_v \frac{1}{2} \mathbf{v}^T \mathbf{K} \mathbf{v} - \frac{1}{2} \omega^2 \mathbf{v}^T \mathbf{M} \mathbf{v} - \mathbf{v}^T \mathbf{b}$$

\mathbf{K} = stiffness matrix

\mathbf{M} = mass matrix

ω = specified frequency

\mathbf{b} = excited force with ω frequency

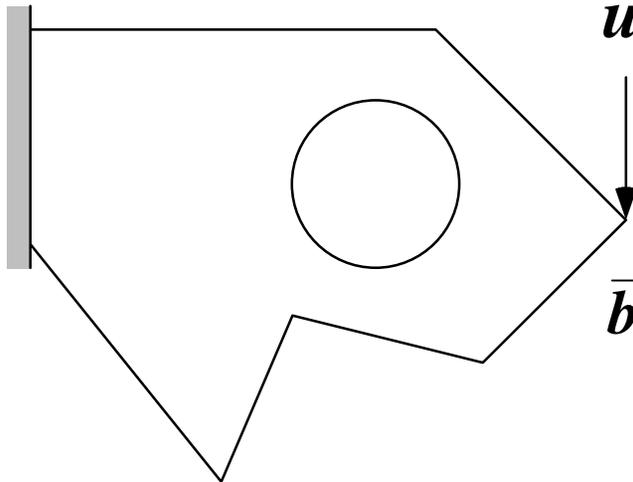


Discrete Equilibrium

$$(K - \omega^2 M)u = b$$

$$\bar{b}(t) = \exp(i\omega t)b = \text{exciting force}$$

$$\bar{u}(t) = \exp(i\omega t)u = \text{dynamical response}$$



$$\bar{b}(t) = \exp(i\omega t)b$$



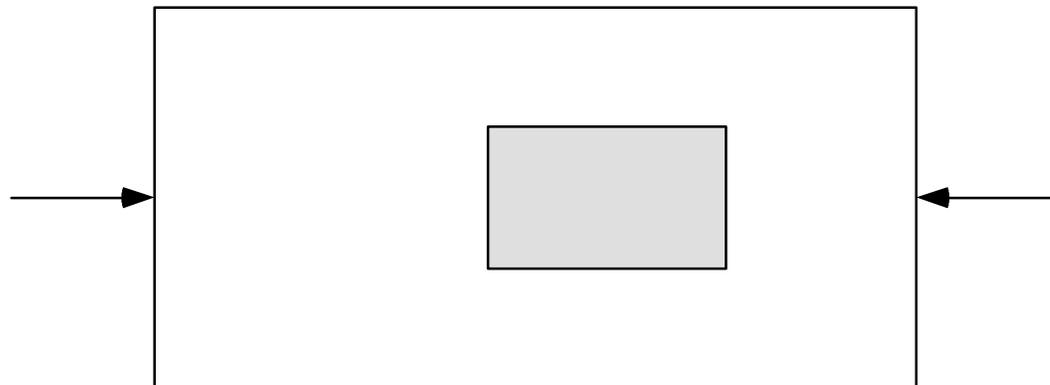
Buckling Problem

$$\max_{\{a,b,\theta\}} \lambda_{\min} \quad \mathbf{K}u = \lambda \mathbf{K}_G u$$

\mathbf{K} = stiffness matrix

\mathbf{K}_G = geometric stiffness matrix

Rodrigues and Gedes



possibility of local buckling modes



Linear Combinations of the above cases

OPTISHAPE



OPTISHAPE in Practice

- OPTISHAPE is currently integrated into SDRC/I-DEAS. In future an icon of OPTISHAPE will be added into I-DEAS option menu
- OPTISHAPE design models can be developed by MSC/PATRAN and the input and output data are fully compatible with MSC/NASTRAN



The screenshot displays the HSC/PATRAN Version 6.0 software interface. The main window shows a 3D finite element model of a beam, rendered in a wireframe style. The beam is oriented along the X-axis, with the Y and Z axes also visible. The model is composed of a grid of elements. The software interface includes a menu bar (File, Group, Viewport, Viewing, Display, Preferences, Tools, Insight Control, Help) and a toolbar with various icons for file operations, viewing, and analysis. The status bar at the bottom left shows the file name '6-6.db - default_viewport - default_group - Entity'. The 'Results Attributes' dialog box is open on the right side, showing settings for 'Element Scalar Labeling'. The 'Format Type' is set to 'Integer' with 4 significant digits. The 'Scale Factor' is set to 1.0. The 'Deformation' section has 'Show Deformation' checked. The 'Results Attributes' dialog box also includes options for 'Show Result Values', 'Show Min/Max Values', 'Show Undeformed Entities', 'Show Result Title', and 'Show Vector Results'. The 'Reset Graphics' button is visible at the bottom of the dialog box. The bottom of the screen shows a terminal window with the following text: 'gin 42: xwd -out beam1sh.xwd', 'gin 43: xwd -in beam1sh.xwd', and 'gin 44: xwd -out beam1.xwd'. The terminal window is titled 'hpterm' and has a '10' in the top left corner. The 'HITACHI' logo is visible in the bottom left corner of the terminal window.



OPTISHAPE to CAD/CAM

- OPTISHAPE can translate the image data of the optimum topology/configuration into a STL file (as well as SLC file in near future) so that smoothed 3D surfaces can be plotted to show the concept of the design (NC Link)
- OPTISHAPE can produce a set of wire frame models of sliced models perpendicular to a specified direction (CAD Link)



OPTISHAPE

More in Practice



Current Development Research

OPTISHAPE

is

constantly enhanced



Flexible Body Design

compliant mechanism design

by

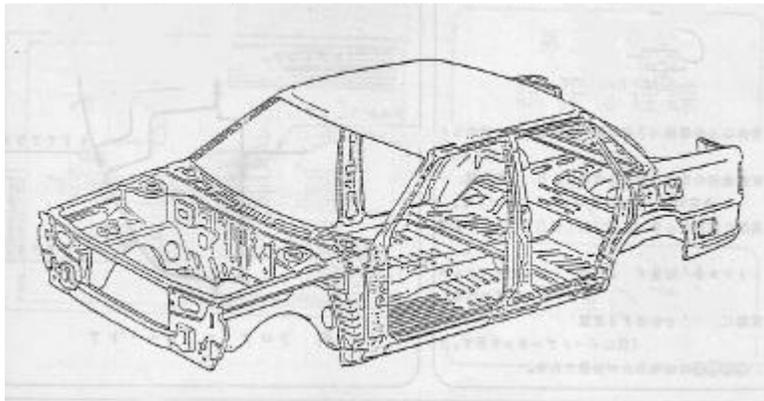
Mary Frecker & Shinji Nishiwaki

Shinji Ejima

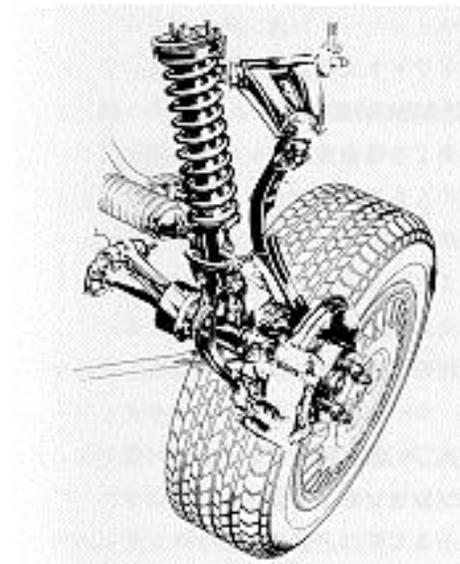


Mechanical Design

Structure Design



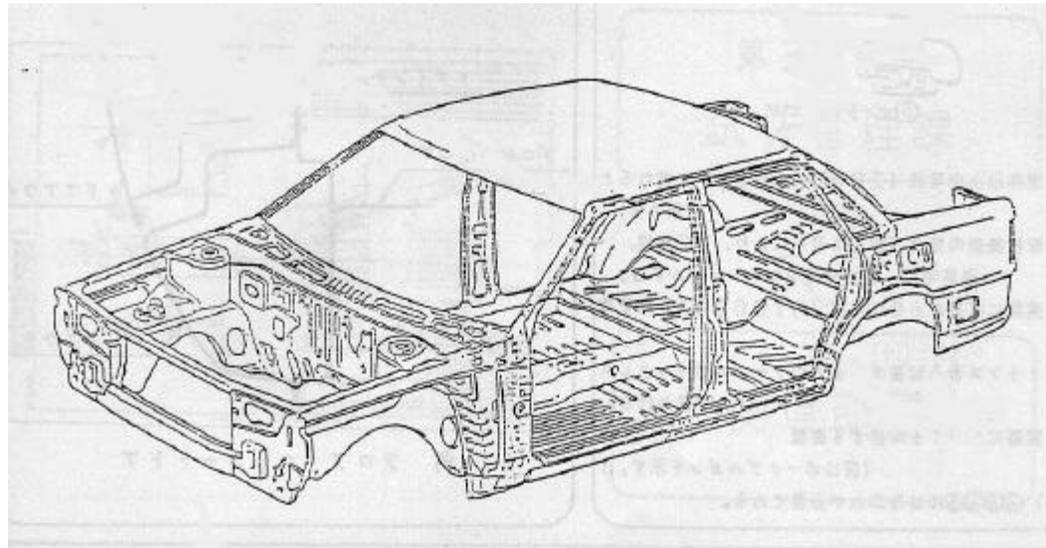
Mechanism Design



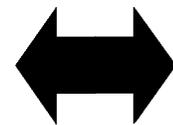
New Design Based on Flexibility



Structure Design



Flexible Structure

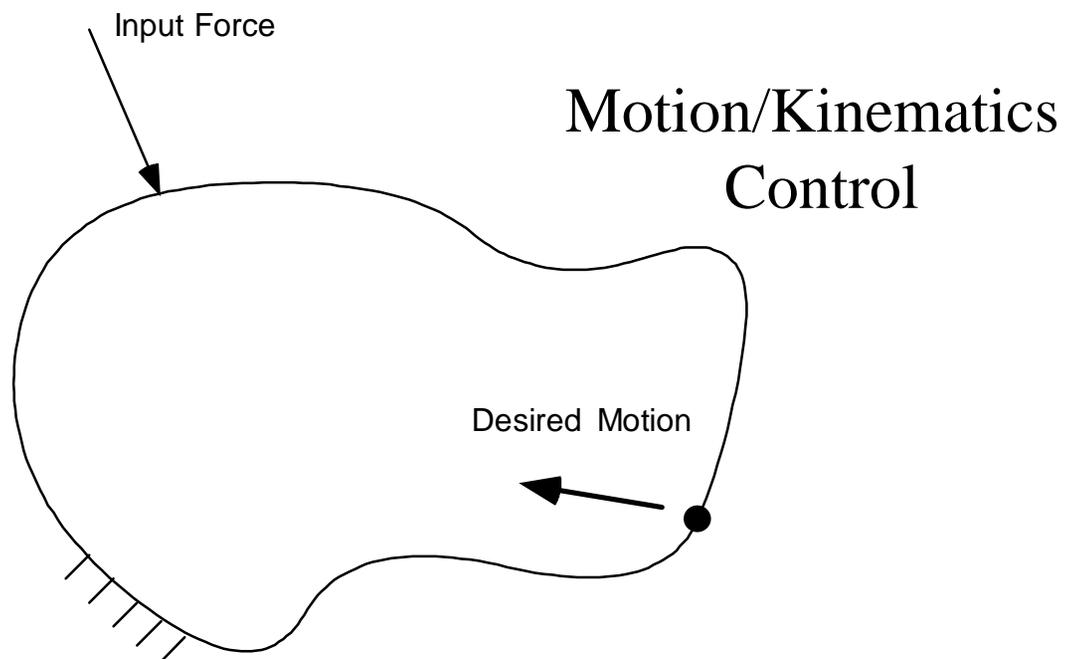


Stiff Structure



Flexible Body Design

- Design a structure that moves to the specified direction as much as possible when input forces are given

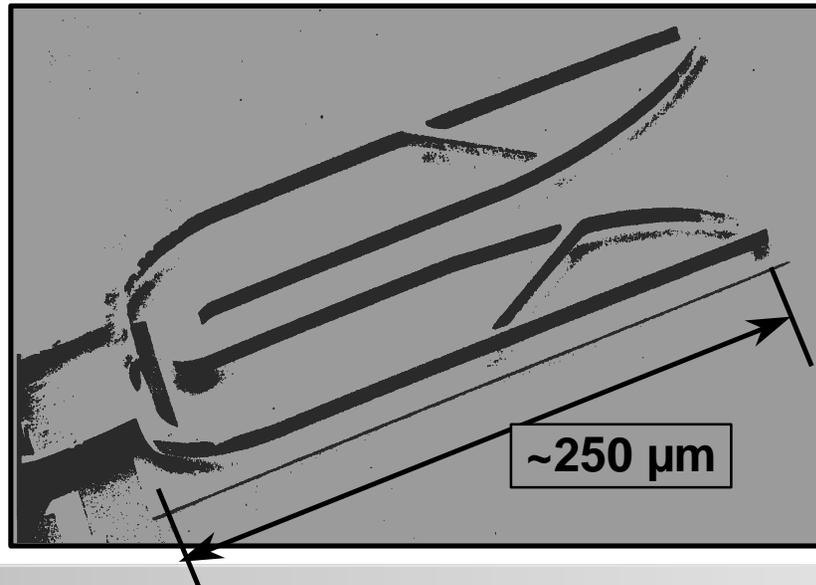




Compliant Mechanism

- MEMS (Micro Electro Mechanical System)
- Basic Ideas & Clues for
Rigid Link Mechanism Design

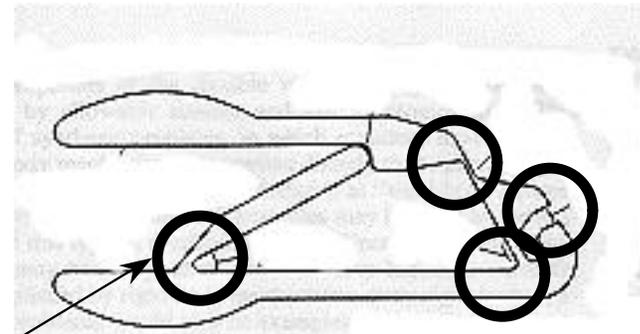
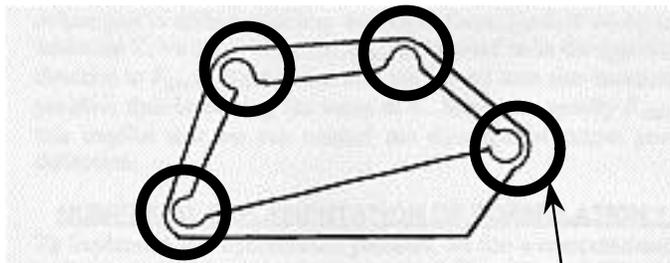
Microcompliant crimping
mechanism





Kinematic Synthesis

Based on traditional rigid body kinematics

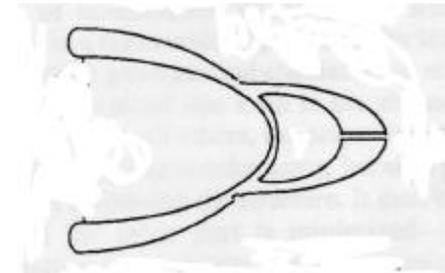
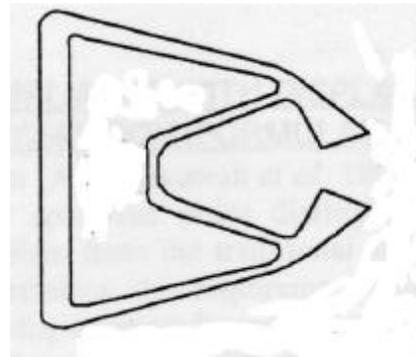
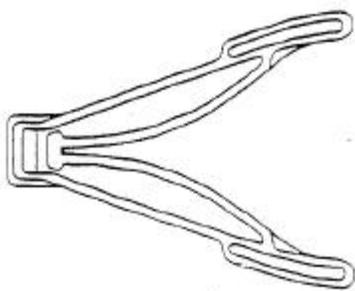


Lumped compliance (Pivot)

I. Her and A. Midha (1986),
L. L. Howell and A. Midha (1994), (1996)



Continuum Synthesis



Distributed compliance

O. Sigmund (1995), (1996)

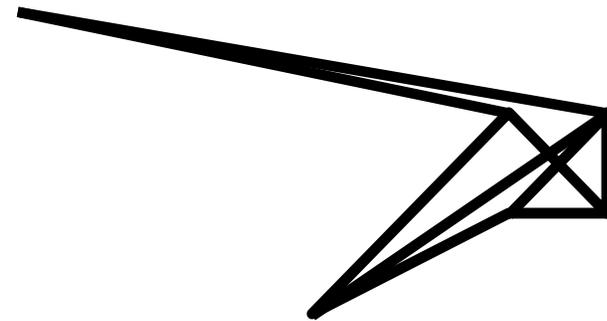
U. D. Larsen, O. Sigmund and S. Bouswstra (1996)



Design of Flexible Structures

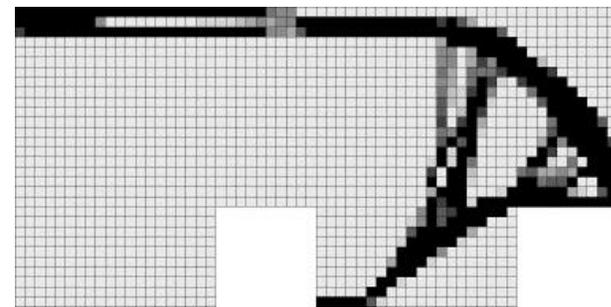
- Truss Approach

Configuration by Trusses



- Continuous Approach

Distribution of Materials

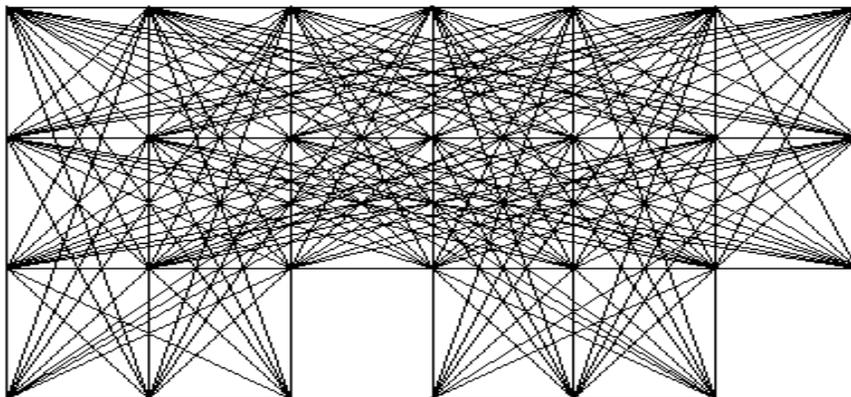




Truss Approach

Based on Ground Structure Design

→ Compliant Mechanism Design

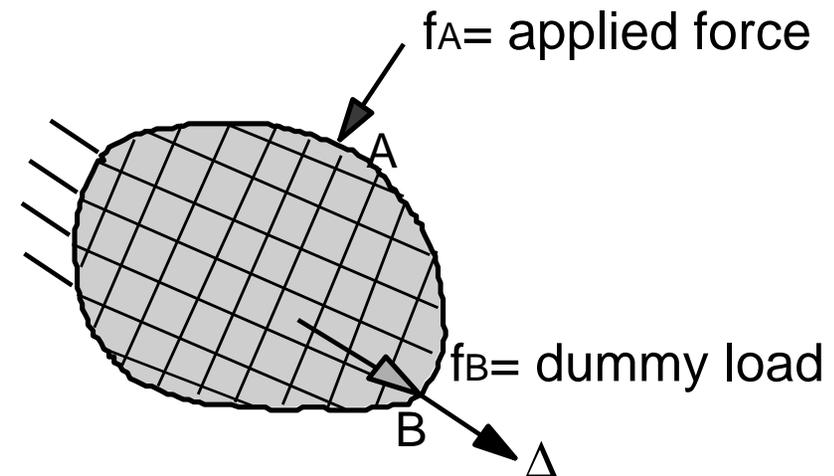


By Mary I. Frecker



Problem Statement 1

Kinematic Function



$$\max (\text{mutual energy}) \Leftrightarrow \max (\mathbf{v}_B^T \mathbf{K} \mathbf{u}_A)$$

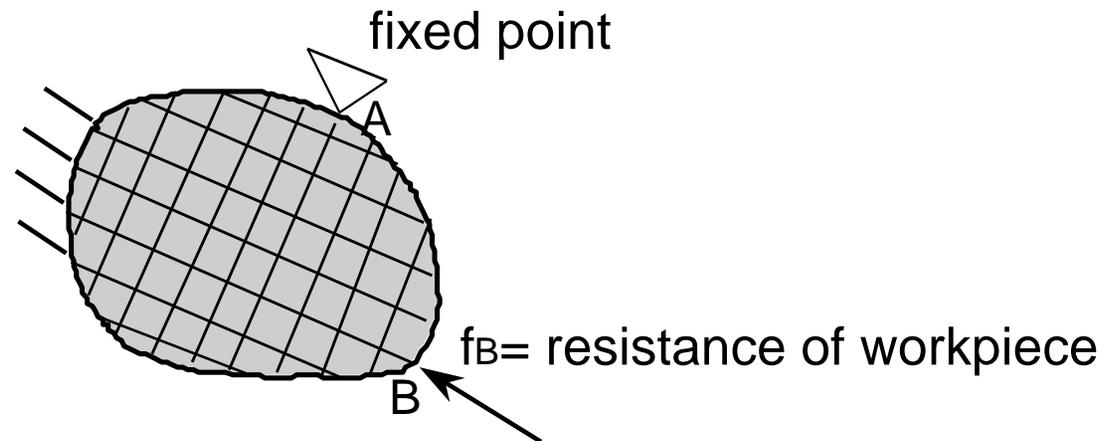
$$\text{subject to : } \mathbf{K} \mathbf{u}_A = \mathbf{f}_A$$

$$\mathbf{K} \mathbf{v}_B = \mathbf{f}_B$$



Problem Statement 2

Structural Function



$$\min(\textit{strain energy}) \Leftrightarrow \min (\mathbf{u}_B^T \mathbf{K} \mathbf{u}_B)$$
$$\text{subject to} : \mathbf{K} \mathbf{u}_B = -\mathbf{f}_B$$



Multicriteria Optimization

$$\max \left[\frac{\textit{mutual energy}}{\textit{strain energy}} \right] \Leftrightarrow \max \left[\frac{\mathbf{v}_B^T \mathbf{K} \mathbf{u}_A}{\mathbf{u}_B^T \mathbf{K} \mathbf{u}_B} \right]$$

subject to : $\mathbf{K} \mathbf{u}_A = \mathbf{f}_A$

$$\mathbf{K} \mathbf{v}_B = \mathbf{f}_B$$

$$\mathbf{K} \mathbf{u}_B = -\mathbf{f}_B$$

total resource constraint

lower and upper bounds



A Simple Design Problem

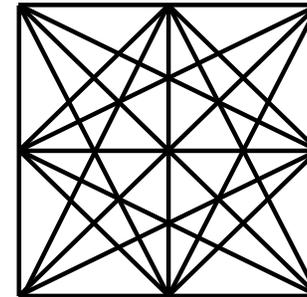
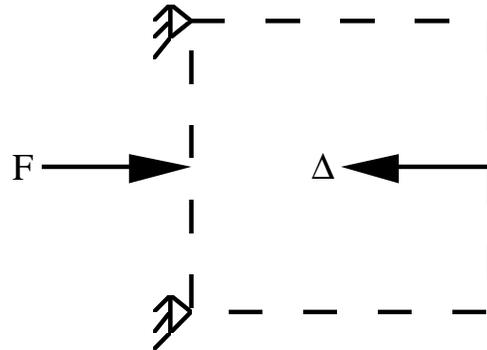


Figure 5a. Design Problem. Figure 5b. Initial Guess.

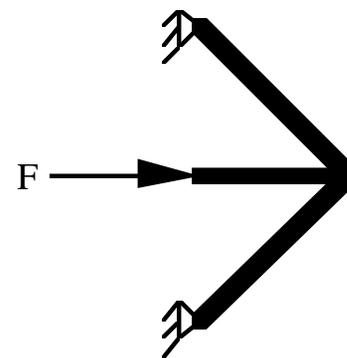
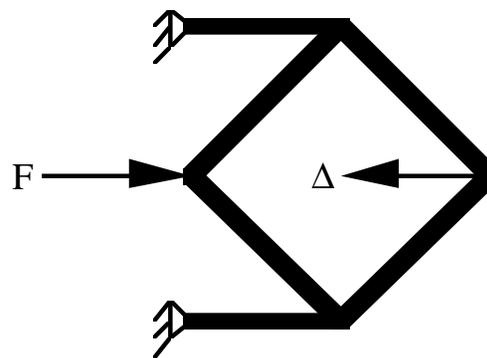
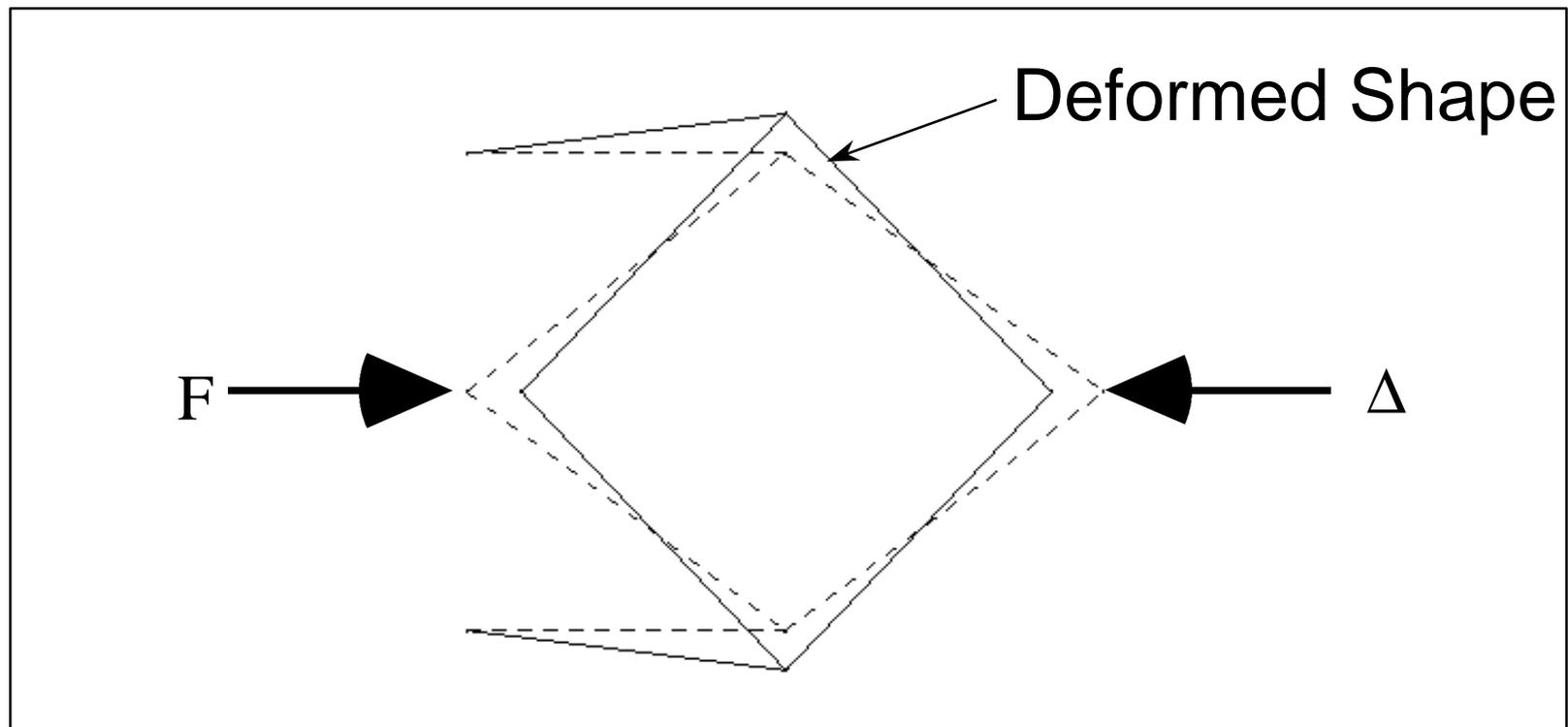


Figure 5c. Compliant Mechanism Solution.

Figure 5d. Stiffest Structure Solution.



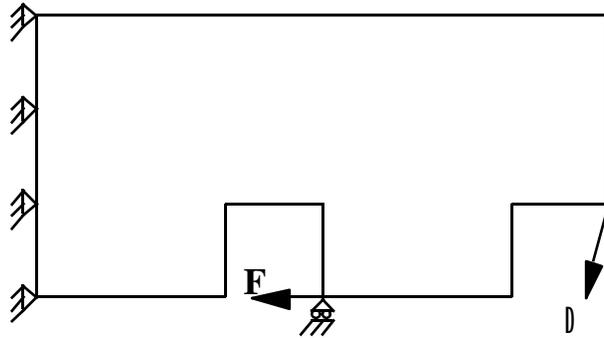
Verification of Function



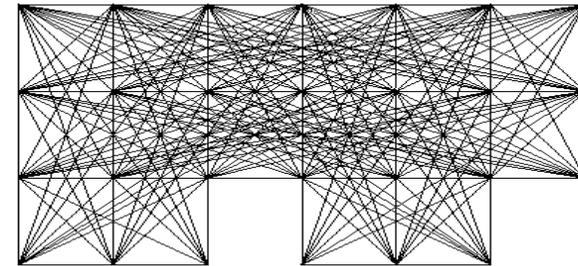
This can satisfy the original objective



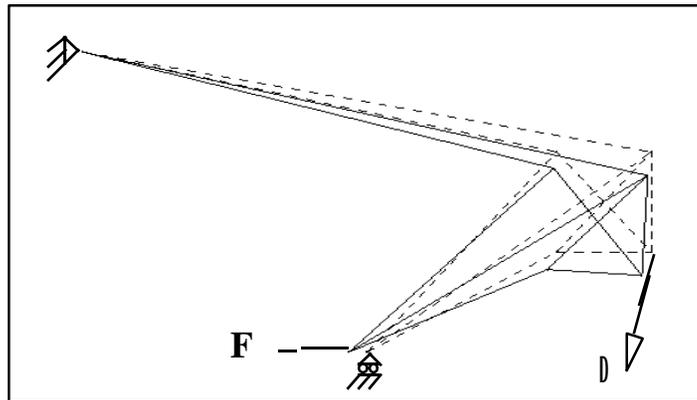
Compliant Gripper



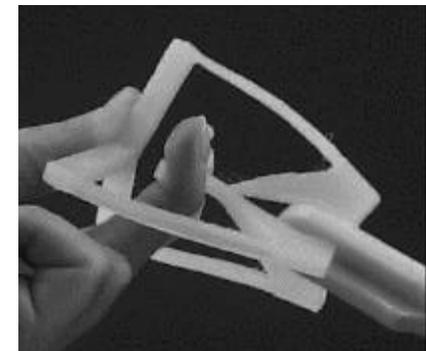
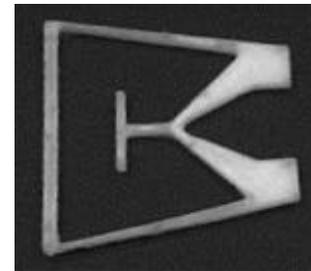
Design Problem



Initial Guess



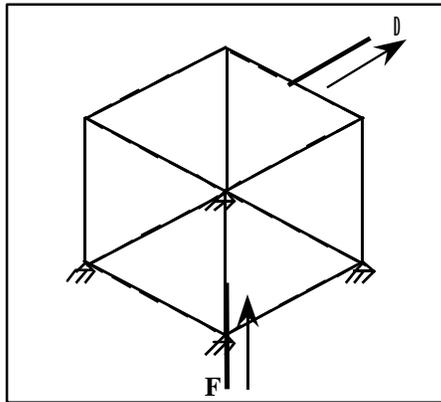
Solution and Finite Element Model
Computational Mechanics Laboratory



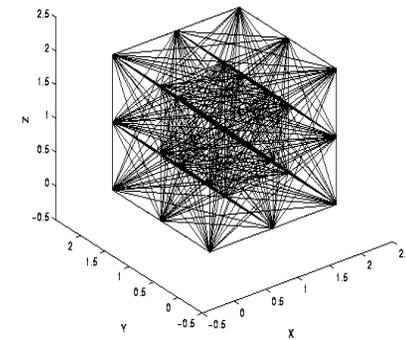
Compliant Grippers



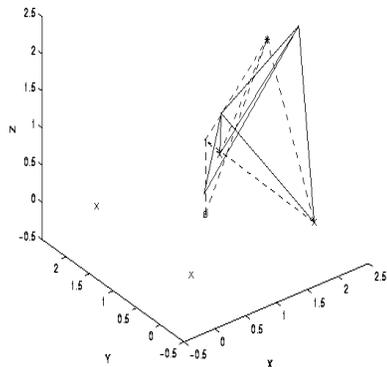
3D Compliant Gripper



Design Problem



Initial Guess



Solution and Finite Element Model
Computational Mechanics Laboratory



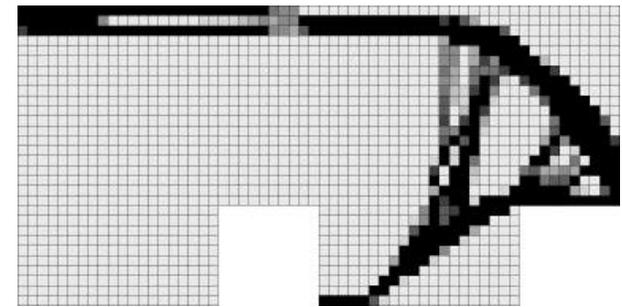
Three Dimensional
Compliant Mechanism



Continuous Approach

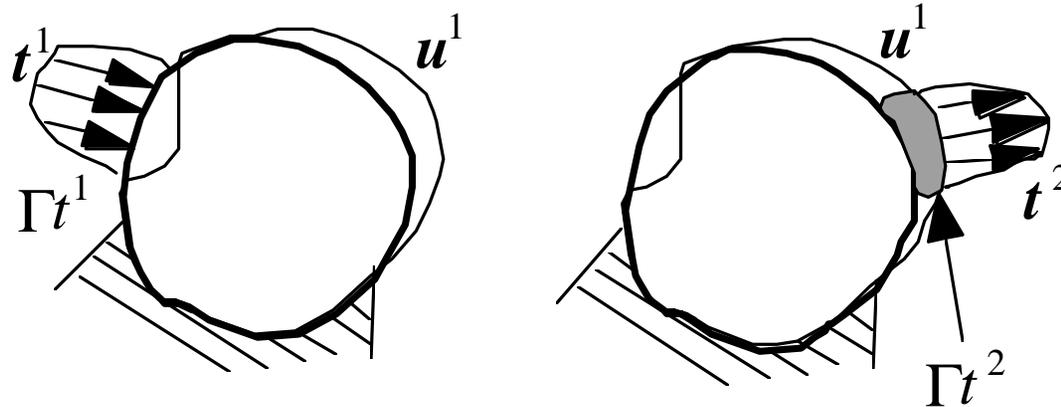
Based on Homogenized Design Method

- Fixed Grid / Voxel Mesh Method
- Homogenization Method





Formulation of Flexibility



Maximize $L^2(\mathbf{u}^1) = \int_{\Gamma t^2} \mathbf{t}^2 \cdot \mathbf{u}^1 d\Gamma$ (Mutual Mean Compliance)

➔ Flexibility at Γt^2

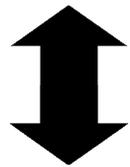
Minimize $L^1(\mathbf{u}^1) = \int_{\Gamma t^1} \mathbf{t}^1 \cdot \mathbf{u}^1 d\Gamma$ (Mean Compliance)

➔ Stiffness at Γt^1



Multicriteria Optimization

Flexibility \rightarrow Max *Mutual Mean Compliance*



Trade Off

Stiffness \rightarrow Min \sum *Mean Compliance*

\rightarrow Compromise Solutions



Multi-Objective Functions

(1)

$$\text{Max } \frac{\textit{Mutual Mean Compliance}}{\sum \textit{Mean Compliance}}$$

(2)

$$\text{Max } w \text{Log}(\textit{Mutual Mean Compliance}) - (1-w)\text{Log}(\sum \textit{Mean Compliance})$$

where w is a weighting Coefficient

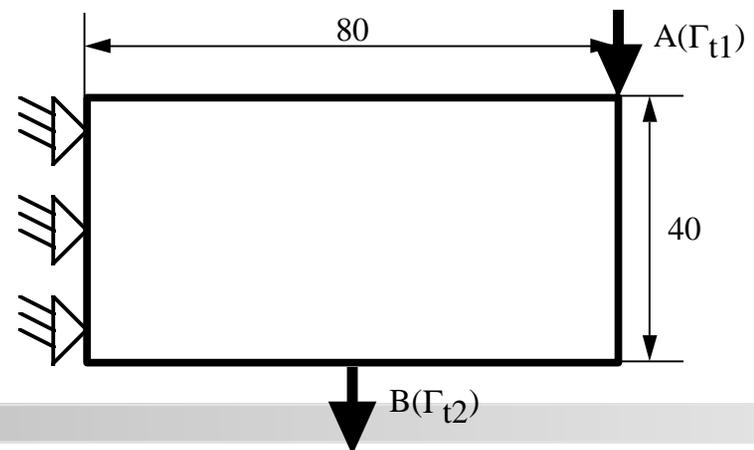


Weighting Method vs. Ratio Formulation

- Max w_1 *Mutual Mean Compliance*
- $(1-w_1)\sum$ *Mean Compliance*
where w_1 is a weighting Coefficient

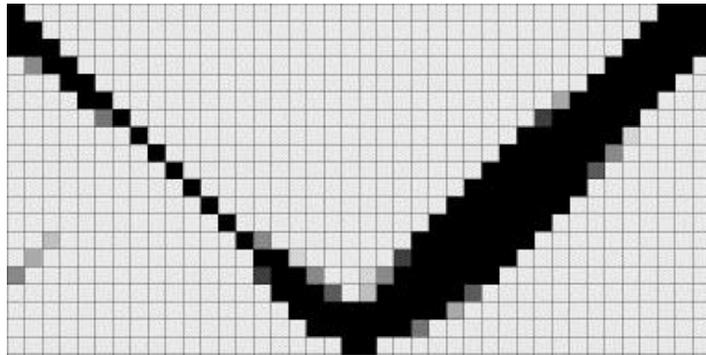
- Max $\frac{\textit{Mutual Mean Compliance}}{\sum \textit{Mean Compliance}}$

Example

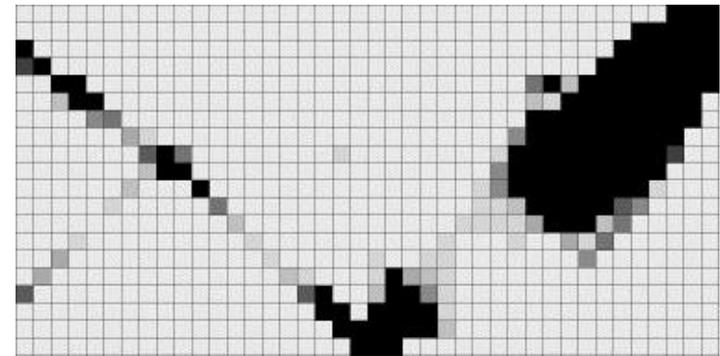




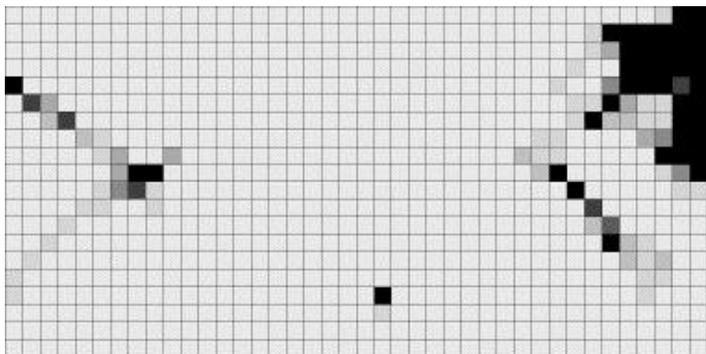
Weighting Method vs. Ratio Formulation



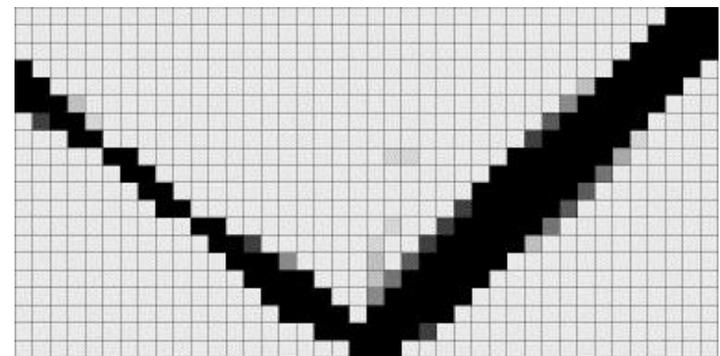
$w_1 = 0.001$



$w_1 = 0.01$



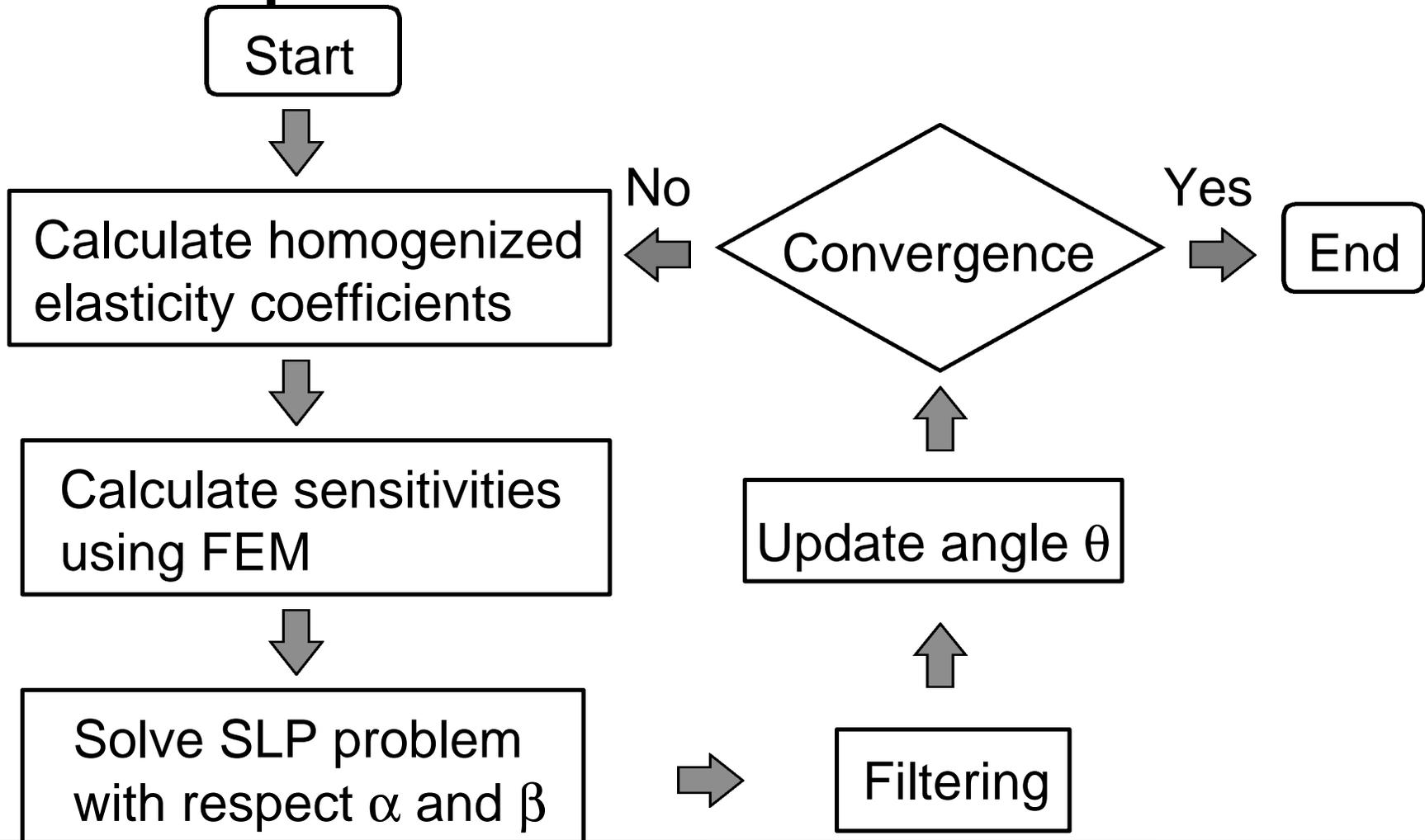
$w_1 = 0.1$



Ratio Formulation



Optimization Procedure





SLP vs. OC

- Sequential Linear Programming (SLP)
 - Linear Approximation + Simplex Method
 - Slow Convergence
 - Easy Implementation for Any Objective Functions
- Optimality Criteria Method (OC)
 - KKT-Conditions + Heuristics
 - Quick Convergence
 - Difficult to Construct Heuristics for General Objective Functions



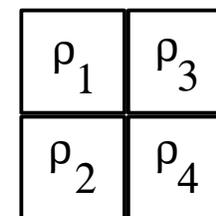
Checkerboard Pattern

$$\begin{array}{ccc}
 \rho_1 = \rho_4 = 1 & \begin{array}{|c|c|} \hline \blacksquare & \square \\ \hline \square & \blacksquare \\ \hline \end{array} & > & \begin{array}{|c|c|} \hline \square & \square \\ \hline \square & \square \\ \hline \end{array} & \rho_1 = \rho_2 = 1/2 \\
 \rho_2 = \rho_3 = 0 & & & & \rho_3 = \rho_4 = 1/2
 \end{array}$$

Artificially stiffer

Filtering scheme

$$\begin{array}{ll}
 \bar{\rho}_1 = \frac{1}{4}(3\rho_1 + \rho_2 + \rho_3 - \rho_4) & \bar{\rho}_2 = \frac{1}{4}(\rho_1 + 3\rho_2 - \rho_3 + \rho_4) \\
 \bar{\rho}_3 = \frac{1}{4}(\rho_1 - \rho_2 + 3\rho_3 + \rho_4) & \bar{\rho}_4 = \frac{1}{4}(-\rho_1 + \rho_2 + \rho_3 + 3\rho_4)
 \end{array}$$



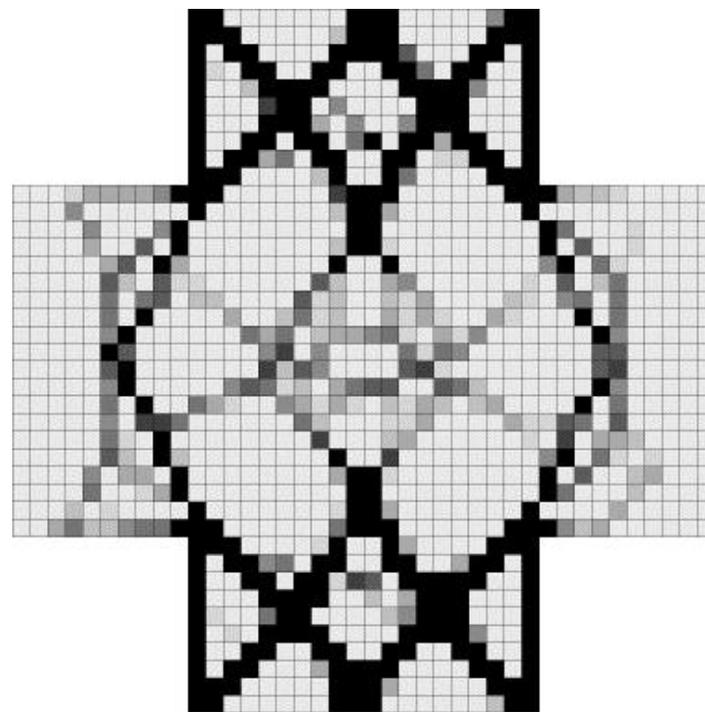
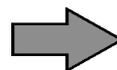
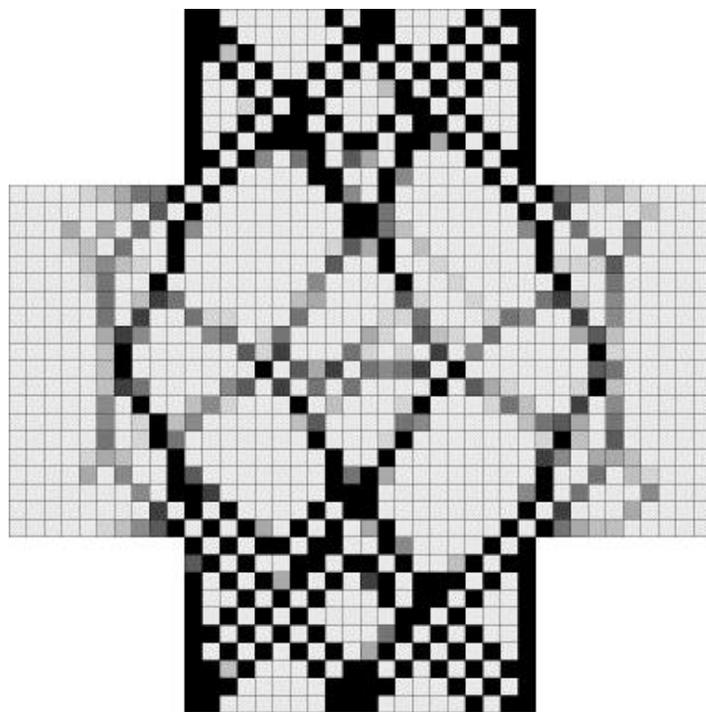
One group

where $\rho_i = 1 - \alpha_i \beta_i$ ($i = 1, \dots, 4$)



Elimination of Checkerboard

Eigen-frequency problem

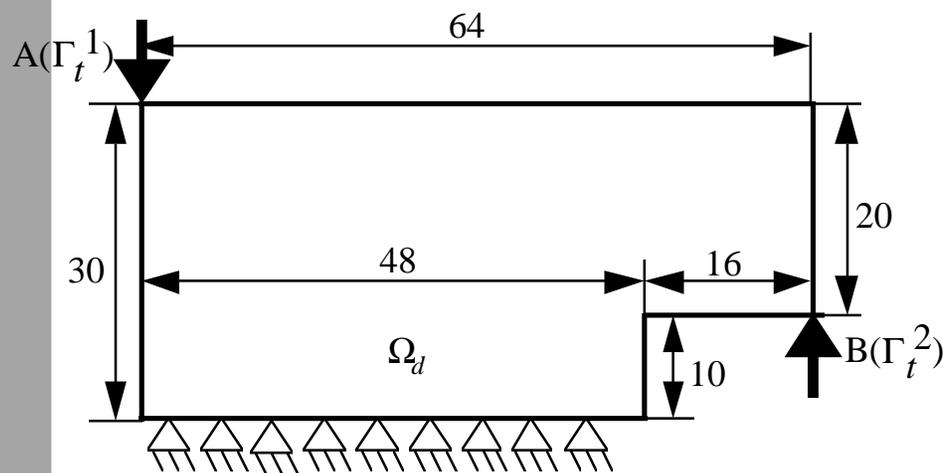


No Filtering

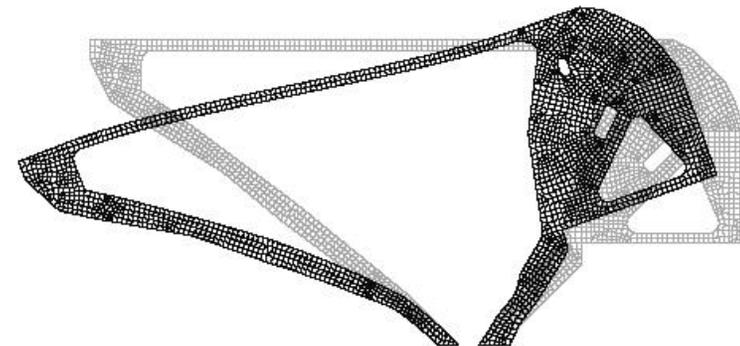
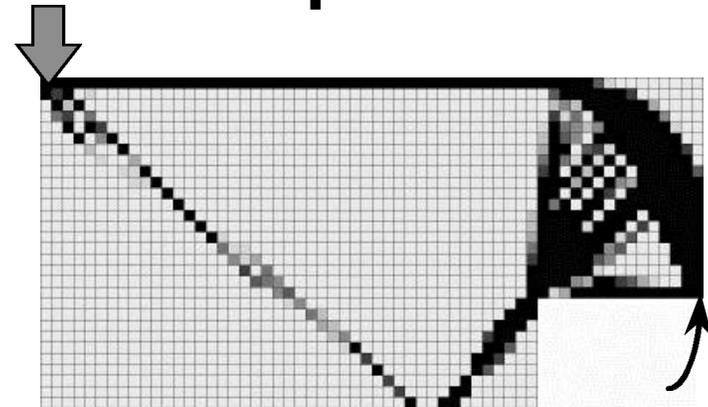
Filtering



Compliant Clamp



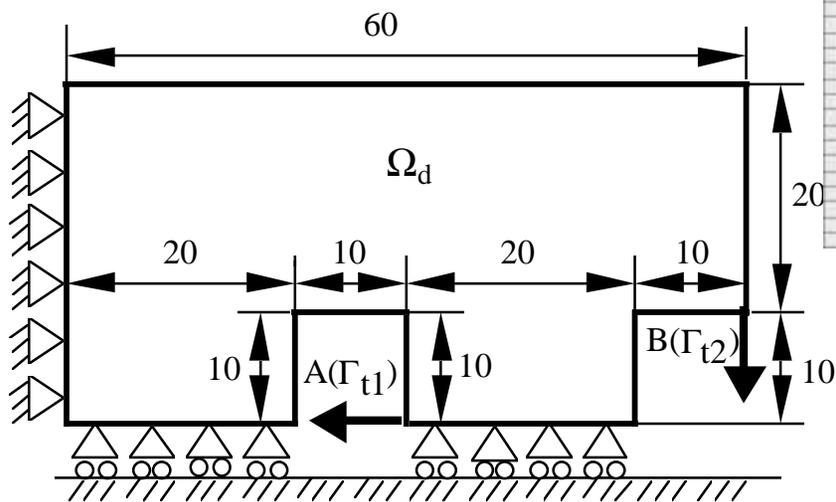
Design domain



Large displacement analysis

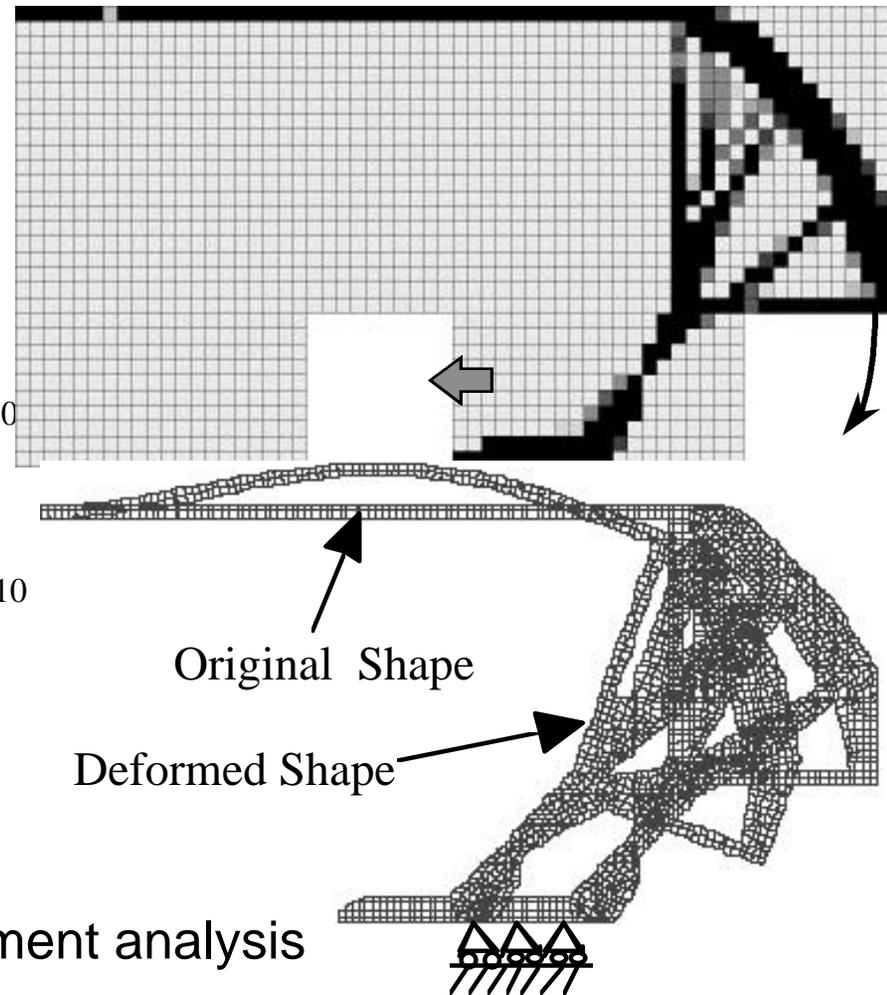


Compliant Gripper



Design domain

Large displacement analysis

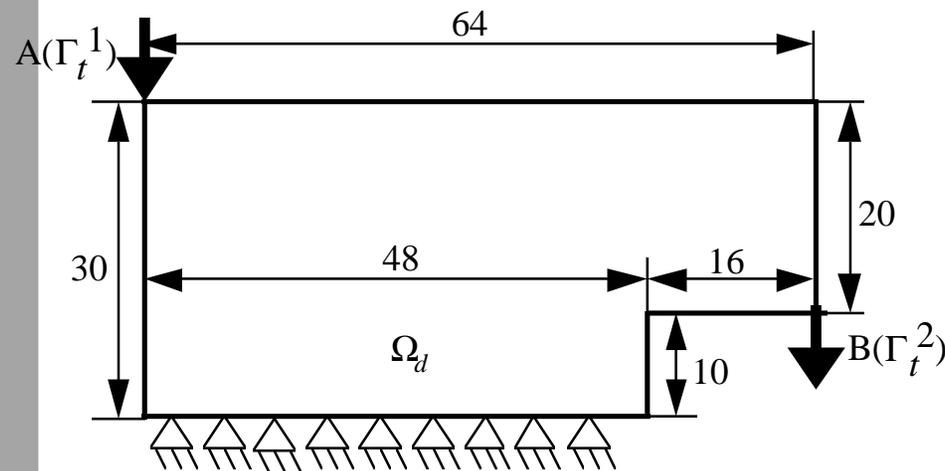


Original Shape

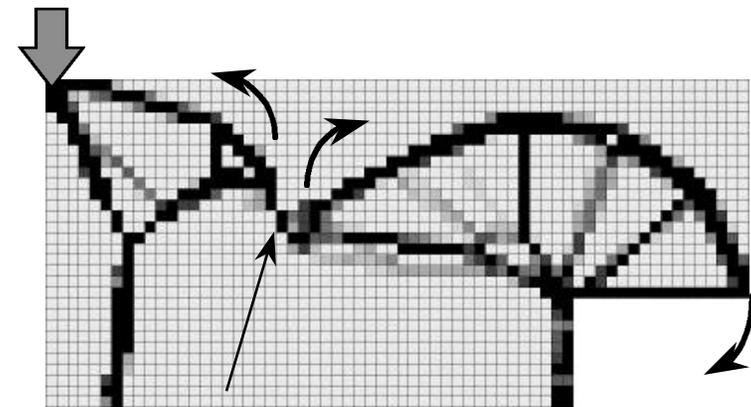
Deformed Shape



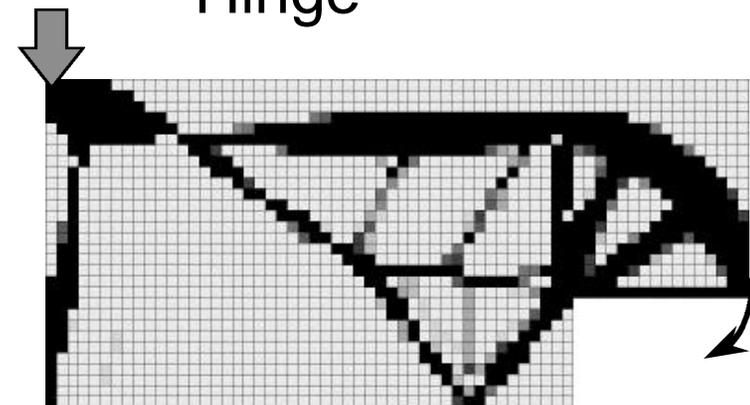
Compliant Pliers



Design domain



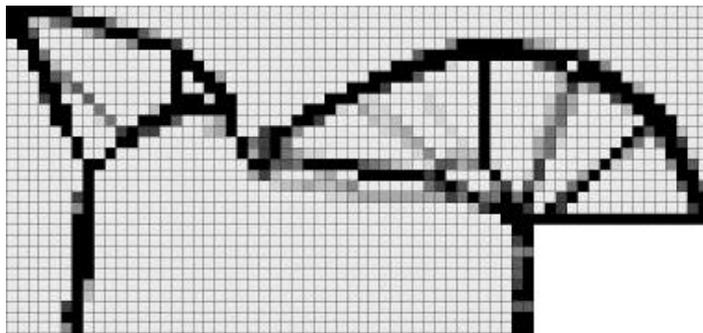
Hinge



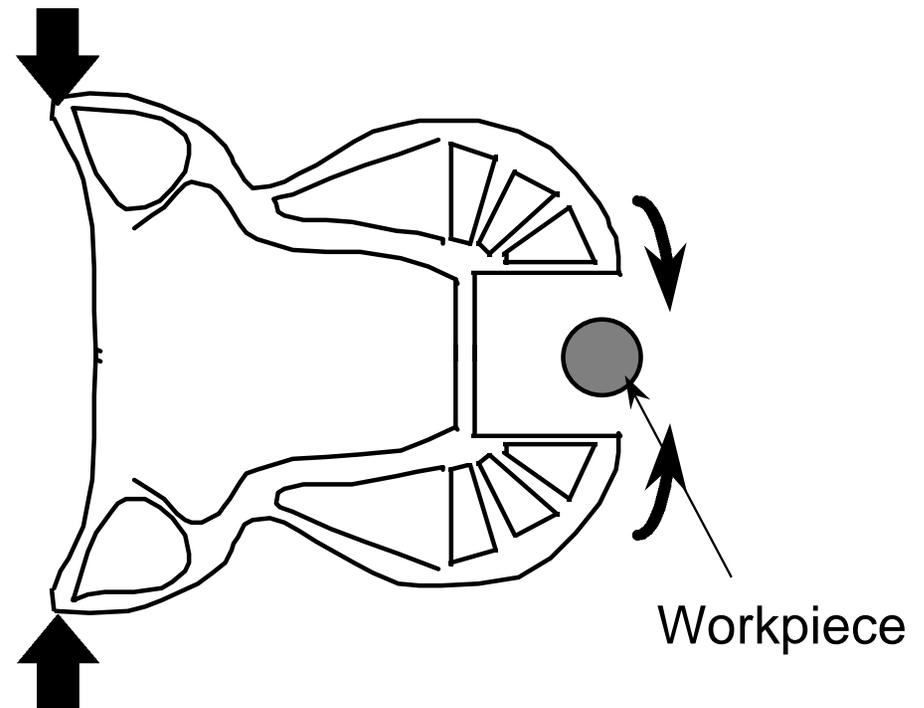
Optimal configurations



Final Configuration



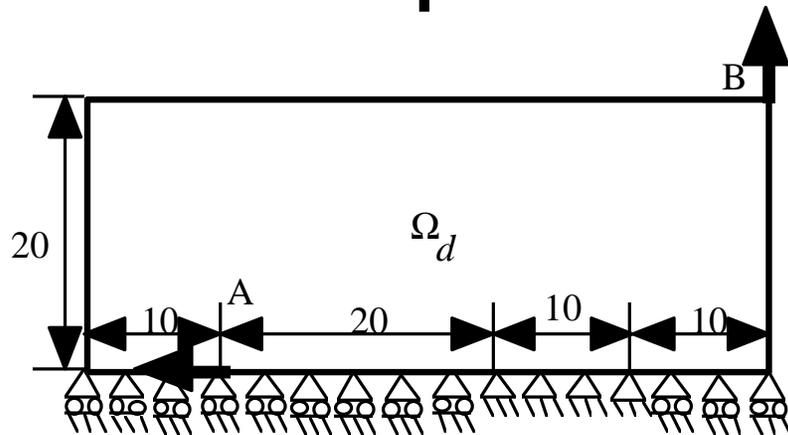
Optimal configuration



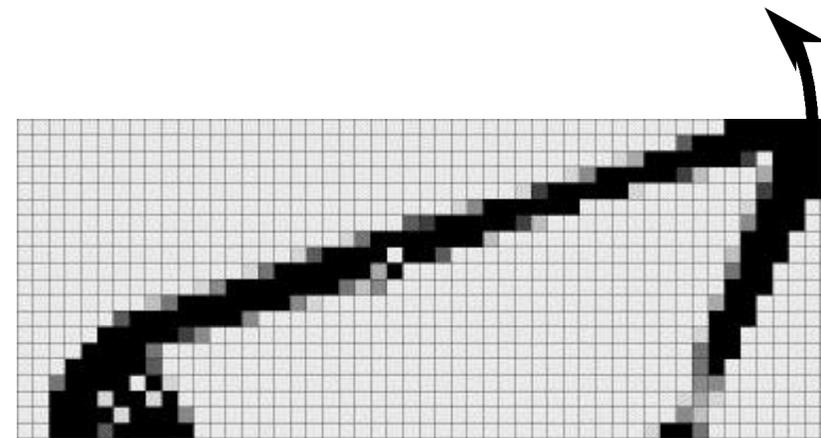
Final configuration



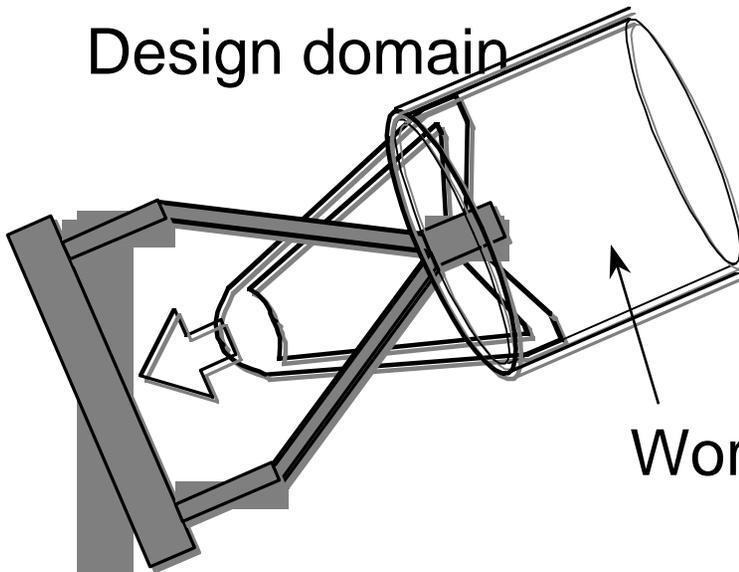
Compliant Internal Gripper



Design domain



Optimal configuration

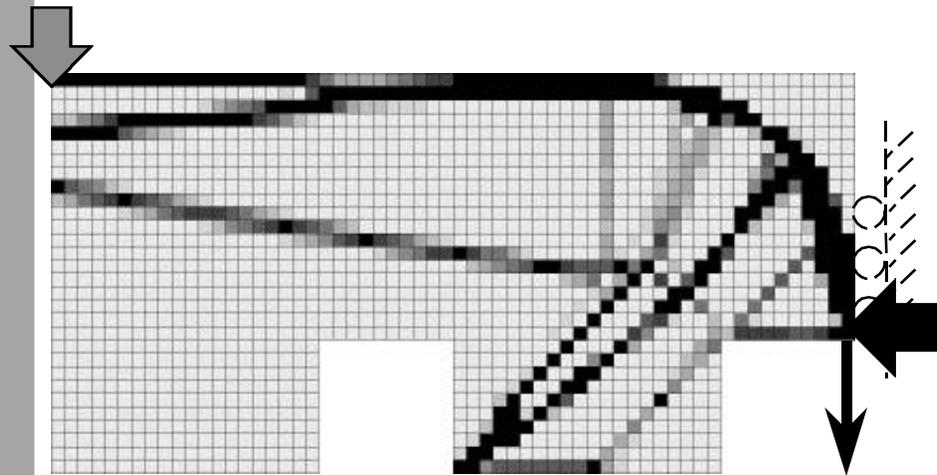
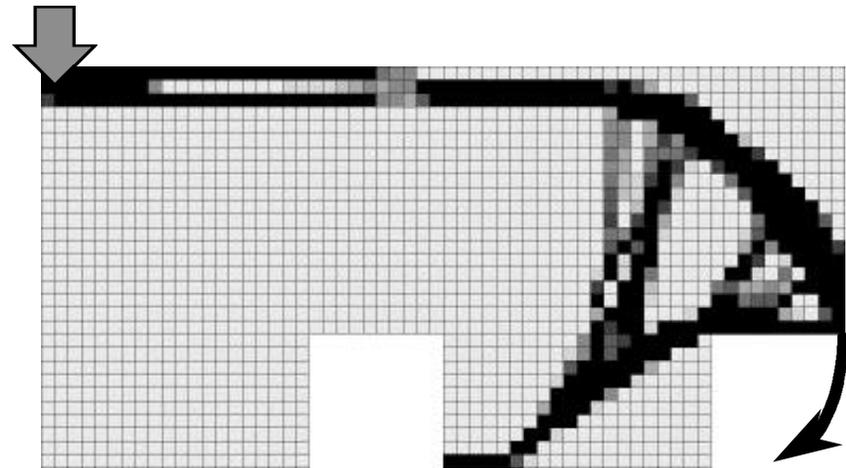


Workpiece



Constrained Motion

Compliant Gripper



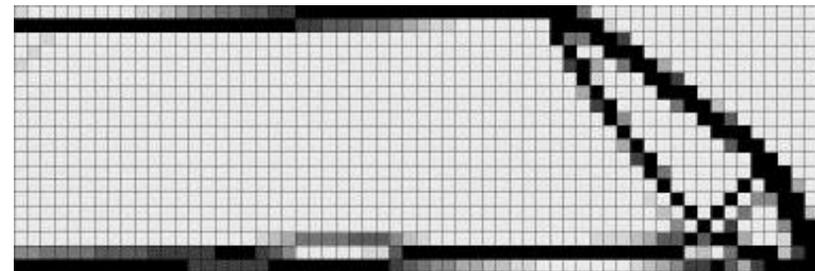
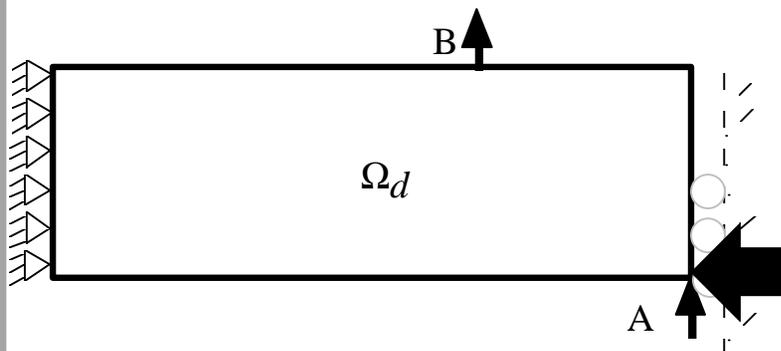
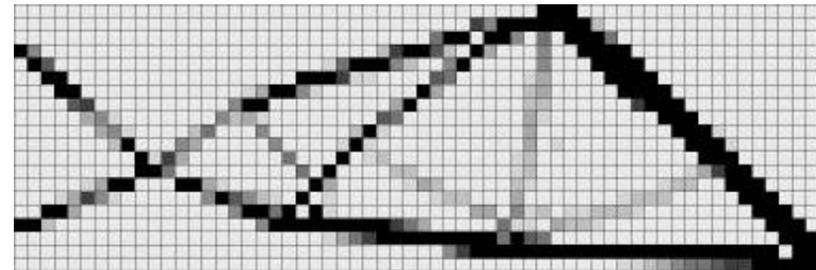
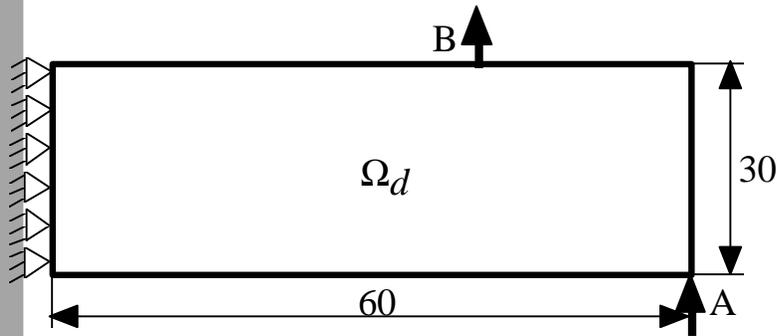
Adding Mutual Stiffness



Transitional Motion



Suspension Design



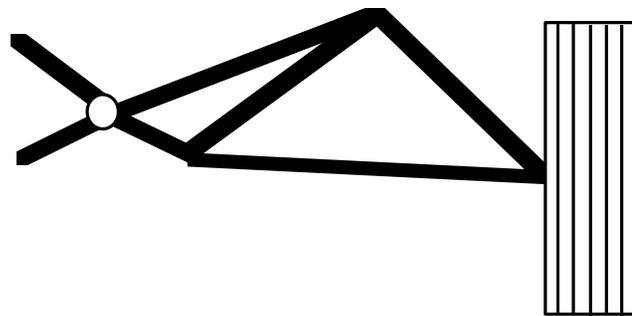
Design domains

Optimal configurations

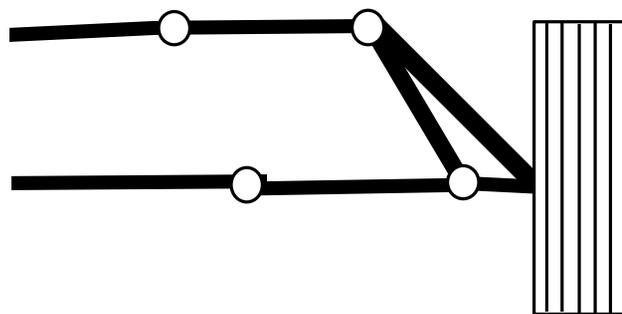
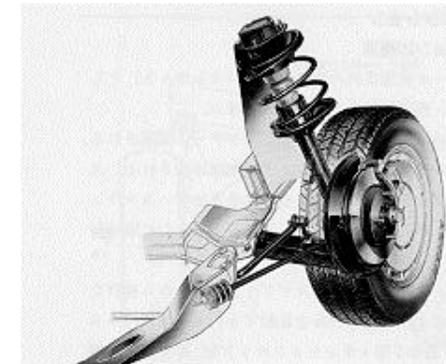


Interpretation

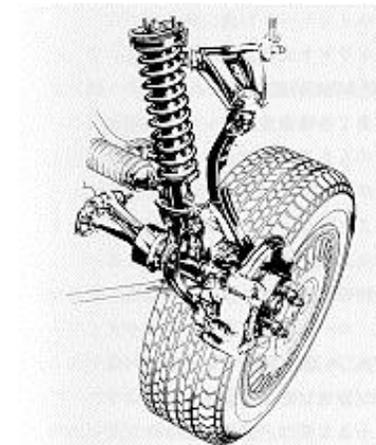
Optimal Configuration + Deformed Shape



Strut
Type



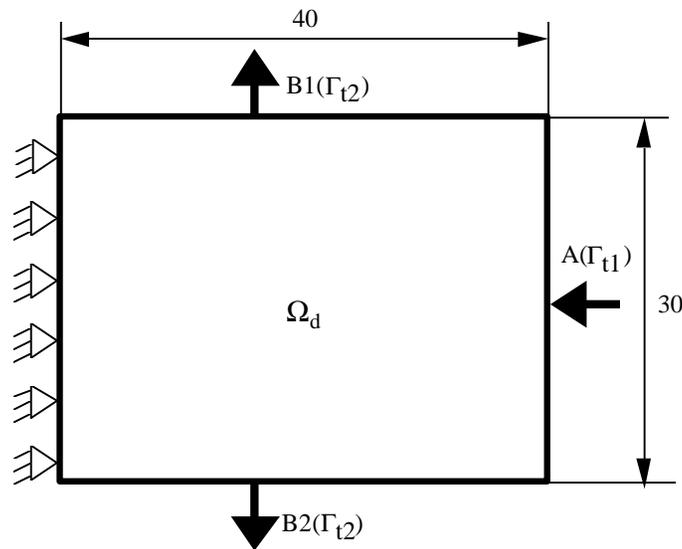
Double wish-
bone Type



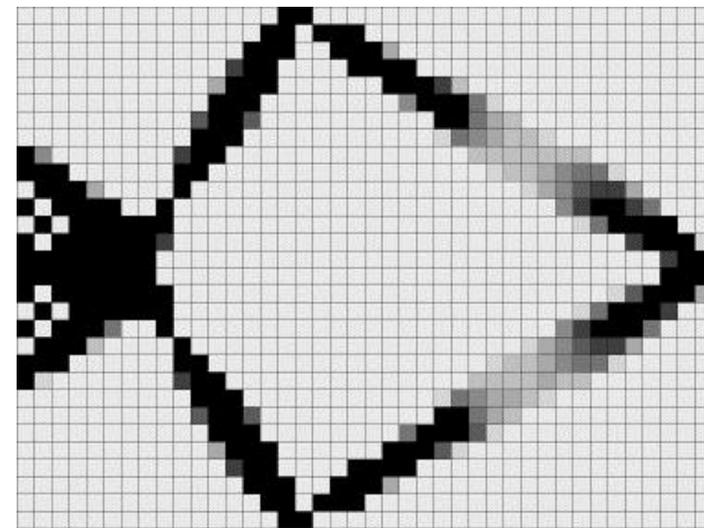
Rigid Link Mechanisms



Two Displacement Outputs



Design domain



Optimal configuration



Extension to Multi-Flexibility

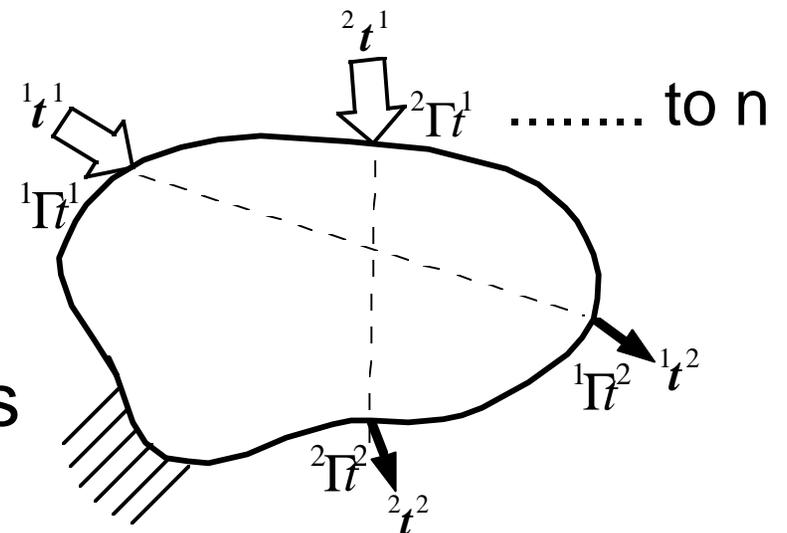
Necessary for Many Performance Criteria

➔ Automotive Body Design

n Flexibilities Required



n Mutual Mean Compliances
Should be **Positive**





Multi-criteria Optimization

Max i -th Mutual Mean Compliance (i MMC)
(for $i=1, \dots, n$)

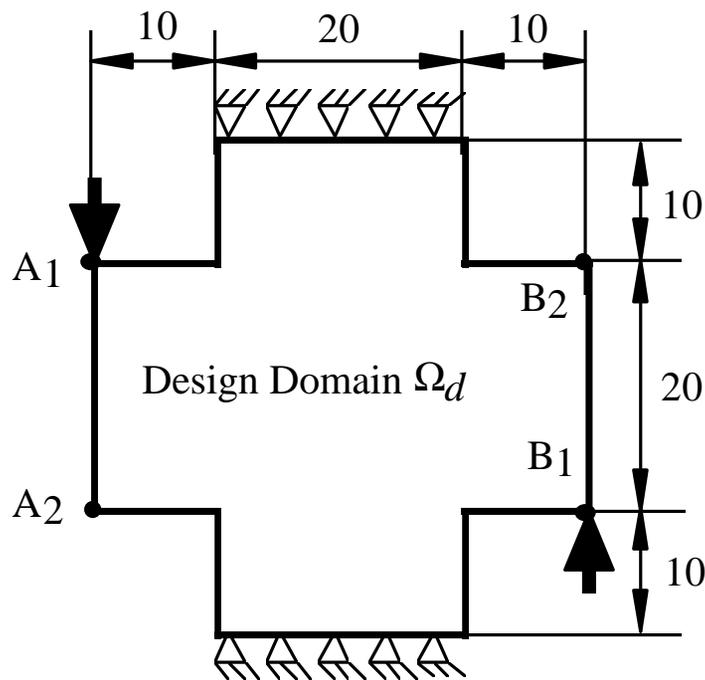
Min i -th \sum Mean Compliance (i MC)
(for $i=1, \dots, n$)

- Multi-Objective Function

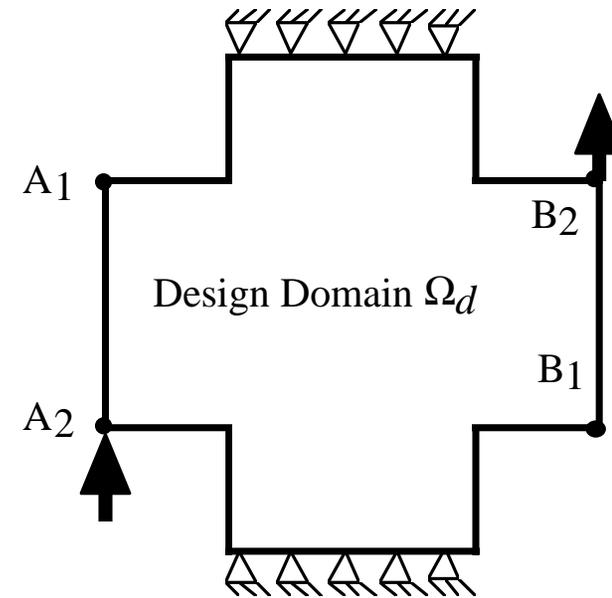
$$\text{Max } \frac{-1/C_f \text{Log}(\sum \text{Exp}(-C_f i \text{MMC}))}{1/C_s \text{Log}(\sum \text{Exp}(C_s^i \text{MC}))}$$



Multi-flexibility Design



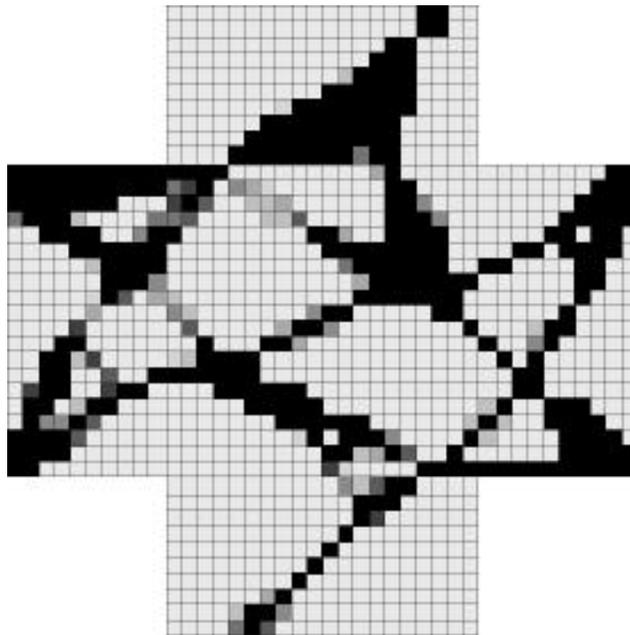
First flexible mode



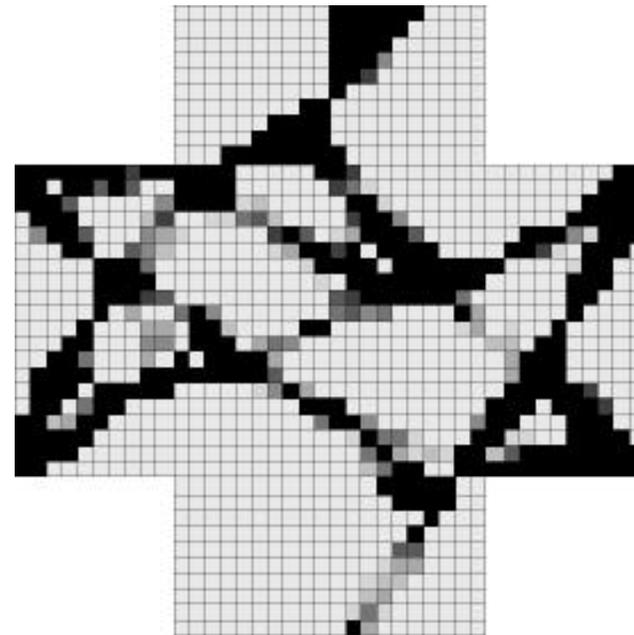
Second flexible mode



Optimal Configurations



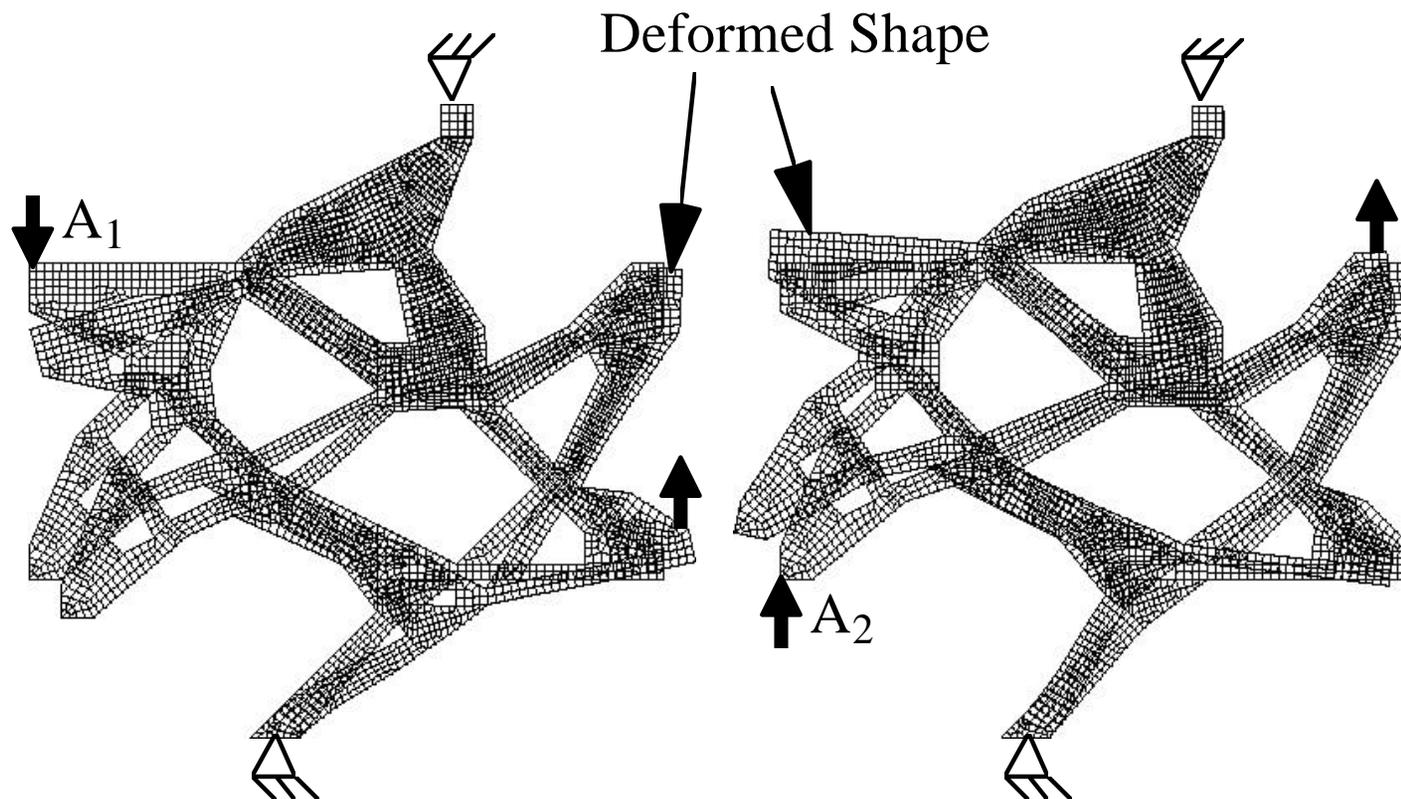
$$C_f = C_s = 3$$



$$C_f = C_s = 10$$



Verification of Performance



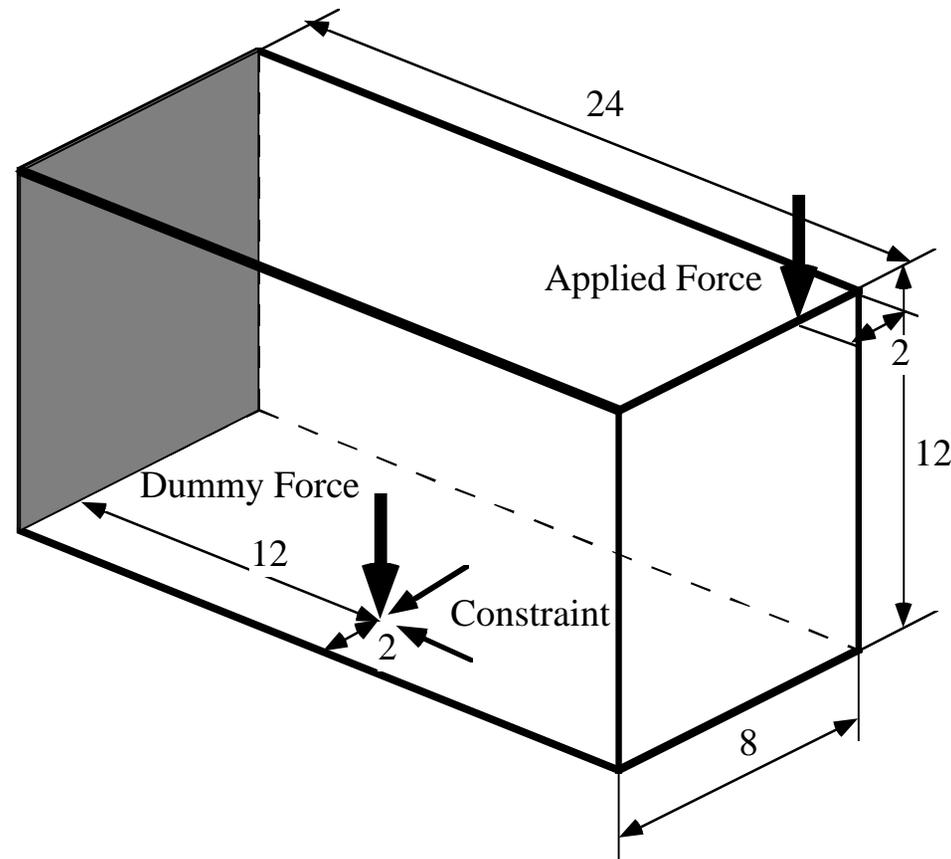


Extension

- 3D Compliant Mechanism Design
 - ➔ New 3D element is used
- Compliant Mechanism Design with a Displacement Constraint
 - ➔ New formulation is introduced
 - + Image based design



Design of Simple Model

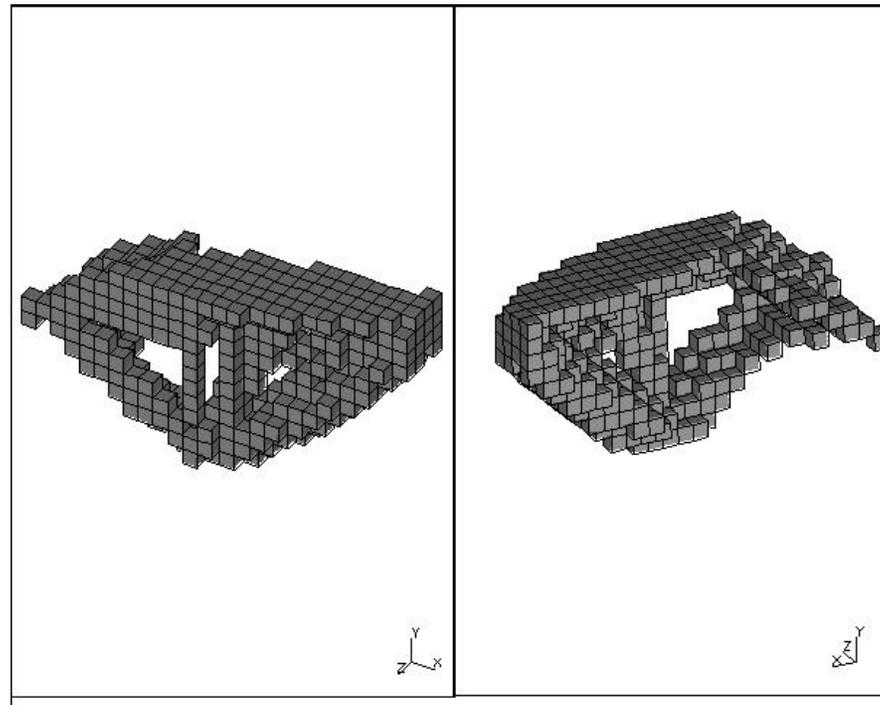


Design Domain



Optimal Configurations (1)

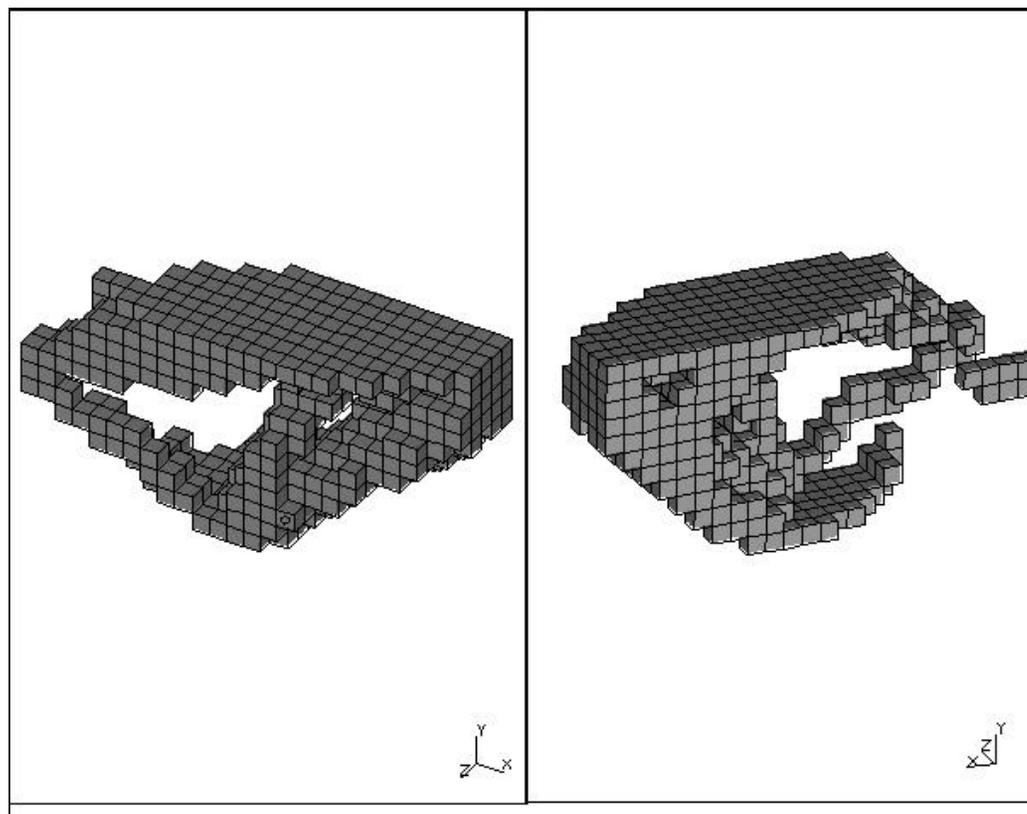
Unconstrained Case



Total Volume = 20%



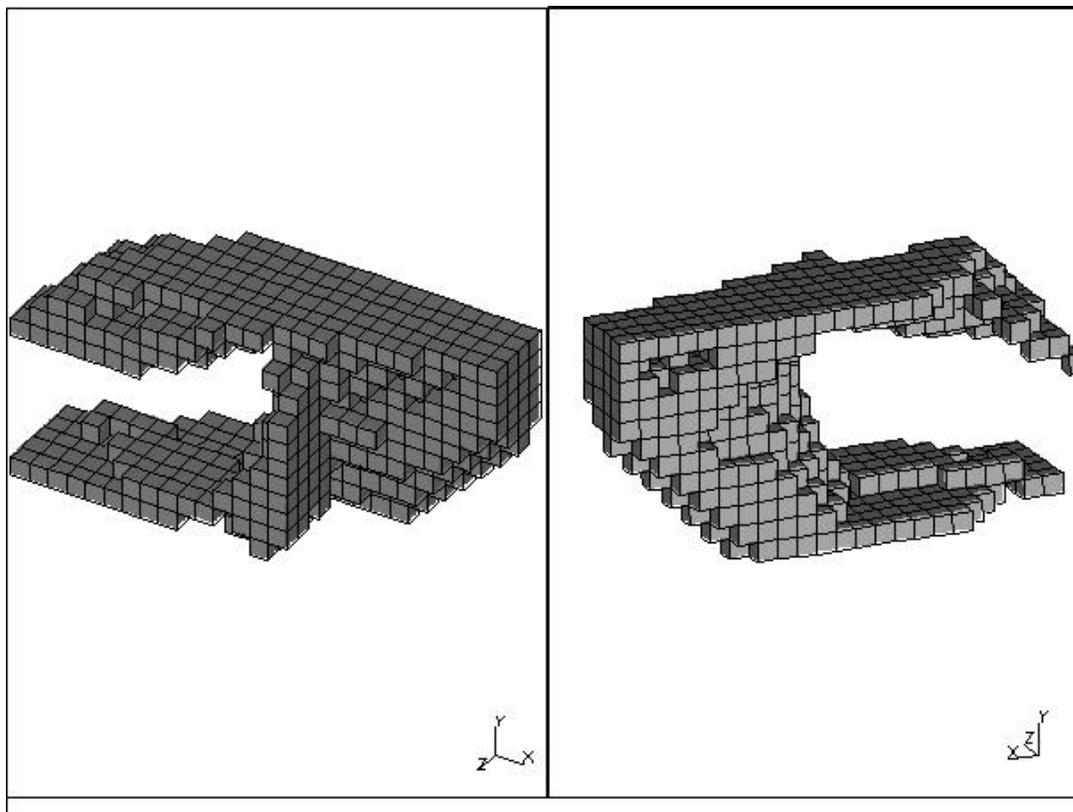
One Constraint Case



(Total Volume = 30%)



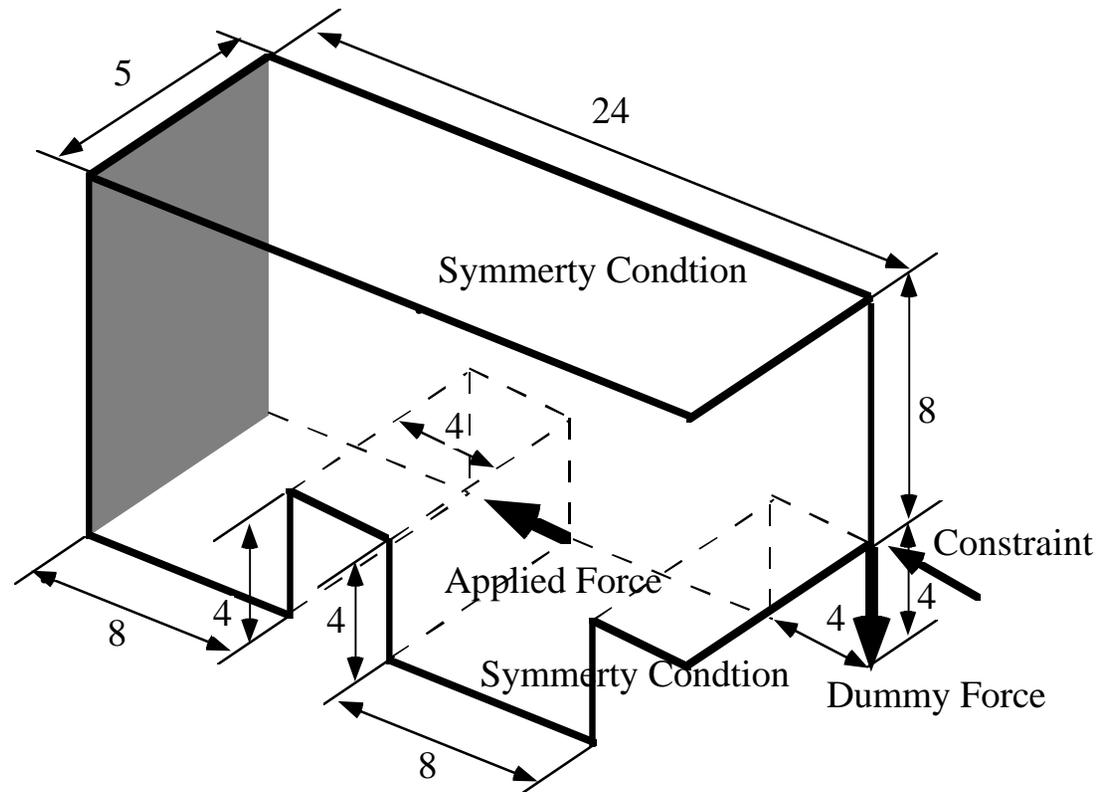
Two Direction Constraint



(Total Volume = 30%)



Design of Gripper

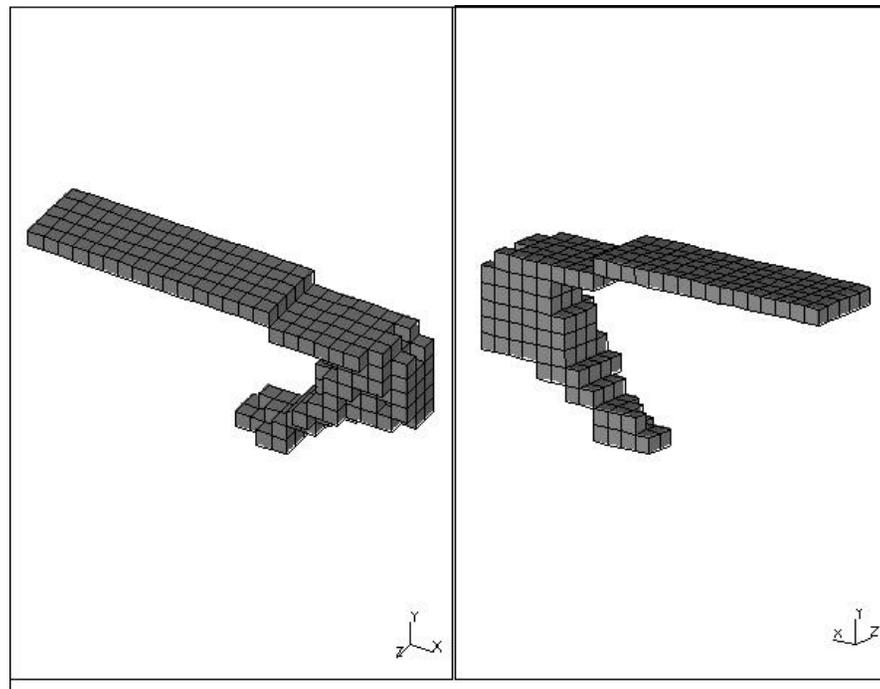


Design Domain



Optimal Configurations (1)

Unconstrained Case

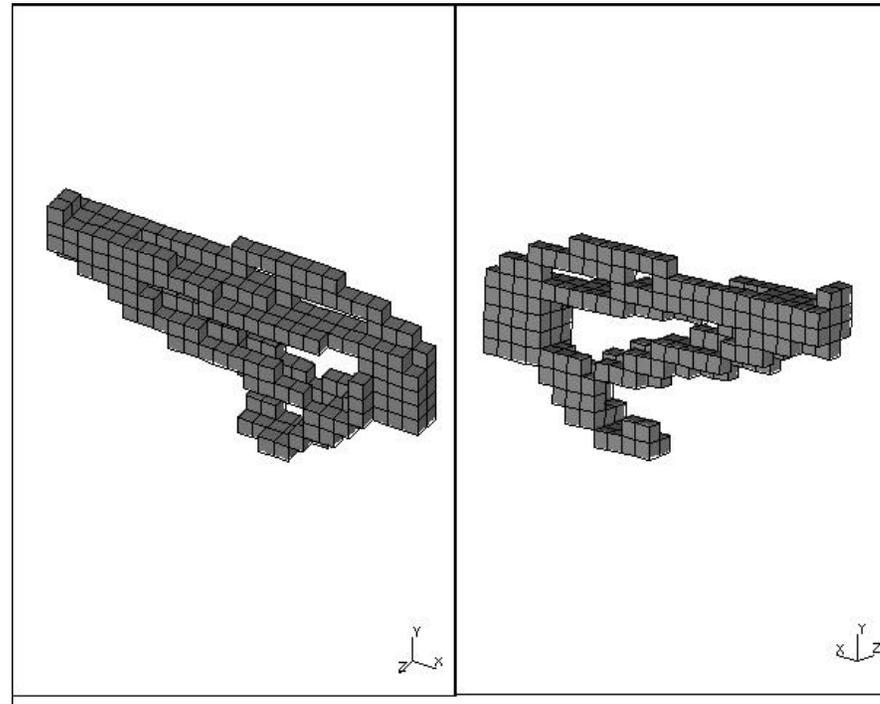


Total Volume = 20%



Optimal Configurations (2)

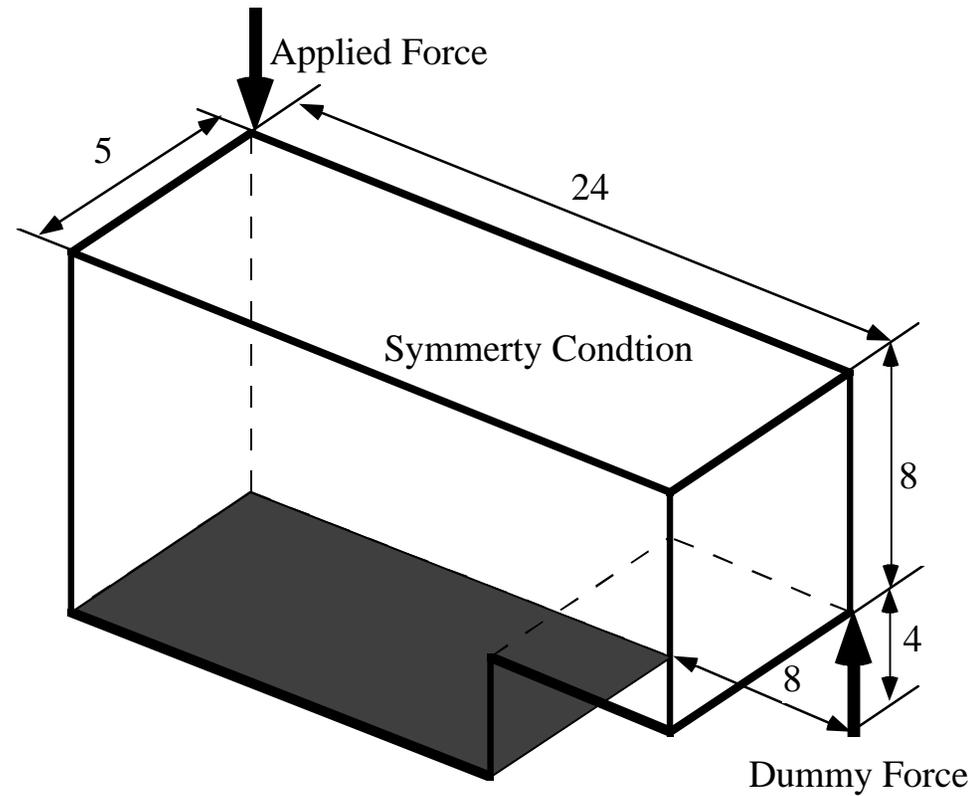
Constrained Case



Total Volume = 20%



Design of Clamp

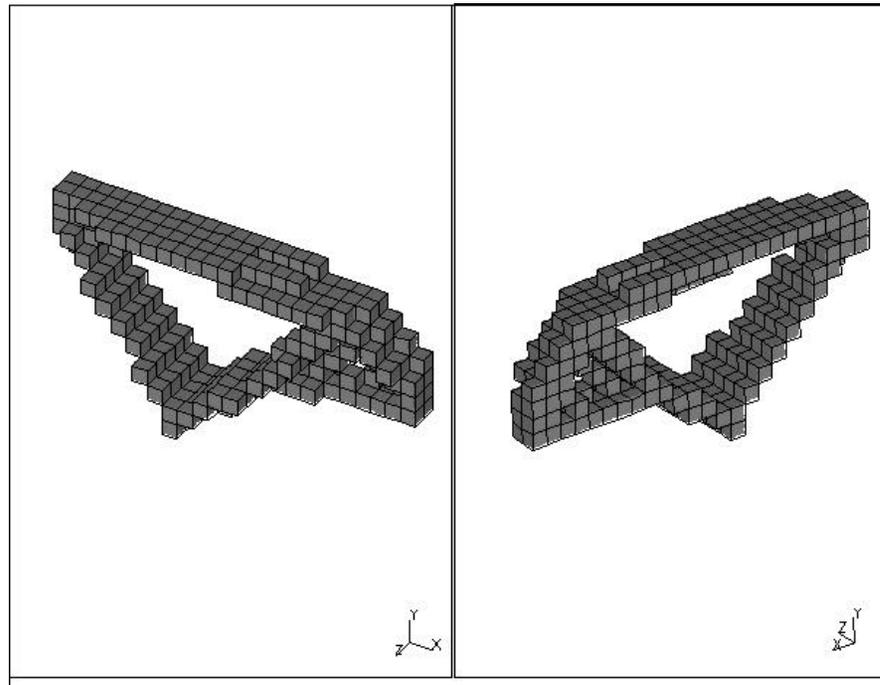


Design Domain



Optimal Configurations

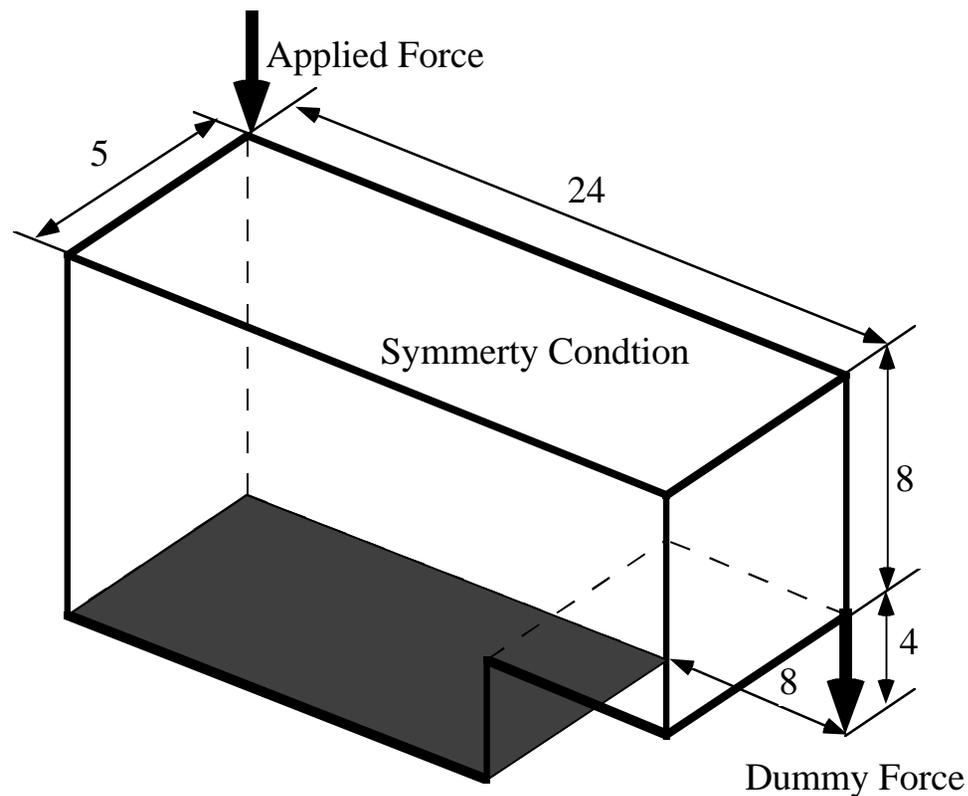
Unconstrained Case



Total Volume = 20%



Design of Pliers

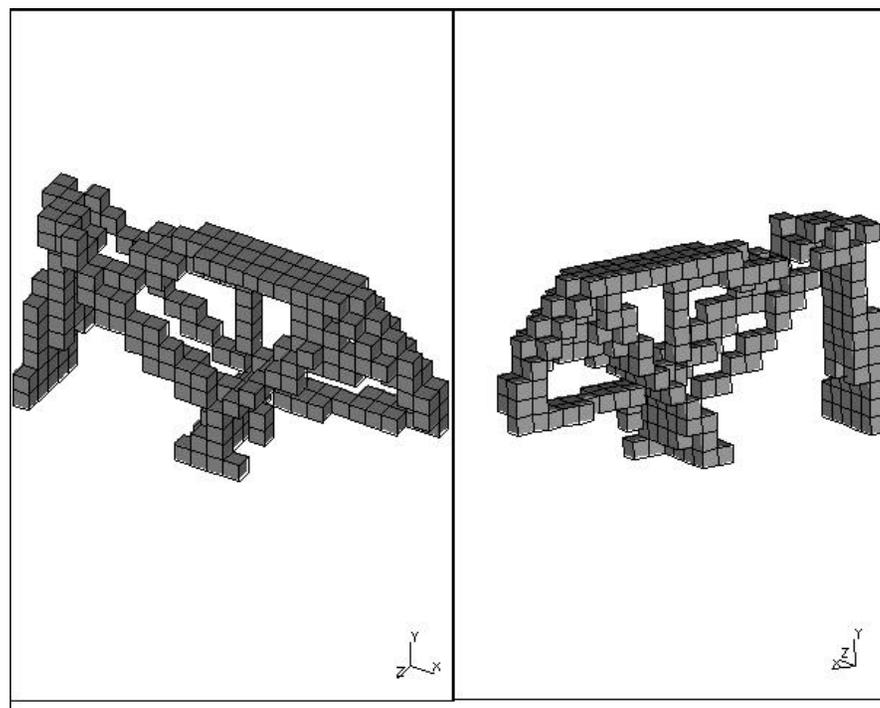


Design Domain



Optimal Configurations

Unconstrained Case



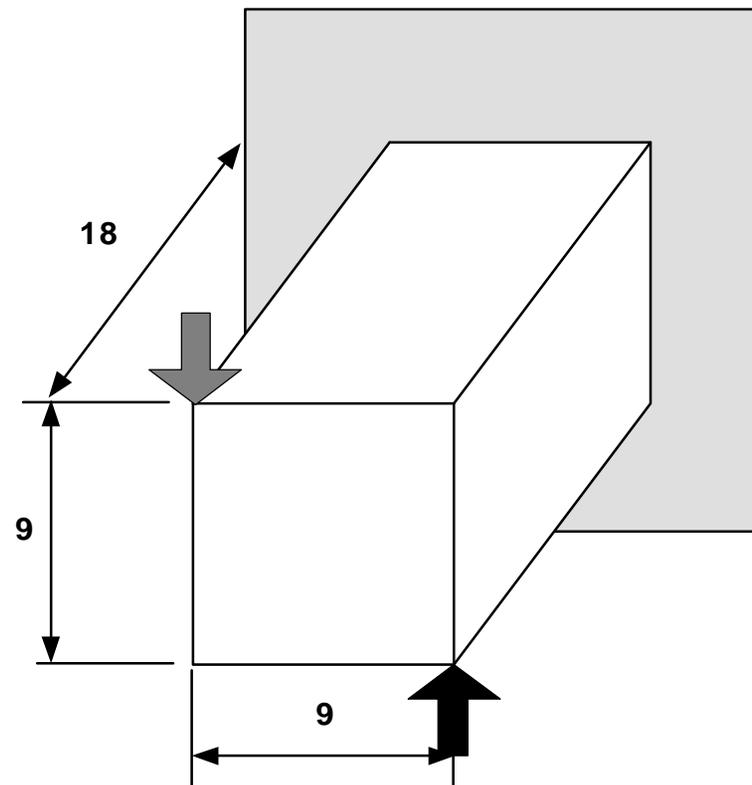
Total Volume = 20%



Design of Torsion Bar

← Applied Force

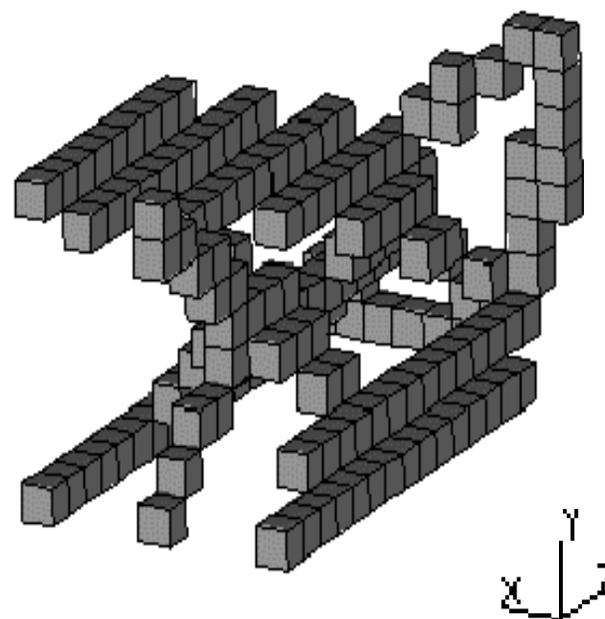
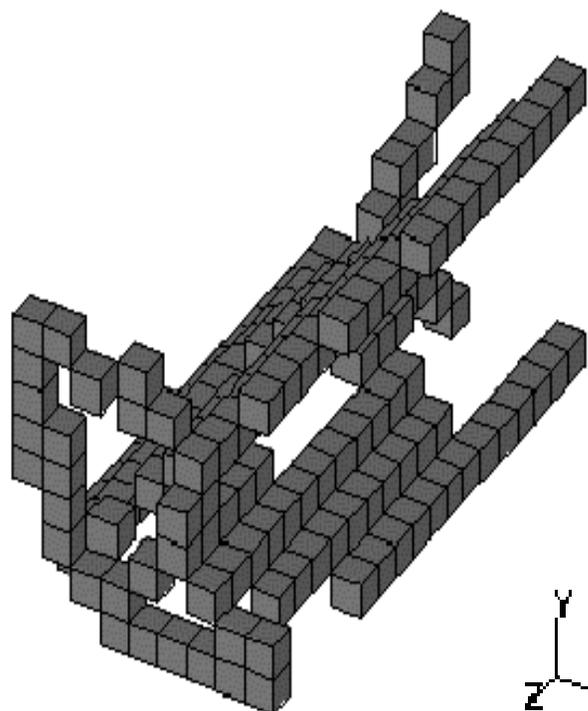
← Flexibility



Design Domain



Optimal Configuration



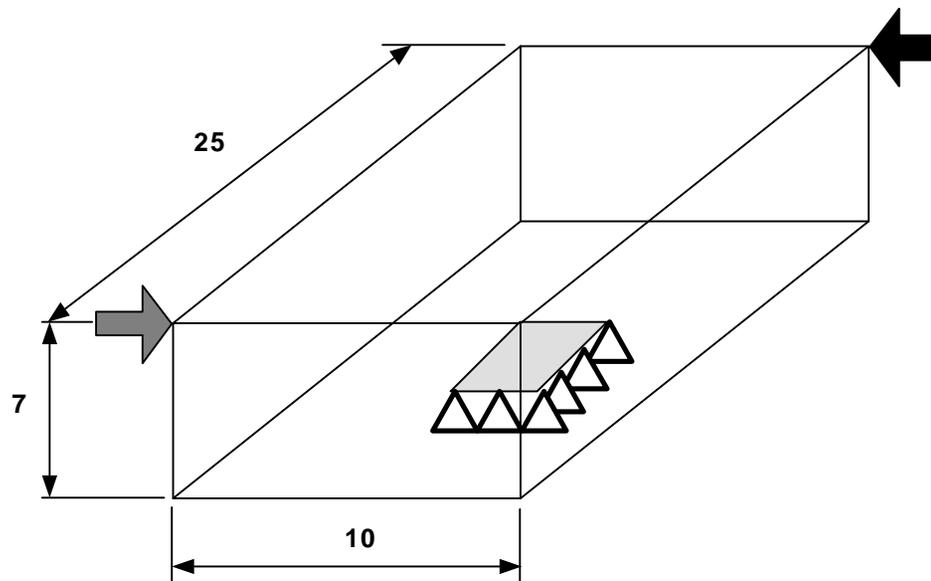
Total Volume = 10%



Design of Torsion Plate (1)

← Applied Force

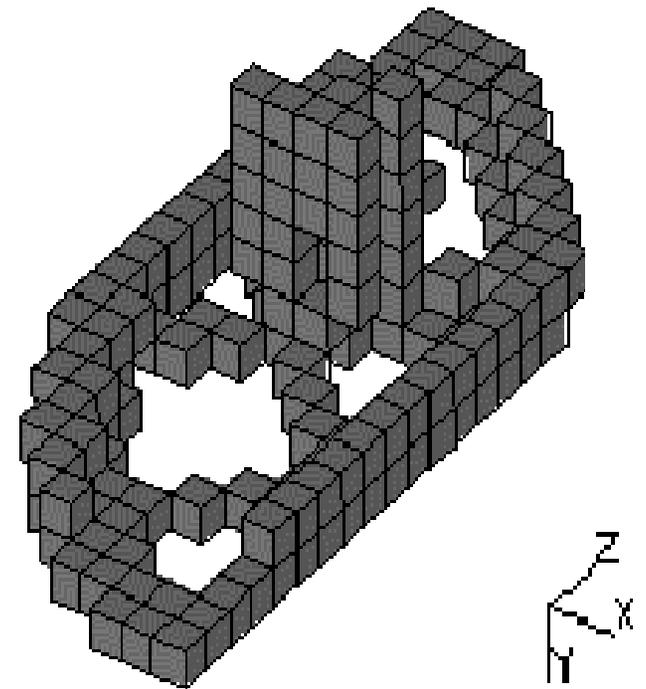
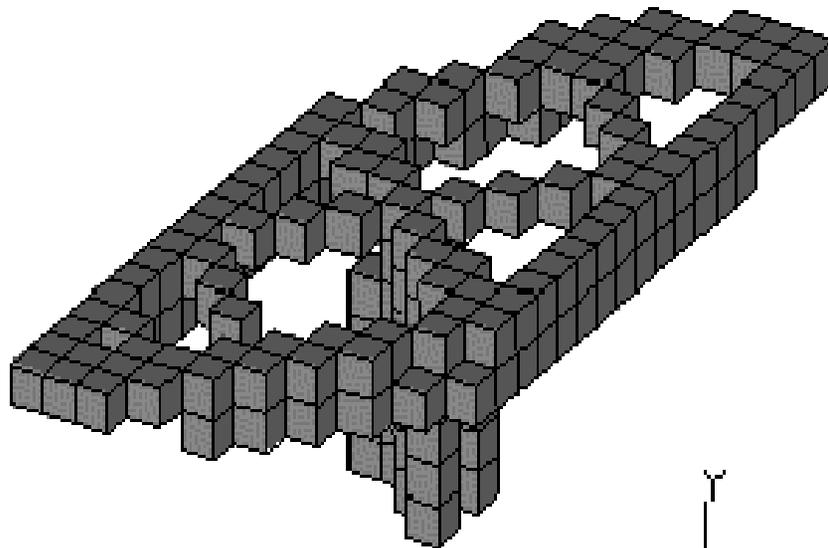
← Flexibility



Design Domain



Optimal Configuration (1)



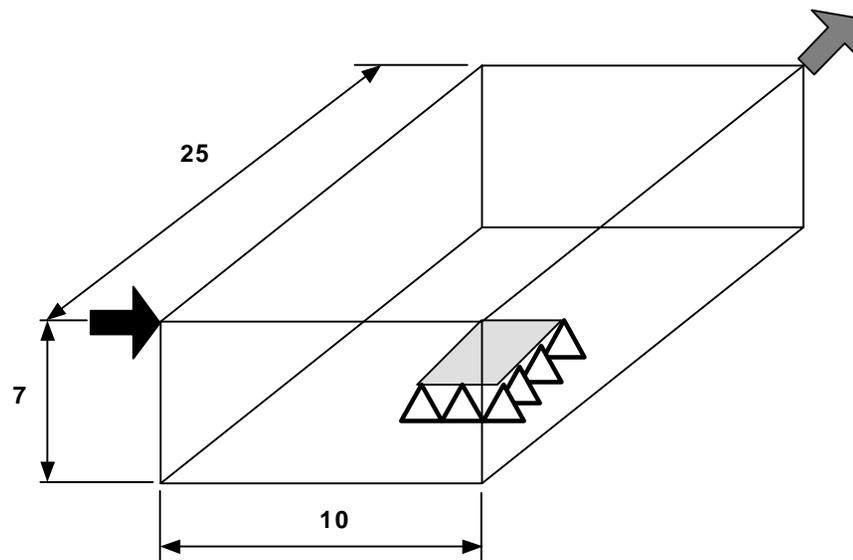
Total Volume = 10%



Design of Torsion Plate (2)

← Applied Force

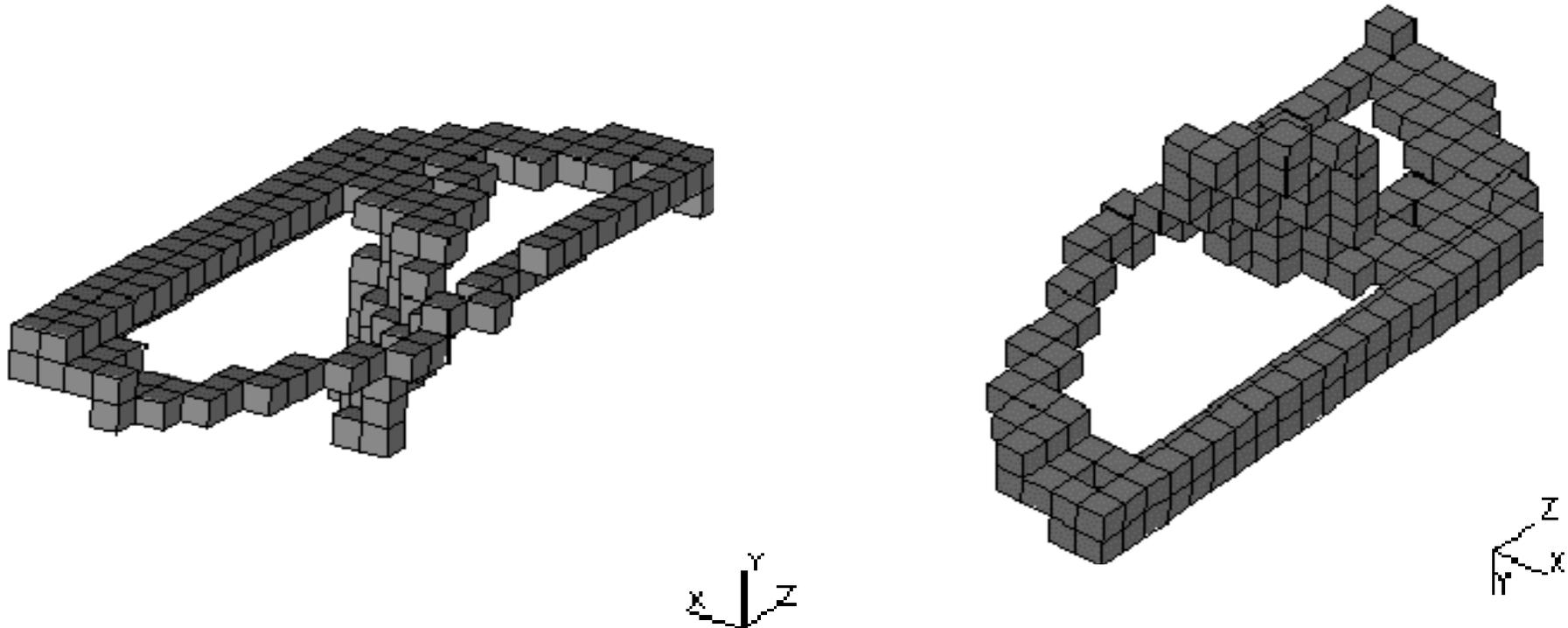
← Flexibility



Design Domain



Optimal Configurations (2)



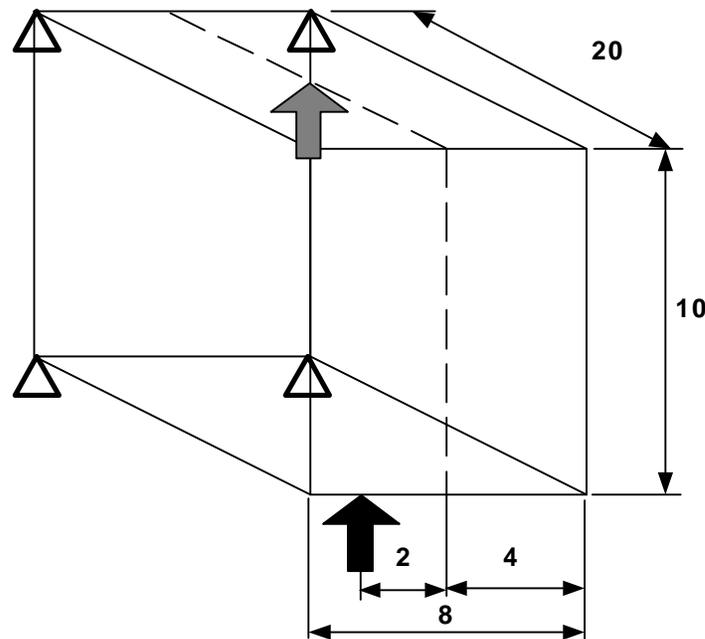
Total Volume = 10%



Design of Sus-Like Structure

← Applied Force

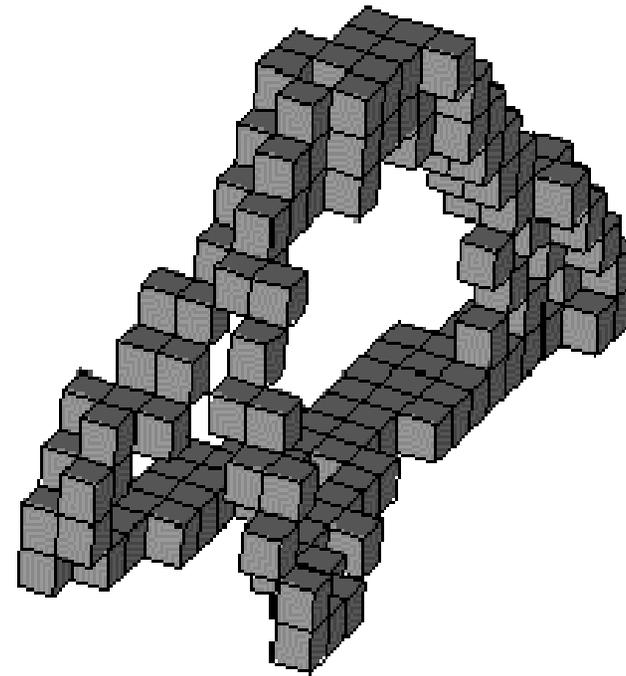
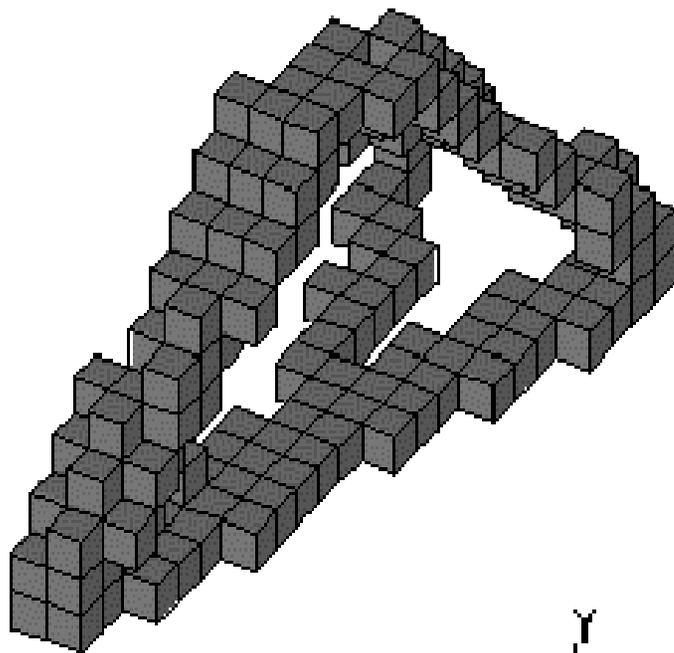
← Flexibility



Design Domain



Optimal Configuration



Total Volume = 10%



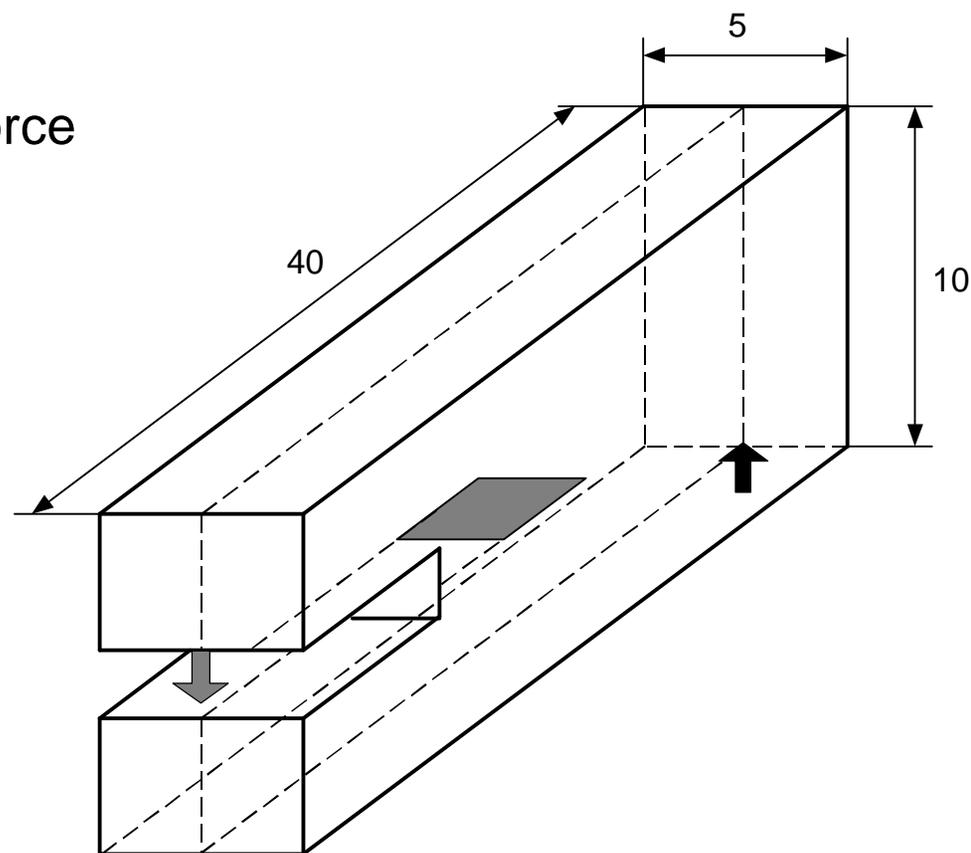
Design of Another Gripper



Applied Force



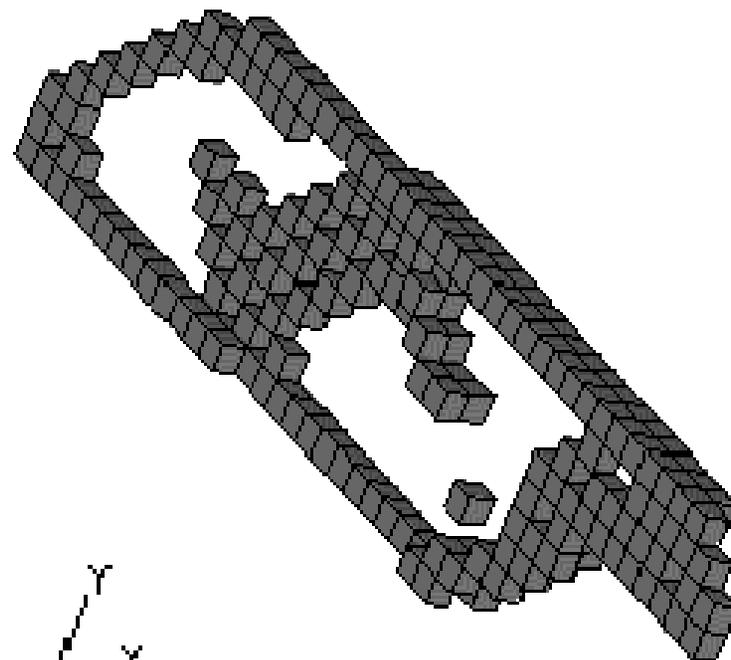
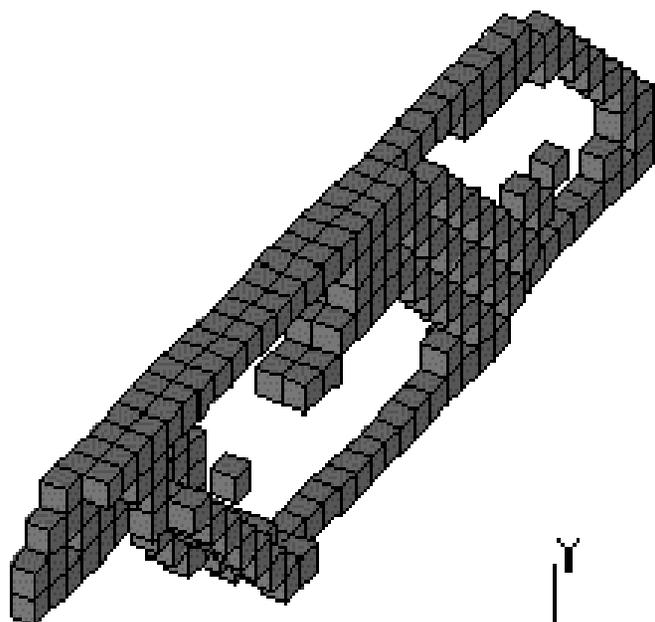
Flexibility



Design Domain



Optimal Configuration



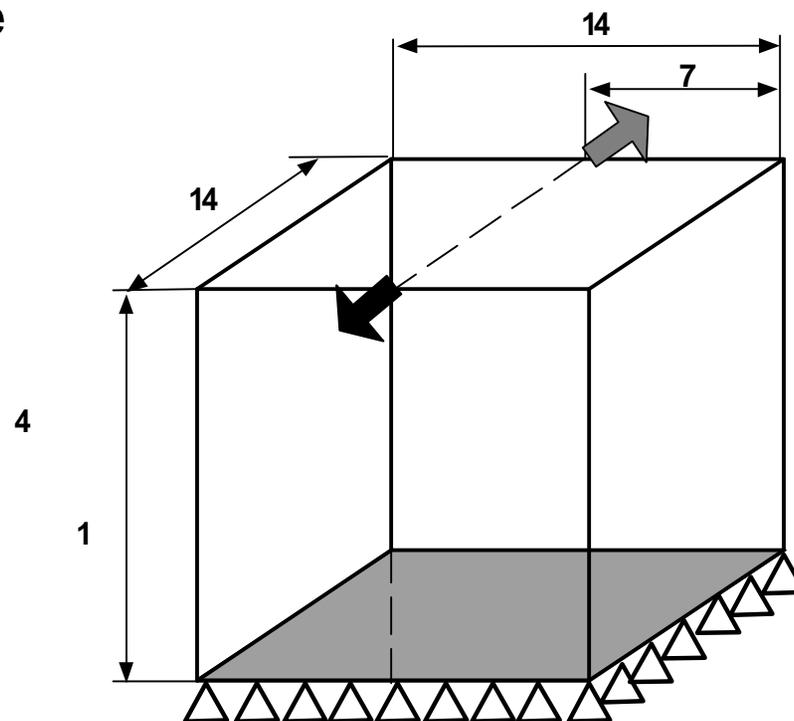
Total Volume = 10%



Design of Tensile Model

← Applied Force

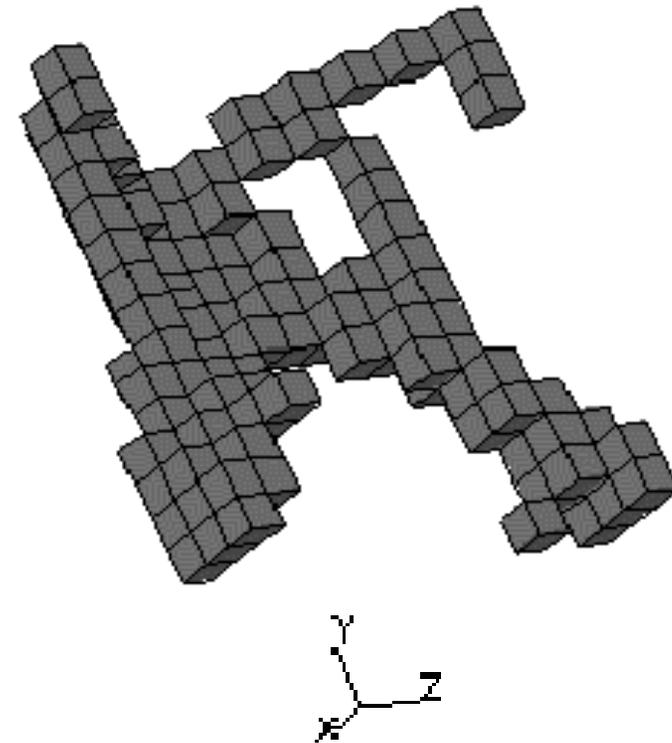
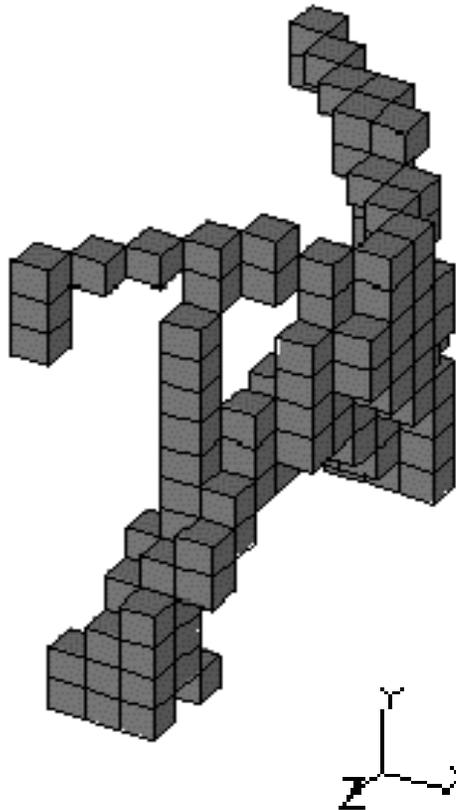
← Flexibility



Design Domain



Optimal Configurations



Total Volume = 10%



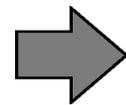
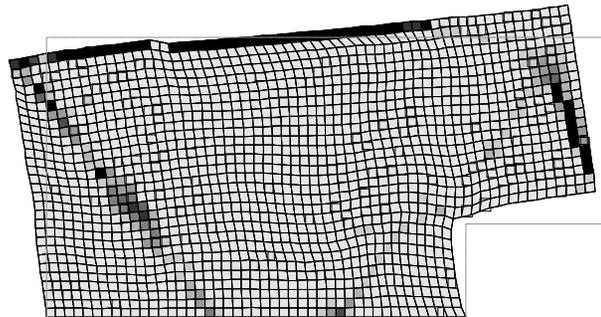
Compliant Mechanism Design with a Displacement Constraint



Multi-Objective Function (a)

Phase 1

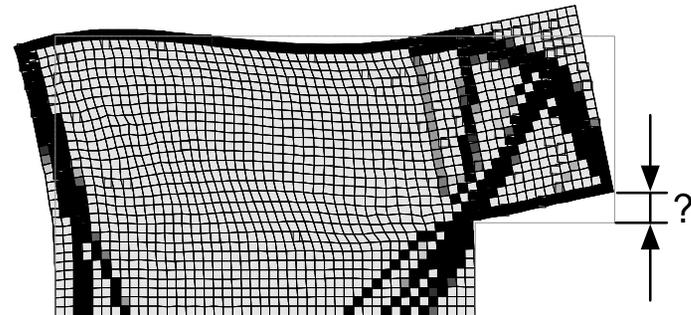
$Max[Flexibility]$



Phase 2

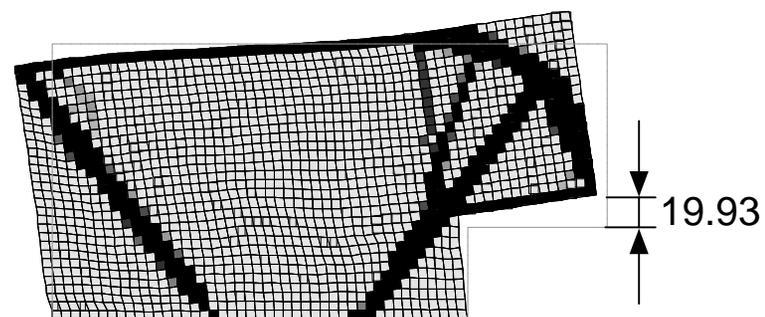
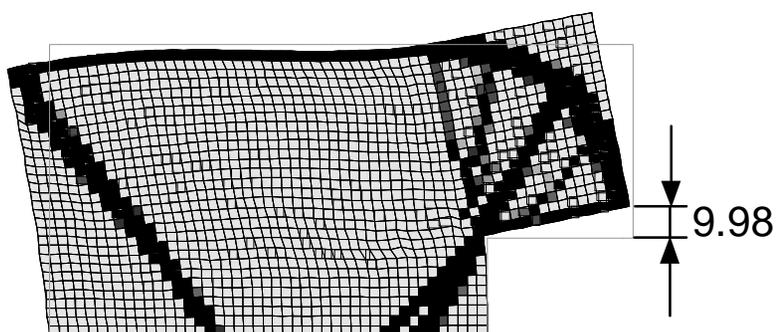
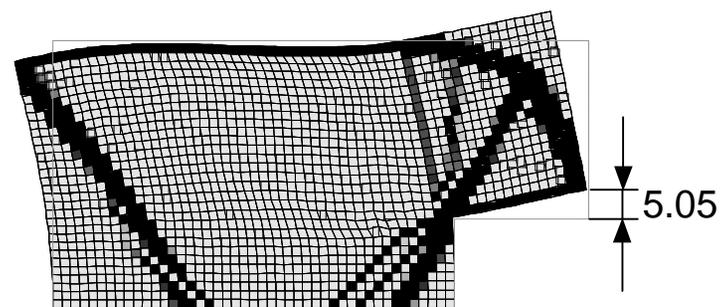
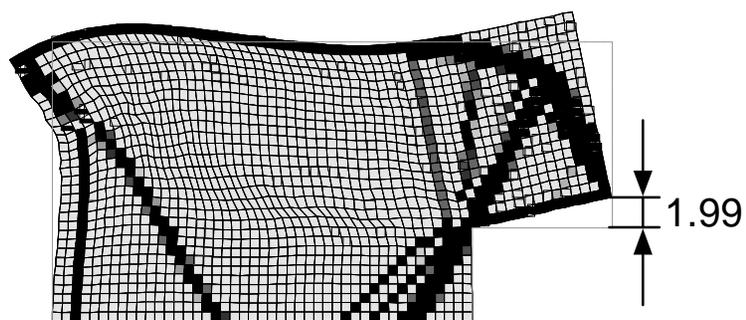
$Max[Stiffness]$

$St. Flexibility \geq Pr$ escribed Flexibility





Optimal Configurations (a)





Multi-Objective Function (b)

Exterior penalty function method

$$\text{Min } \log(\text{Stiffness}) + m \log(\text{Flexibility}) - \log(\text{Prescribed Flexibility}) r^2$$



Optimal Configurations (b)

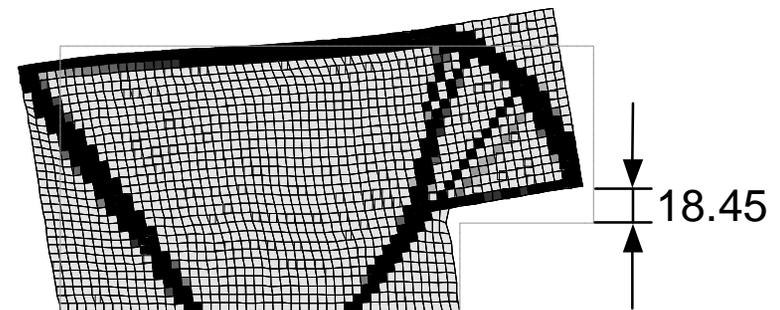
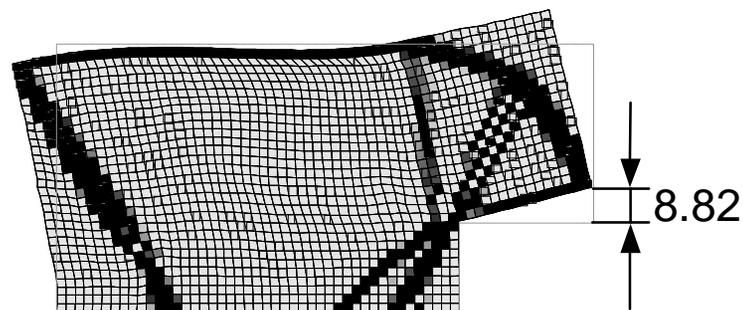
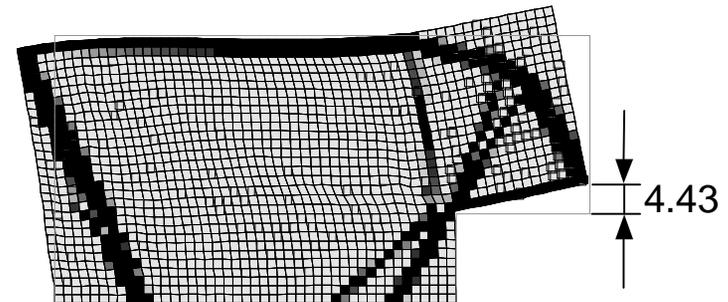
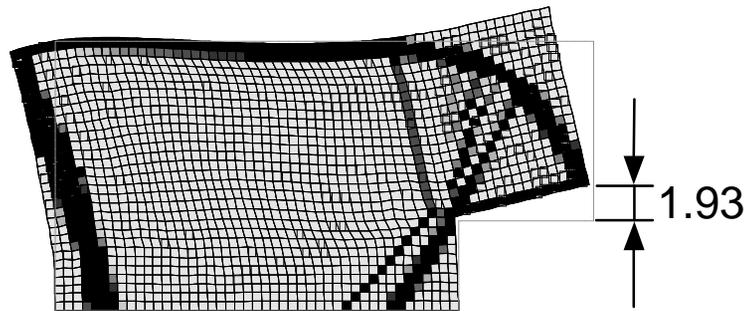




Image Based Design

Optimal Topology → Practical Structure

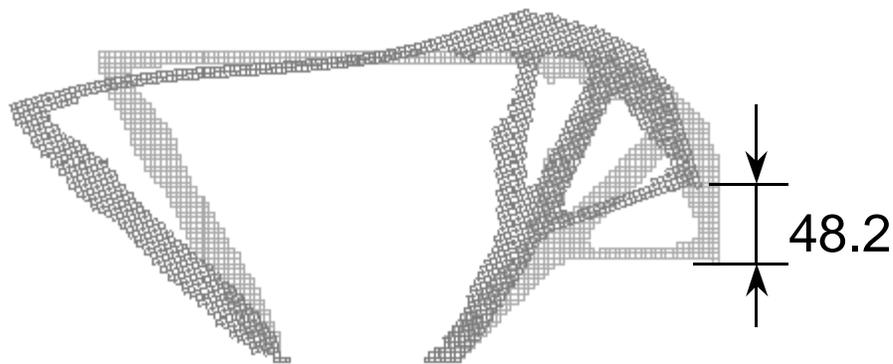
Image processing



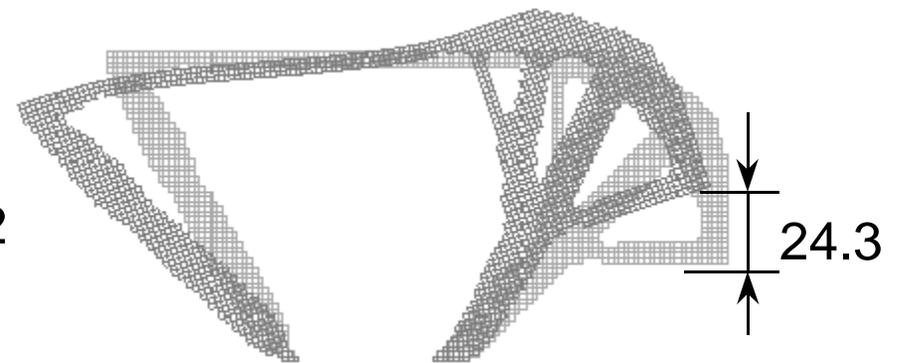


Verification of Displacement

Digital Mesh (Voxelcon)



Threshold=100



Threshold=80

Gray scale (0~256)



Material Micro-structure Design

Jun Fonseca : Brazil

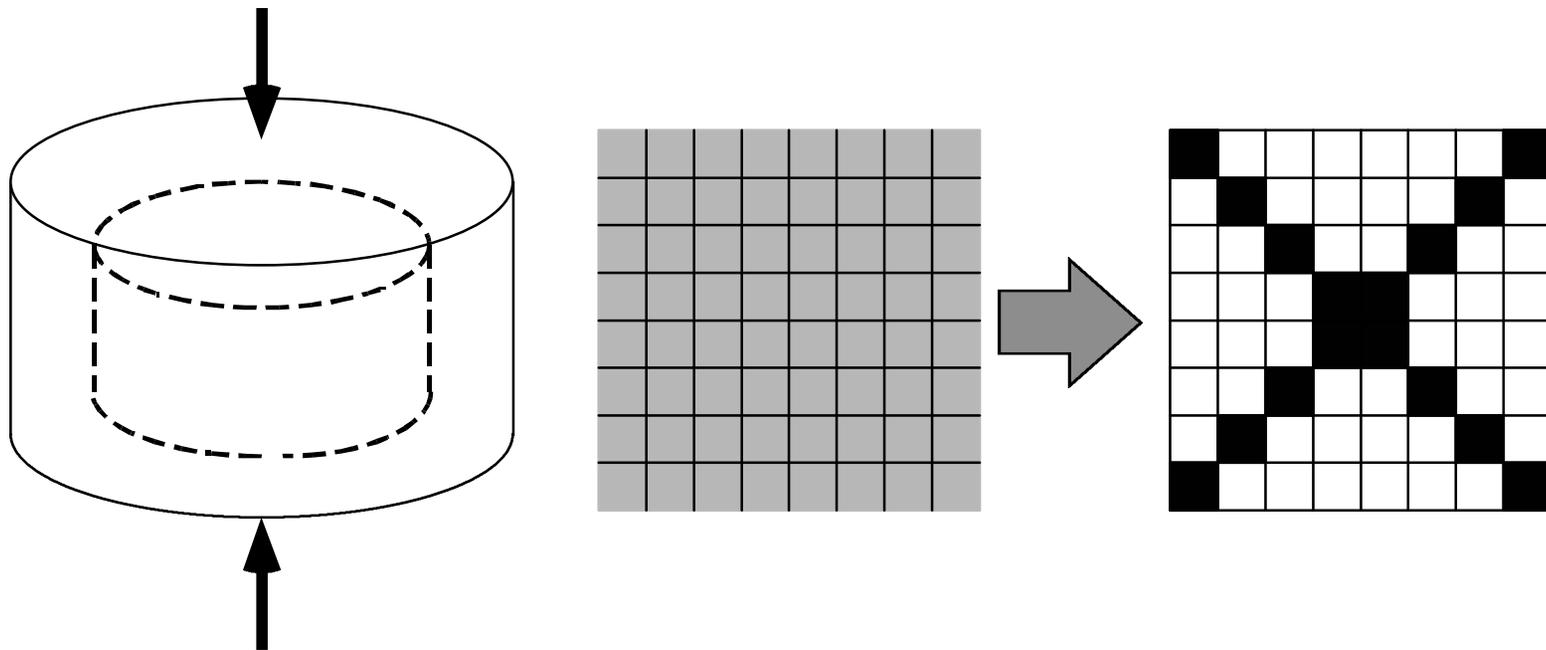
and

O. Sigmund : Denmark



Material Design

- Design negative Poisson's ratio by the homogenization design method





Solution Method 1

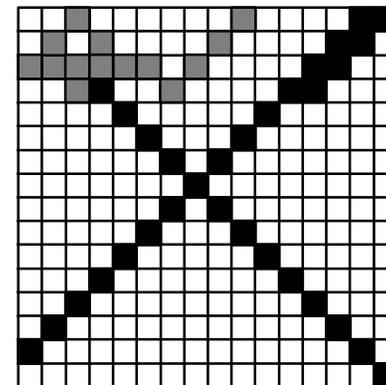
1. Specify the Material Constants Desired

$$D^{-1} = \begin{bmatrix} \frac{1}{E_1} & \frac{\nu_{12}}{E_1} & 0 \\ \frac{\nu_{21}}{E_2} & \frac{1}{E_2} & 0 \\ 0 & 0 & \frac{1}{G_{12}} \end{bmatrix}$$

$$a = \begin{Bmatrix} \alpha_1 \\ \alpha_2 \\ 0 \end{Bmatrix}$$

2. Define the Unit Cell for Microstructural Design

Design Variable = Holes





Solution Method 2

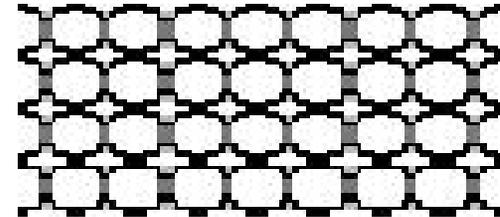
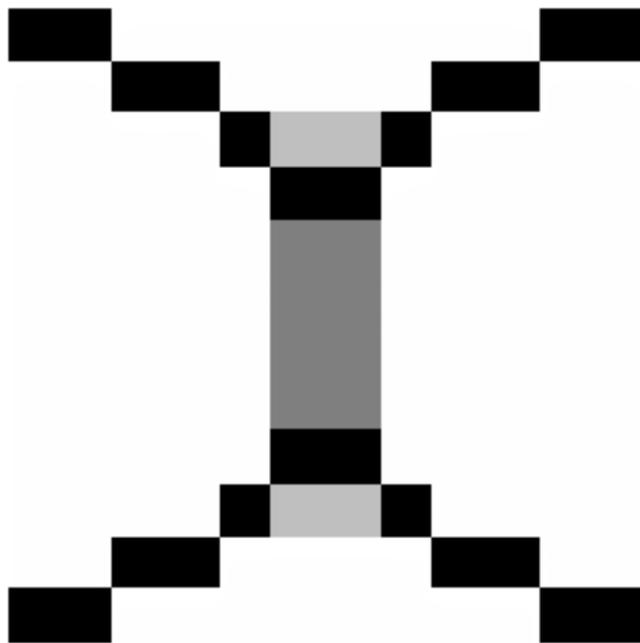
3. Apply the Homogenization Method to find the material constants for a design
4. Solve the Inverse Problem Based on the Least Squares Method

$$\min_{\text{design}} \frac{1}{2} \|\mathbf{D}_d - \mathbf{D}\|^2$$

with symmetry and periodic conditions



Bendsoe's Material



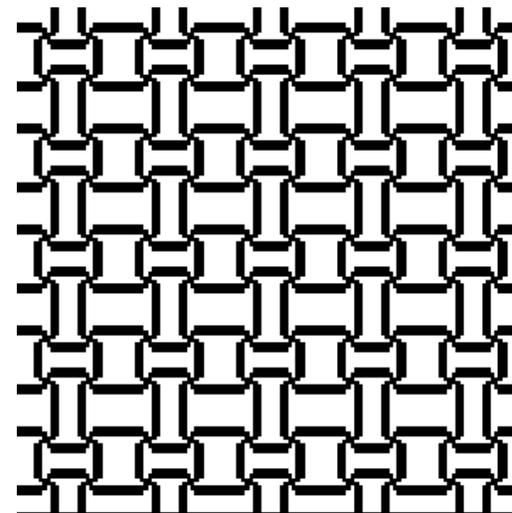
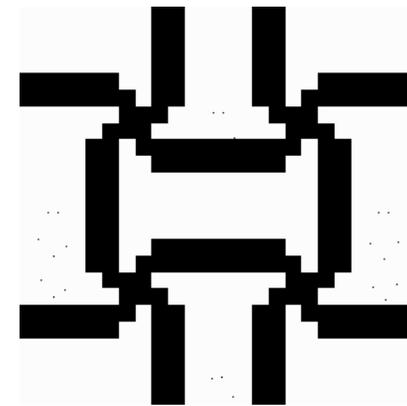
$$D = \begin{bmatrix} 1 & 0.48 & 0 \\ 0.48 & 0.25 & 0 \\ 0 & 0 & 0.02 \end{bmatrix}$$



Negative Poisson's Ratio

- *Plane stress*
- *Cubic Material (3 independent elastic constants)*

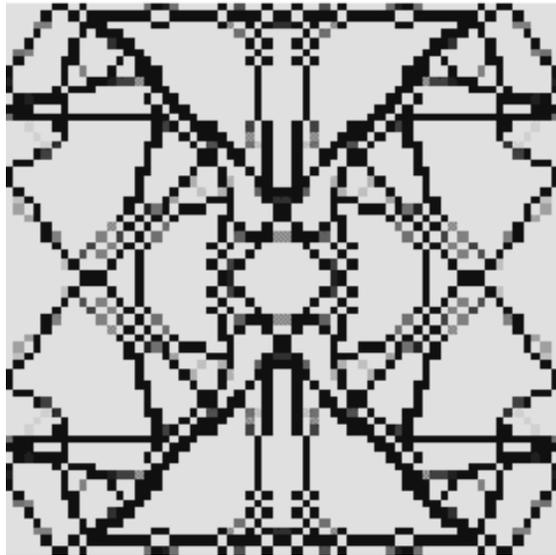
$$\mathbf{E} = 0.3434\mathbf{E}_0 \begin{bmatrix} 1 & -.66 & 0 \\ -.66 & 1 & 0 \\ 0 & 0 & .02 \end{bmatrix}$$



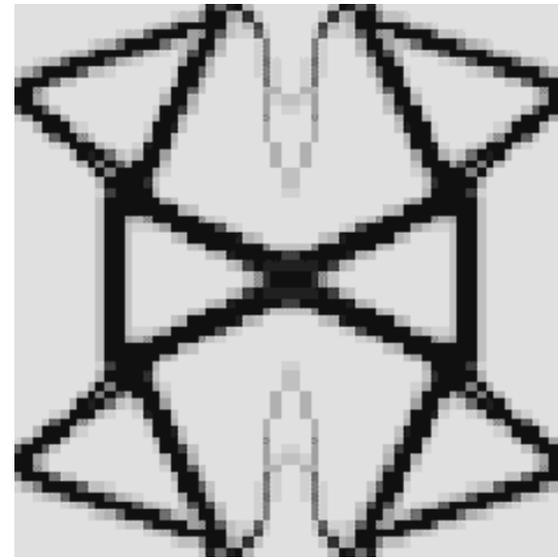


Application of a Filtering

isotropic negative Poisson's ratio microstructures



non filtered 60x60

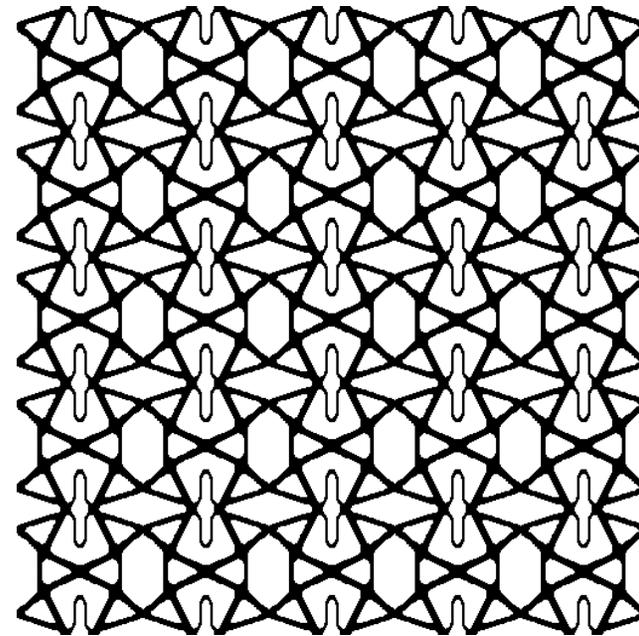
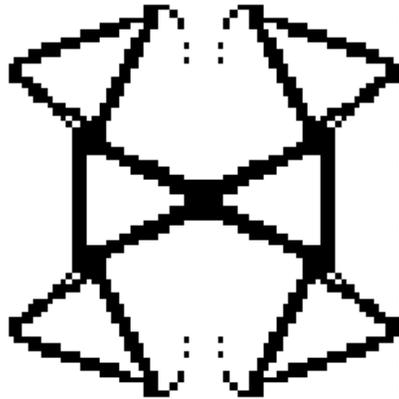


filtered 60x60

Easier Interpretation of the topology without changing of the properties



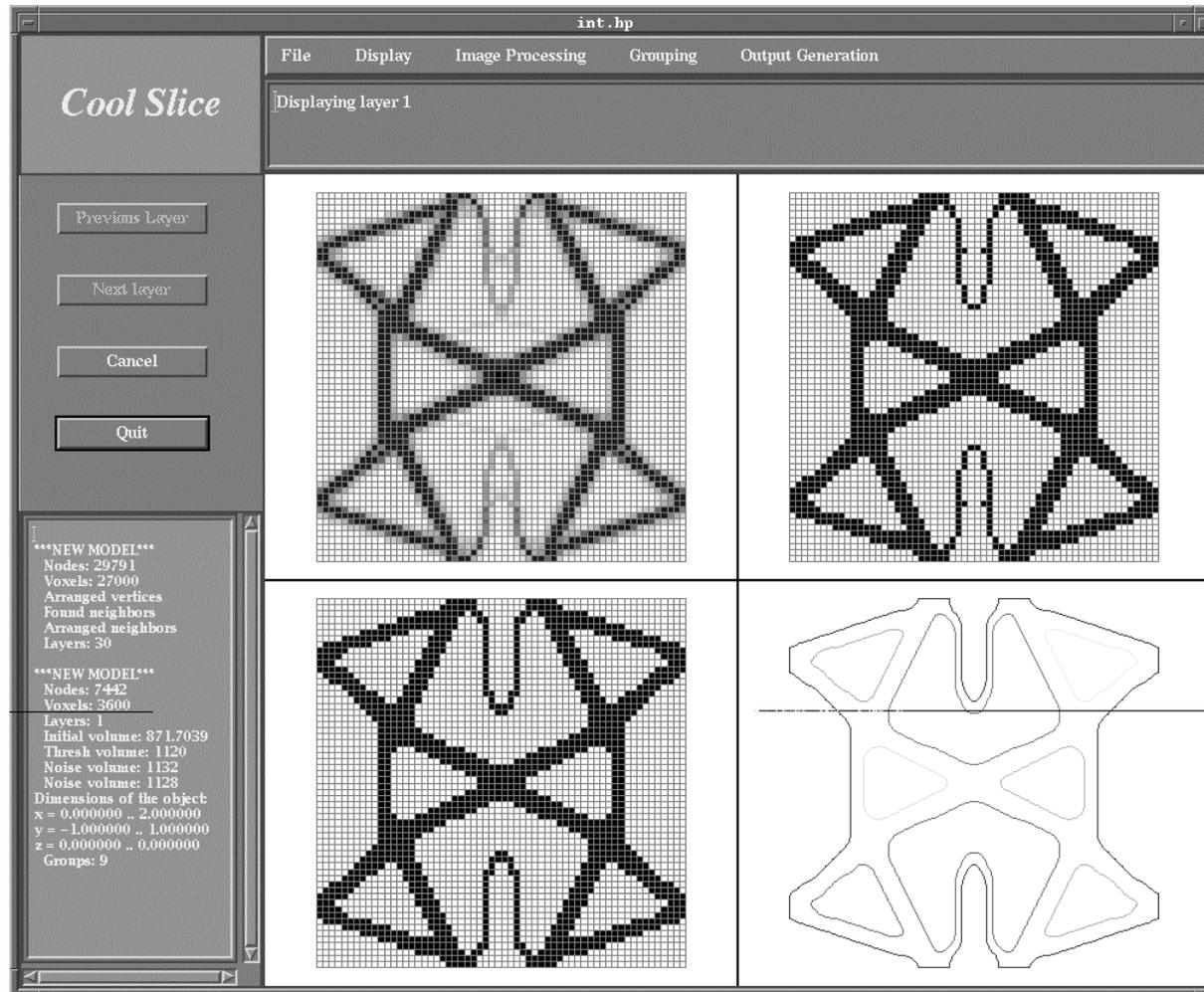
Isotropic Material Design



■ *Poisson's ratio -0.5*

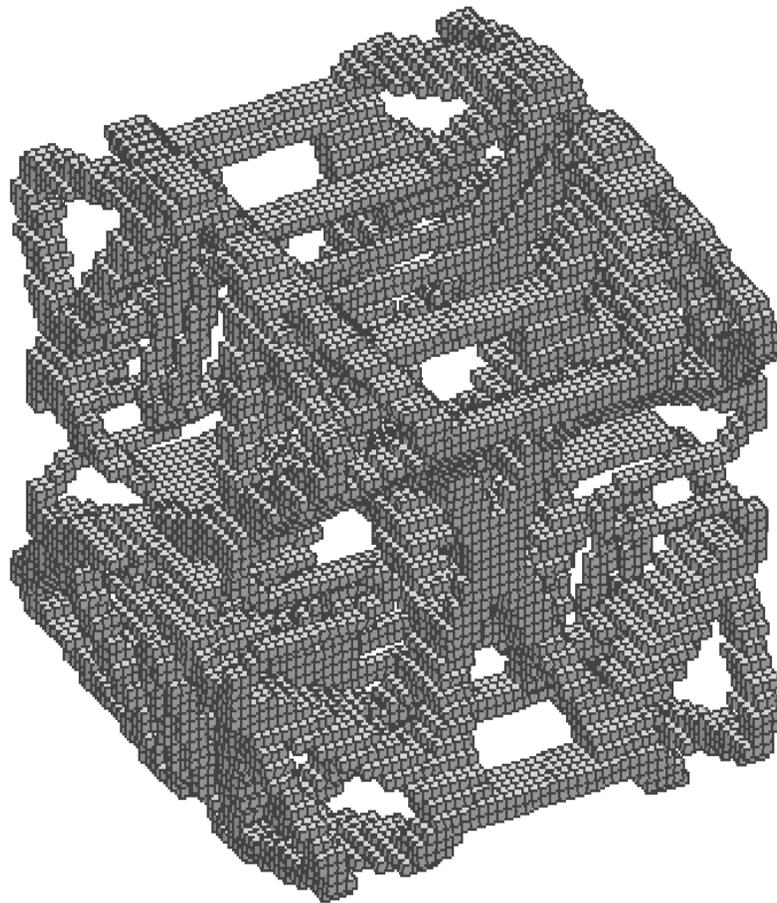


Jun Fonseca & Anne Marsan





Three Dimensional Design

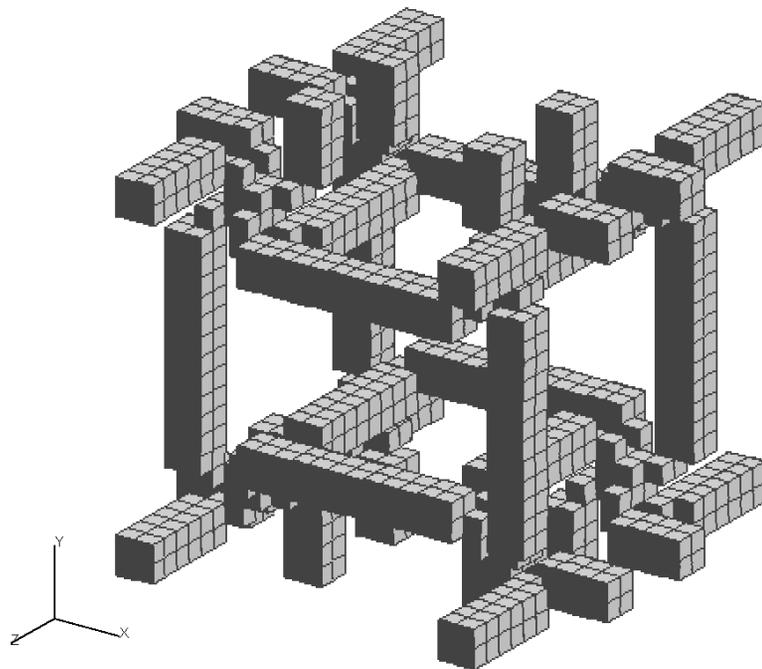


- *Tensor Product*
- *Orthotropic*
- *Two negative and one positive Poisson's ratios*

$$\mathbf{E} = \begin{bmatrix} 1 & -0.513 & 0.276 & 0 & 0 & 0 \\ -0.513 & 0.656 & -0.341 & 0 & 0 & 0 \\ 0.276 & -0.341 & 0.713 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0.254 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0.284 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0.286 \end{bmatrix}$$



3D Microstructure

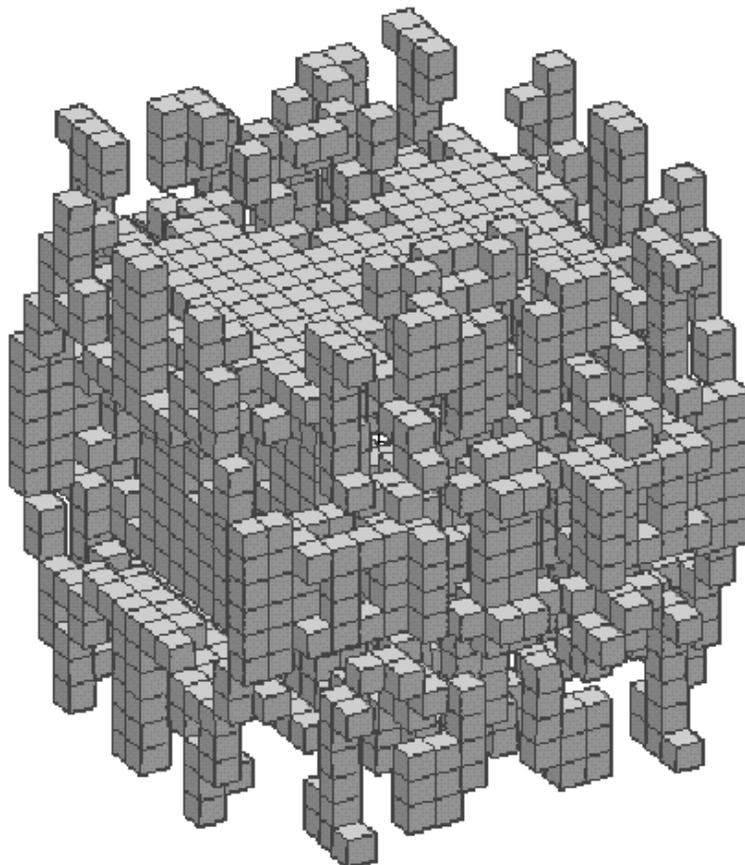


- *Cubic symmetry (3 independent elastic coefficients)*
- *Negative Poisson's ratio*

$$E = 10^{-4} E_0 \begin{bmatrix} 10 & -28 & -28 & 0 & 0 & 0 \\ -28 & 10 & -28 & 0 & 0 & 0 \\ -28 & -28 & 10 & 0 & 0 & 0 \\ 0 & 0 & 0 & .03 & 0 & 0 \\ 0 & 0 & 0 & 0 & .03 & 0 \\ 0 & 0 & 0 & 0 & 0 & .03 \end{bmatrix}$$



Another 3D Microstructure



- *Full 3D design*
- *smaller mesh:
20x20x20*
- *isotropic with
negative Poisson's
 $\nu = 1/4$*



Future Material Design

Jose M. Gedes : Portugal

and

J.E. Taylor : USA



Direct Use of Elasticity Matrix

$$\max_E \min_v \frac{1}{2} a(v, v) - f(v)$$

subject to

$$\int_D \rho dD \leq W_0$$

More design variables, more optimum



Still More Research on Homogenization Design

Continuous Development
and Enhancement of
OPTISHAPE



Practice Demands

many new feature
of
OPTISHAPE



Request from Practice

- Automatic surface recognition
 - Applying the image based modeling method developed by Minako Sekiguchi, we can define a smoothed three dimensional body
 - Then convert to STL format
- Automatic Mesh Generation for Detailed FE analysis
 - Applying VOXELCON to extract Wire Frame Model, then go to a CAD system & FE soft